



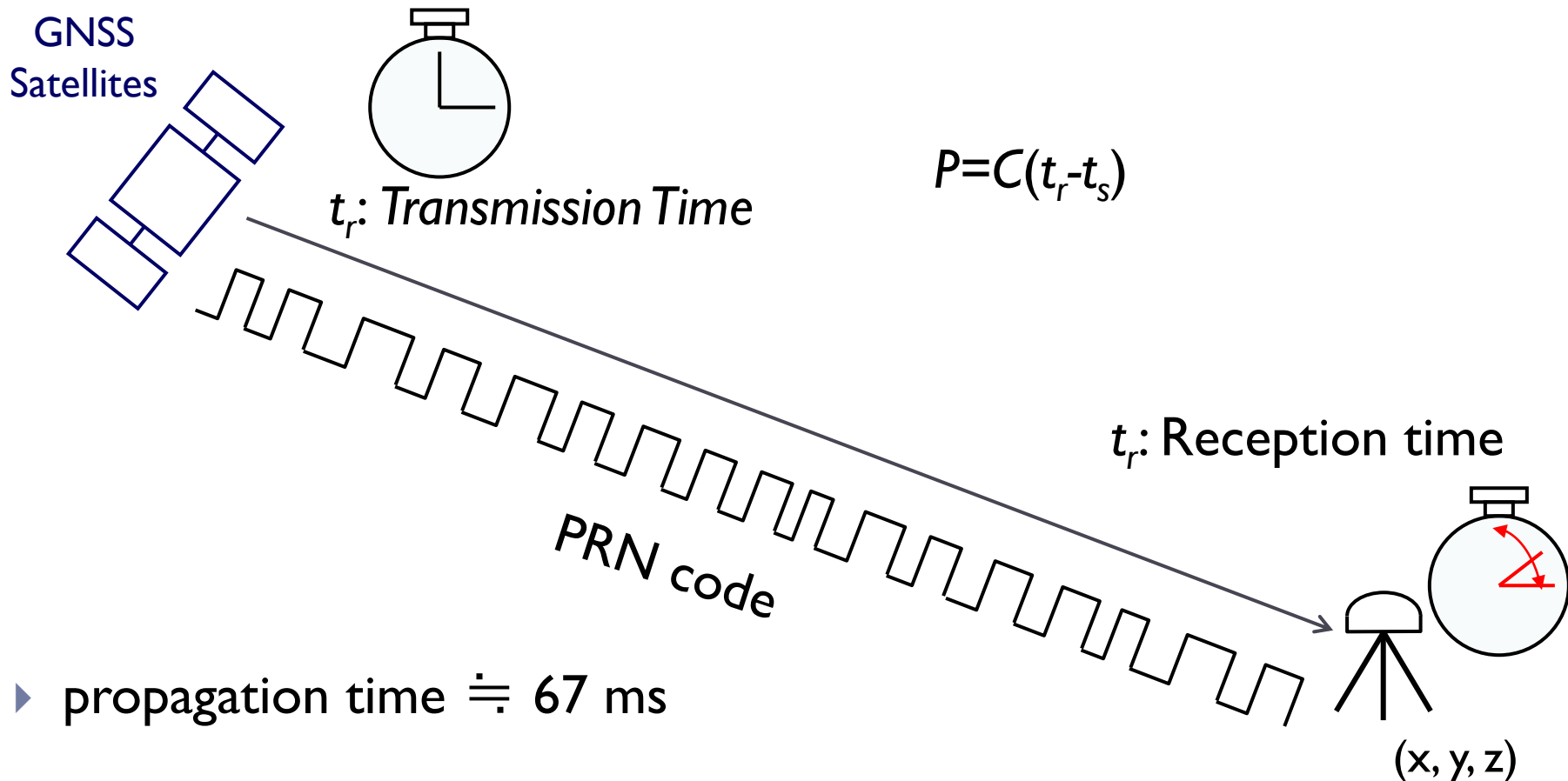
# 10: Positioning



Taro Suzuki

# Pseudorange Calculation (1)

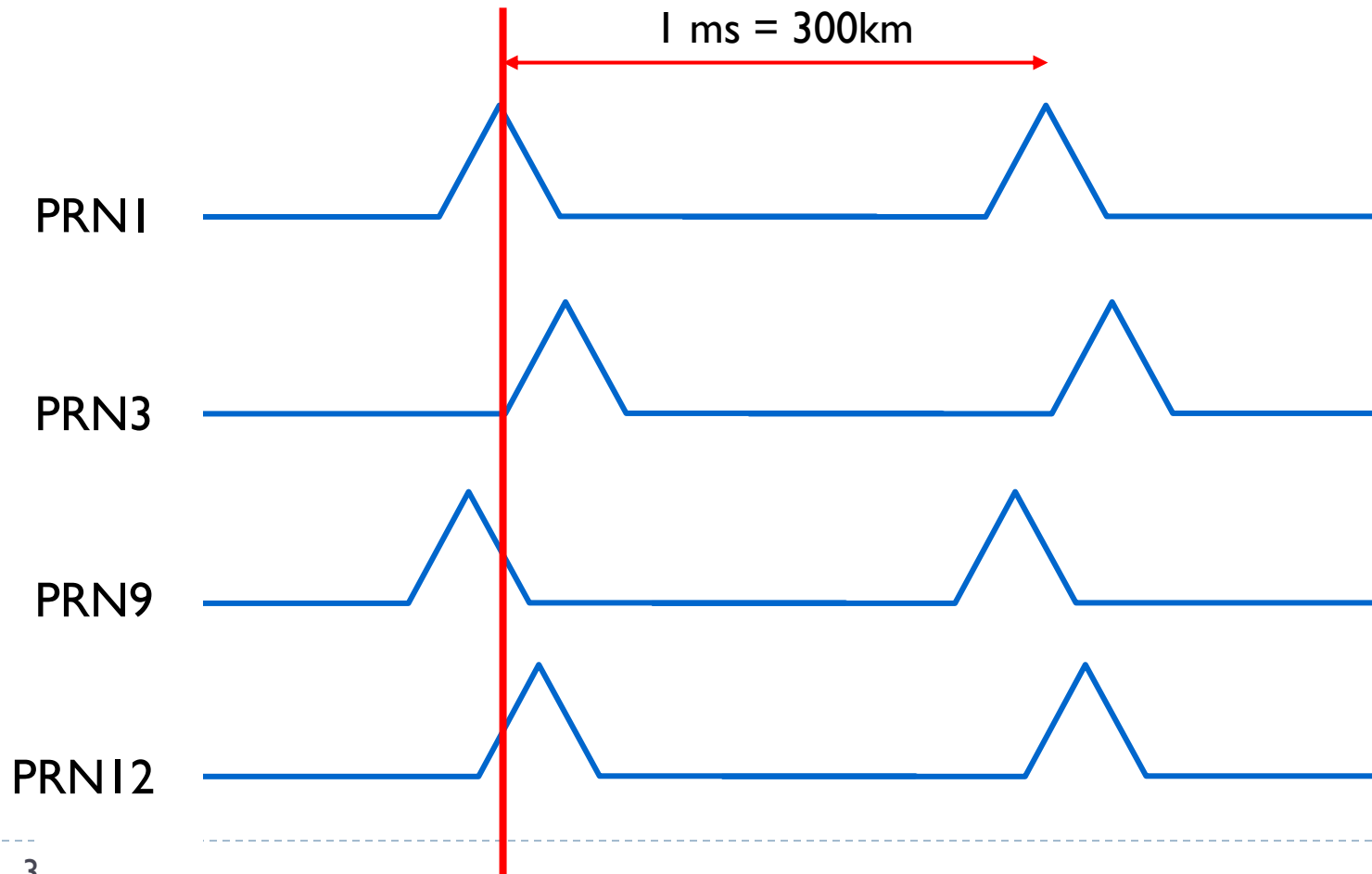
- ▶ Calculate distance based on difference between transmission time and reception time



- ▶ propagation time  $\doteq 67$  ms

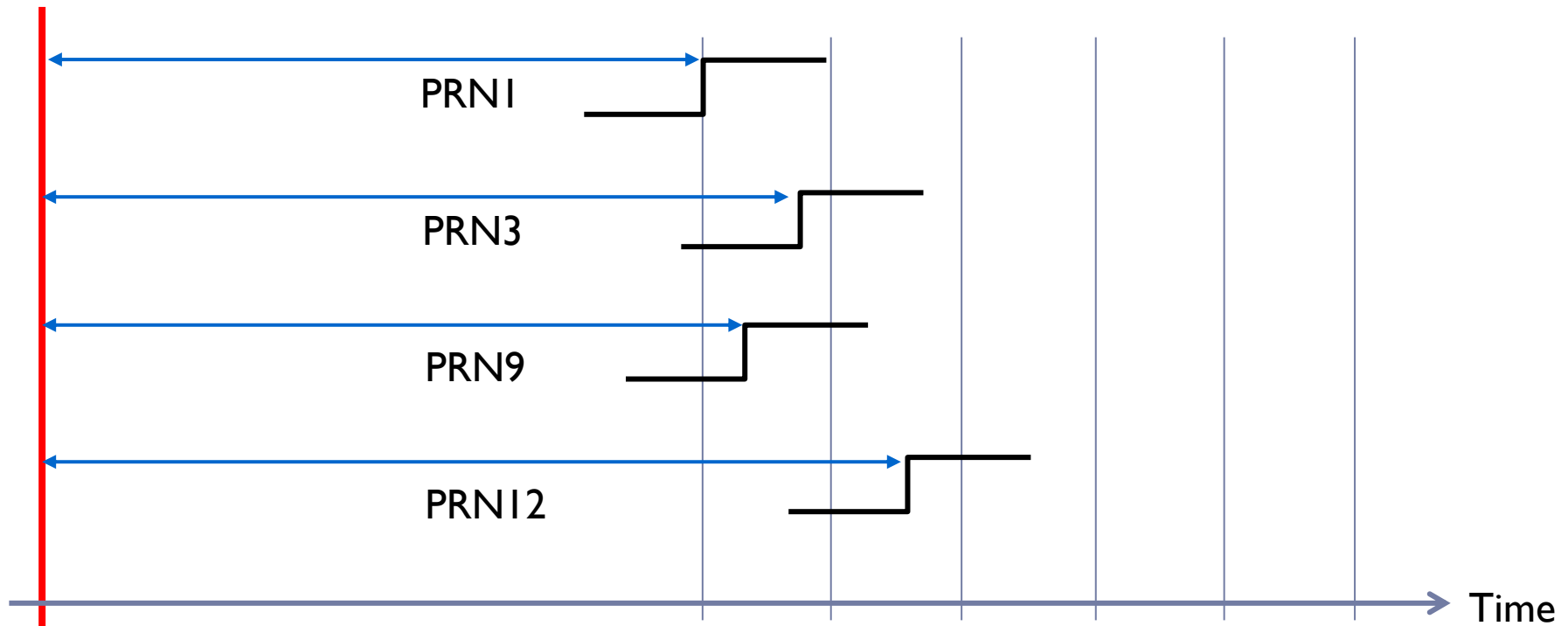
# Pseudorange Calculation (2)

- ▶ The code phase shift for each satellite is the relative difference in distance

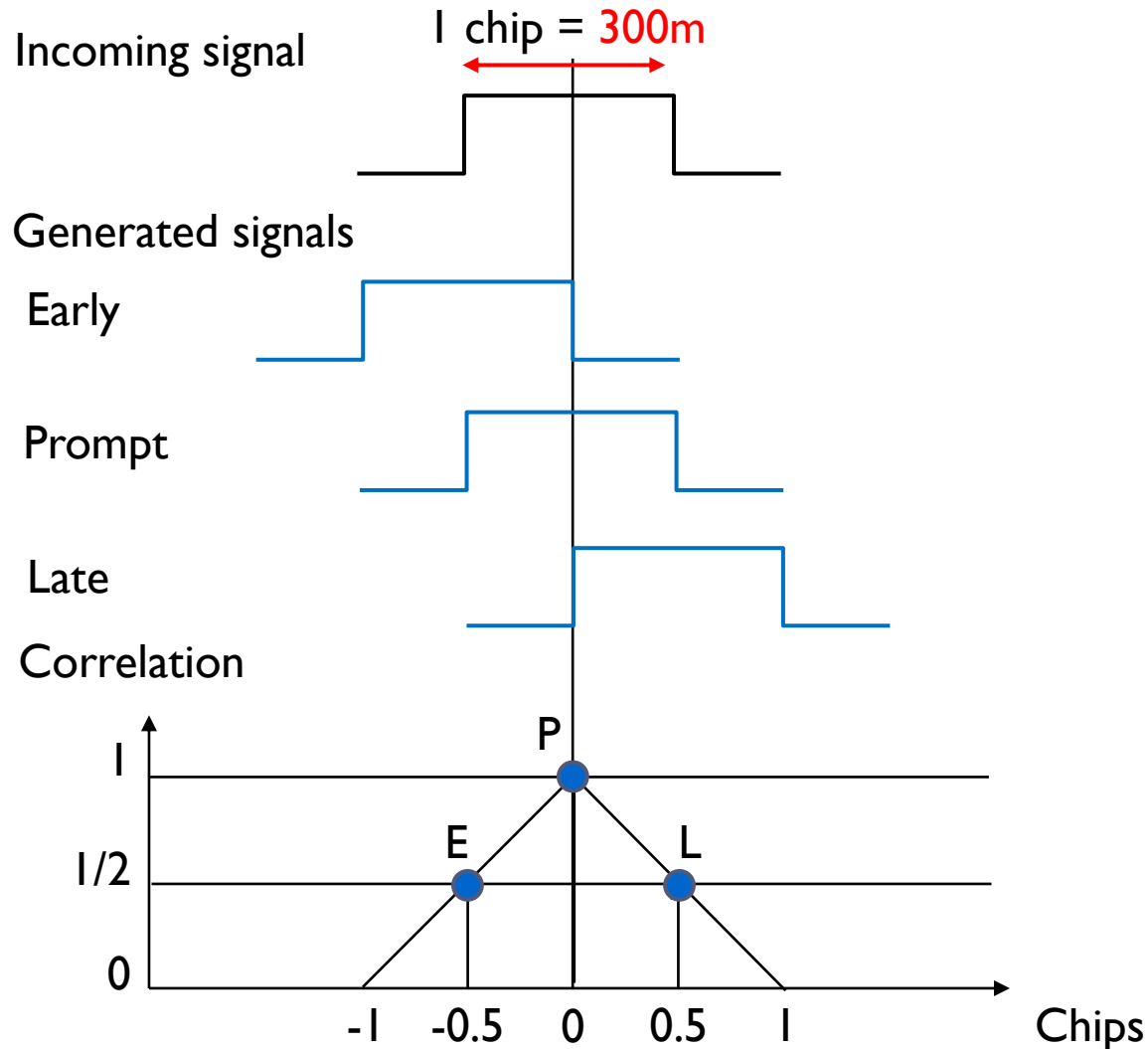


# Pseudorange Calculation (2)

- ▶ Find the time difference at the beginning of the subframe
- ▶ Time synchronization of less than 1 ms is adjusted by E-P-L correlator



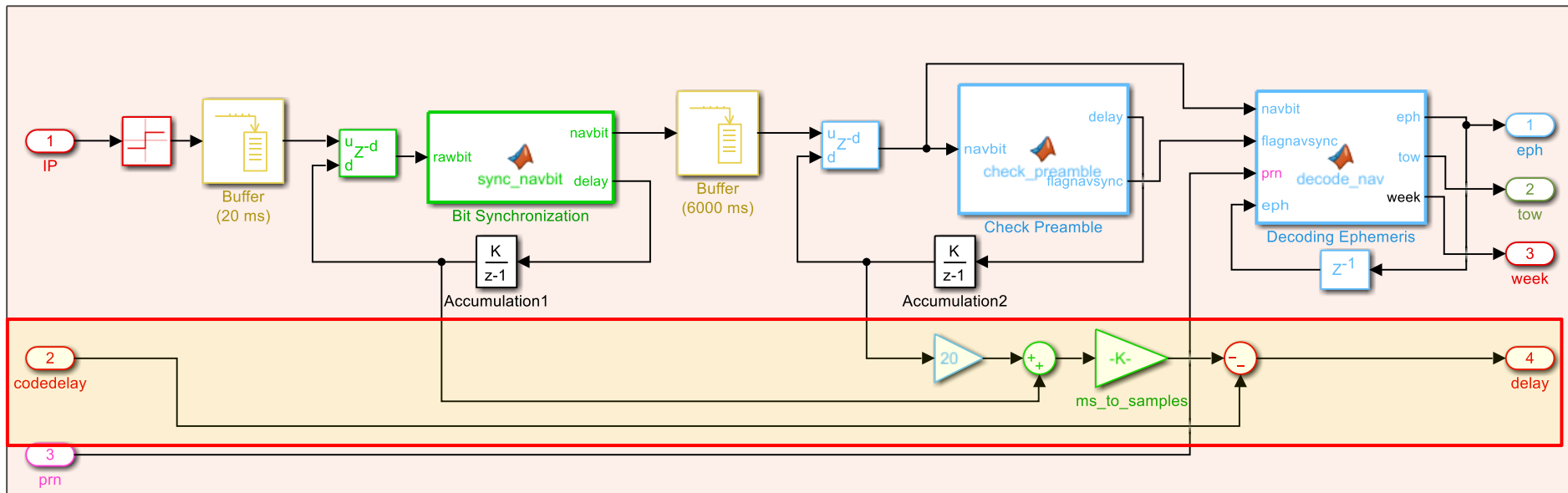
# Pseudorange Calculation (3)



# Delay Computation

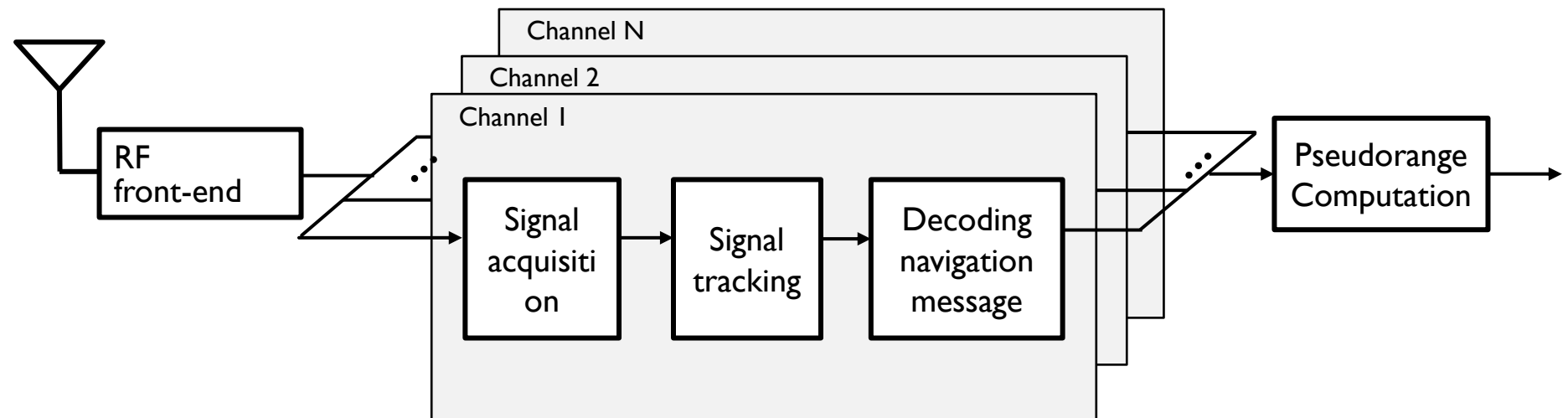
- ▶ Initial code phase + remcode
- ▶ Bit synchronization + subframe synchronization

## Decoding



# Parallel processing for each satellite signal

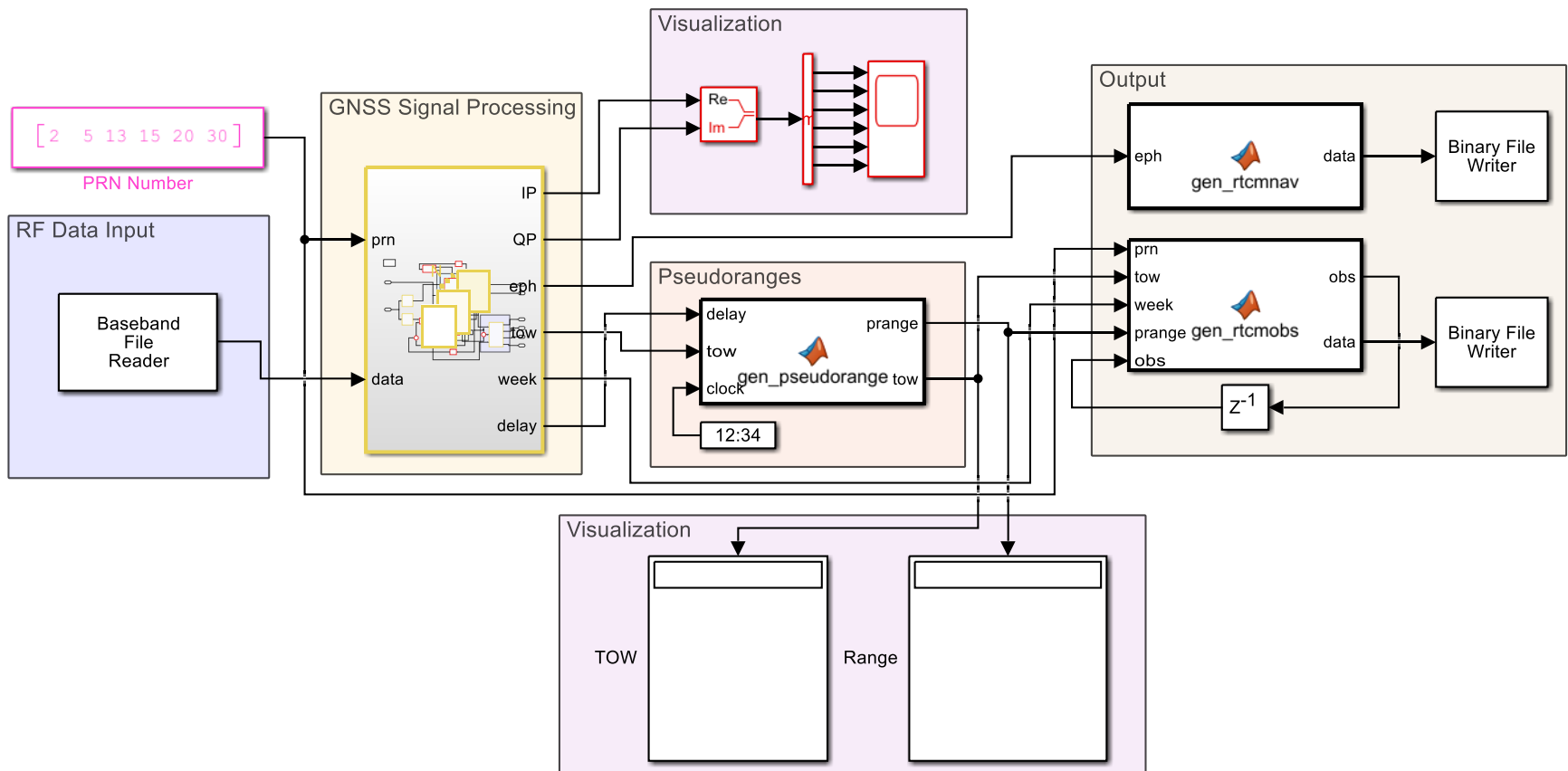
- ▶ Calculate the pseudoranges from the delay information of each channel at a certain timing of the receiver
- ▶ Aggregate the results of satellite signal tracking executed in parallel and calculate the range



# Simulink Model

## ► Use of For Each Subsystem

- Parallelized signal processing (acquisition, tracking, decoding)





# RINEX

## ► Receiver Independent Exchange Format

- ASCII
- Pseudorange, carrier phase, Doppler, Signal-to-Noise ratio

1	3.03	OBSERVATION DATA						M: Mixed	RINEX VERSION / TYPE↓
2	RTKCONV 2.4.3 b34							20220806 173351 UTC	PGM / RUN BY / DATE ↓
3									MARKER NAME ↓
4									MARKER NUMBER ↓
5									MARKER TYPE ↓
6									OBSERVER / AGENCY ↓
7									REC # / TYPE / VERS ↓
8									ANT # / TYPE ↓
9		0.0000		0.0000		0.0000		APPROX POSITION XYZ ↓	
10		0.0000		0.0000		0.0000		ANTENNA: DELTA H/E/N ↓	
11	G	1	C1C					SYS / # / OBS TYPES ↓	
12	2022	08	05	06	48	36.9000000	GPS	TIME OF FIRST OBS ↓	
13	2022	08	05	06	49	18.9000000	GPS	TIME OF LAST OBS ↓	
14	0							GLONASS SLOT / FRQ # ↓	
15	C1C	0.000	C1P	0.000	C2C	0.000	C2P	0.000	GLONASS COD/PHS/BIS ↓
16									END OF HEADER ↓
17	>	2022	08	05	06	48	42.0000000	0 6	
18	G02	20385887.144							
19	G05	19637026.467							
20	G13	19162987.397							
21	G15	20929127.718							
22	G20	19759870.608							
23	G30	21472839.361							

# RTCM

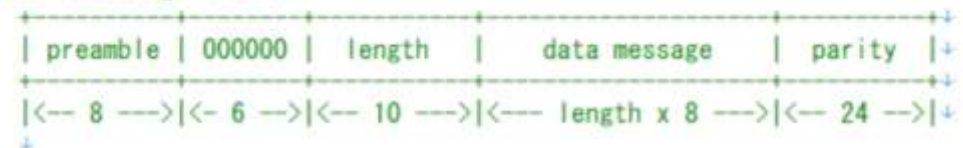
- ▶ Format for real-time GNSS communication
  - ▶ Binary
  - ▶ Originally a data format for differential correction, RTCM MSM, a format for communicating GNSS observations, was also specified

TYPE	:	GPS	GLONASS	Galileo	QZSS	BDS	SBAS	NavIC
OBS COMP L1	:	1001~	1009~	-	-	-	-	~↓
FULL L1	:	1002	1010	-	-	-	-	~↓
COMP L1L2	:	1003~	1011~	-	-	-	-	~↓
FULL L1L2	:	1004	1012	-	-	-	-	~↓
NAV	:	1019	1020	1045** 1046**	1044 -	1042 63*	-	1041~ ~↓
MSM 1	:	1071~	1081~	1091~	1111~	1121~	1101~	1131~↓
2	:	1072~	1082~	1092~	1112~	1122~	1102~	1132~↓
3	:	1073~	1083~	1093~	1113~	1123~	1103~	1133~↓
4	:	1074	1084	1094	1114	1124	1104	1134↓
5	:	1075	1085	1095	1115	1125	1105	1135↓
6	:	1076	1086	1096	1116	1126	1106	1136↓
7	:	1077	1087	1097	1117	1127	1107	1137↓
SSR ORBIT	:	1057	1063	1240*	1246*	1258*	-	~↓
CLOCK	:	1058	1064	1241*	1247*	1259*	-	~↓
CODE BIAS	:	1059	1065	1242*	1248*	1260*	-	~↓
OBT/CLK	:	1060	1066	1243*	1249*	1261*	-	~↓
URA	:	1061	1067	1244*	1250*	1262*	-	~↓
HR-CLOCK	:	1062	1068	1245*	1251*	1263*	-	~↓
PHAS BIAS	:	11*	-	12*	13*	14*	-	~↓
ANT/RCV INFO	:	1007	1008	1033↓				
STA POSITION	:	1005	1006↓					
PROPRIETARY	:	4076 (IGS)↓						

(\* draft, \*\* 1045:F/NAV, 1046:I/NAV, ~ only encode)↓

for MSM observation data with multiple signals for a frequency, ↓  
a signal is selected according to internal priority. to select ↓  
a specified signal, use the input options. ↓

RTCM 3 message format: ↓



# Positioning Computation (1)

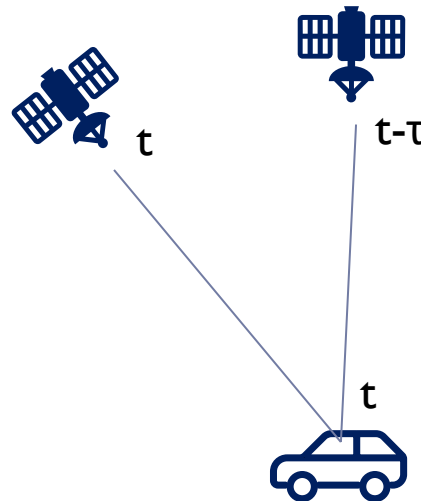
$$\rho_c^{(k)} = r^{(k)} + c\delta t_u + \tilde{\varepsilon}_\rho^{(k)}$$

$$\begin{cases} r^{(k)} = \sqrt{(x^{(k)} - x)^2 + (y^{(k)} - y)^2 + (z^{(k)} - z)^2} = \|\mathbf{x}^{(k)} - \mathbf{x}\| \\ b = c\delta t_u \end{cases}$$

$$\rho_c^{(k)} = \|\mathbf{x}^{(k)} - \mathbf{x}\| + b + \tilde{\varepsilon}_\rho^{(k)}$$

Satellite position at  $t-\tau$

$$\mathbf{x}^{(k)} = \begin{bmatrix} \cos \omega_E \tau & \sin \omega_E \tau & 0 \\ -\sin \omega_E \tau & \cos \omega_E \tau & 0 \\ 0 & 0 & 1 \end{bmatrix} \tilde{\mathbf{x}}^{(k)}$$



$\rho_c^{(k)}$  Compensated Pseudorange

$r^{(k)}$  True range

$\delta t_u$  Receiver clock error

$\tilde{\varepsilon}_\rho^{(k)}$  Pseudorange noise

$\mathbf{x}^{(k)}$  Satellite position at  $t$

$\tilde{\mathbf{x}}^{(k)}$  Satellite position at  $t-\tau$

$\mathbf{x}$  User position

$b$  Receiver clock error [m]

$\omega_E$  Earth rotation rate

$\tau$  Propagation time

# Positioning Computation (2)

---

$$\rho_c^{(k)} = \|\mathbf{x}^{(k)} - \mathbf{x}\| + b$$

Linearize the observed equations for the number of and find the correct solution by iteration

$$\mathbf{x} = \mathbf{x}_0 + \delta\mathbf{x} = \begin{pmatrix} x_0 + \delta x \\ y_0 + \delta y \\ z_0 + \delta z \end{pmatrix}$$

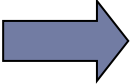
$$b = b_0 + \delta b$$

$$\mathbf{l}^{(k)} = \begin{pmatrix} \alpha^{(k)} \\ \beta^{(k)} \\ \gamma^{(k)} \end{pmatrix} = \begin{pmatrix} \frac{x^{(k)} - x_0}{R_0^k} \\ \frac{y^{(k)} - y_0}{R_0^k} \\ \frac{z^{(k)} - z_0}{R_0^k} \end{pmatrix}$$


$$\begin{aligned} \delta\rho^{(k)} &= \rho_c^{(k)} - \rho_0^{(k)} \\ &= \|\mathbf{x}^{(k)} - \mathbf{x}_0 - \delta\mathbf{x}\| - \|\mathbf{x}^{(k)} - \mathbf{x}_0\| + (b - b_0) + \tilde{\varepsilon}_\rho^{(k)} \\ &\cong -\frac{(\mathbf{x}^{(k)} - \mathbf{x}_0)}{\|\mathbf{x}^{(k)} - \mathbf{x}_0\|} \cdot \delta\mathbf{x} + \delta b + \tilde{\varepsilon}_\rho^{(k)} \\ &= -\mathbf{l}^{(k)} \cdot \delta\mathbf{x} + \delta b + \tilde{\varepsilon}_\rho^{(k)} \end{aligned}$$

$$R_0^{(k)} = \sqrt{(x^{(k)} - x_0)^2 + (y^{(k)} - y_0)^2 + (z^{(k)} - z_0)^2}$$

# Positioning Computation (3)

$$\delta \mathbf{p} = \begin{bmatrix} \delta \rho^{(1)} \\ \delta \rho^{(2)} \\ \vdots \\ \delta \rho^{(K)} \end{bmatrix} = \begin{bmatrix} (-\mathbf{I}^{(1)})^T & 1 \\ (-\mathbf{I}^{(2)})^T & 1 \\ \vdots & \vdots \\ (-\mathbf{I}^{(K)})^T & 1 \end{bmatrix} \begin{bmatrix} \delta \mathbf{x} \\ \delta b \end{bmatrix} + \tilde{\boldsymbol{\epsilon}}_\rho$$


$$\delta \mathbf{p} = \mathbf{G} \begin{bmatrix} \delta \mathbf{x} \\ \delta b \end{bmatrix} + \tilde{\boldsymbol{\epsilon}}_\rho$$

$$\mathbf{G} = \begin{bmatrix} -\alpha^{(1)} & -\beta^{(1)} & -\gamma^{(1)} & 1 \\ -\alpha^{(2)} & -\beta^{(2)} & -\gamma^{(2)} & 1 \\ \vdots & \vdots & \vdots & 1 \\ -\alpha^{(K)} & -\beta^{(K)} & -\gamma^{(K)} & 1 \end{bmatrix}$$


least-squares solution  $\begin{bmatrix} \delta \hat{\mathbf{x}} \\ \delta \hat{b} \end{bmatrix} = (\mathbf{G}^T \mathbf{G})^{-1} \mathbf{G}^T \delta \mathbf{p}$

Recalculate the first estimate corrected by the above solution. If the estimates of the next calculation converge, the calculation is completed.

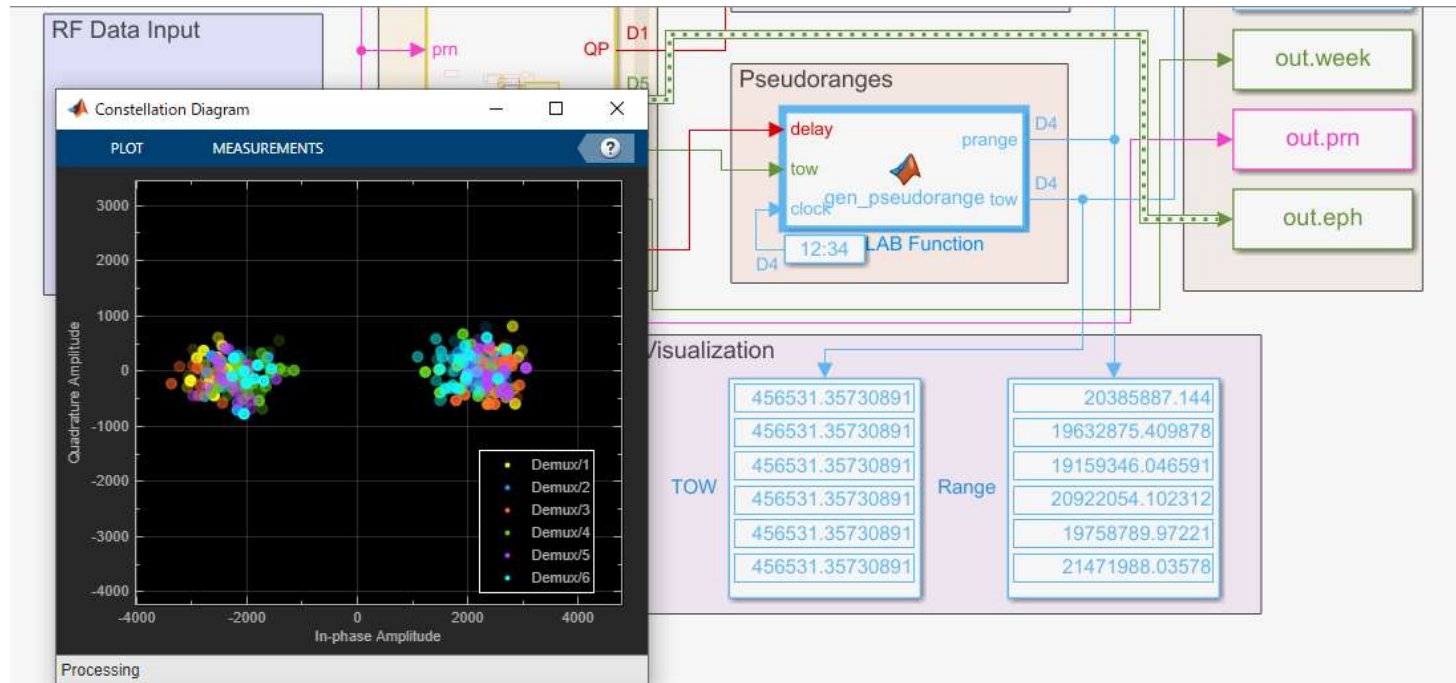
$$\hat{\mathbf{x}} = \mathbf{x}_0 + \delta \hat{\mathbf{x}}$$

$$\hat{b} = b_0 + \delta \hat{b}$$

# Exercise 1: Positioning using RTKLIB (1)

## ► Simulink

- /I0\_Positioning + SDR\_Application/simulink/**Ex1/sdr\_positioning.slx**



- Pseudoranges are saved to MATLAB Workspace

# Exercise 1: Positioning using RTKLIB (2)

## ► MATLAB

- /I0\_Positioning+SDR\_Application/simulink/Ex1/run\_gen\_rinex.m
- Pseudoranges and time information is converted to RINEX file

sdr\_obs.obs

```
1 3.02 OBSERVATION DATA G: GPS RINEX VERSION / TYPE
2 20220809 155724 UTC PGM / RUN BY / DATE
3 MARKER NAME
4 MARKER NUMBER
5 MARKER TYPE
6 OBSERVER / AGENCY
7 REC # / TYPE / VERS
8 ANT # / TYPE
9 0.0000 0.0000 0.0000 APPROX POSITION XYZ
10 0.0000 0.0000 0.0000 ANTENNA: DELTA H/E/N
11 G 1 C1C SYS / # / OBS TYPES
12 2022 08 05 06 48 47.9573097 GPS TIME OF FIRST OBS
13 2022 08 05 06 49 23.9573016 GPS TIME OF LAST OBS
14 0 GLONASS SLOT / FRQ #
15 GLONASS COD/PHS/BIS
16 END OF HEADER
17 > 2022 08 05 06 48 47.9573097 0 6
18 G02 20385887.144 +
19 G05 19634375.952 +
20 G13 19160660.915 +
21 G15 20924612.079 +
22 G20 19759181.563 +
23 G30 21472290.224 +
24 > 2022 08 05 06 48 48.0573097 0 6
25 G02 20385887.144 +
26 G05 19634333.835 +
27 G13 19160622.307 +
28 G15 20924535.654 +
29 G20 19759168.656 +
30 G30 21472279.239 +
31 > 2022 08 05 06 48 48.1573096 0 6
32 G02 20385887.144 +
33 G05 19634333.835 +
```

sdr\_nav.nav

```
1 3.02 N: GNSS NAV DATA M: Mixed RINEX VERSION / TYPE
2 20220809 155724 UTC PGM / RUN BY / DATE
3 END OF HEADER
4 G02 2022 08 05 07 59 44 - 649891328067D-03 - 909494701773D-12 - 000000000000D+00
5 800000000000D+01 - 567812500000D-02 - 439696886569D-08 - 134245450696D+01
6 - 280886888504D-05 - 199471334927D-01 - 915862619877D-05 - 515487974548D+04
7 460784000000D+06 - 335276126862D-06 - 127323360928D+01 - 342726707458D-06
8 966132581037D+00 - 208343750000D+03 - 140052384930D+01 - 824070040111D-08
9 - 167864135072D-09 - 100000000000D+01 - 222100000000D+04 - 000000000000D+00
10 200000000000D+01 - 000000000000D+00 - 172294676304D-07 - 800000000000D+01
11 456546000000D+06 - 000000000000D+00
12 G05 2022 08 05 08 00 00 - 917343422771D-04 - 136424205266D-11 - 000000000000D+00
13 480000000000D+02 - 125625000000D+02 - 451804533758D-08 - 127297544227D+01
14 - 659376382828D-06 - 611729302909D-02 - 811740756035D-05 - 515376311874D+04
15 460800000000D+06 - 335276126862D-07 - 235706787130D+01 - 447034835815D-07
16 961843973782D+00 - 229406250000D+03 - 107979299179D+01 - 812212403335D-08
17 255367779950D-09 - 100000000000D+01 - 222100000000D+04 - 000000000000D+00
18 200000000000D+01 - 000000000000D+00 - 111758708954D-07 - 480000000000D+02
19 456546000000D+06 - 000000000000D+00
20 G13 2022 08 05 08 00 00 - 361989717931D-03 - 716227077646D-11 - 000000000000D+00
21 160000000000D+02 - 255625000000D+02 - 445268547222D-08 - 474524408406D+00
22 - 140443444252D-05 - 623134977650D-02 - 781938433647D-05 - 515367332649D+04
23 460800000000D+06 - 745058059692D-07 - 270109467665D+01 - 912696123123D-07
24 969177497264D+00 - 234468750000D+03 - 920800360242D+00 - 804354933183D-08
25 - 164649715464D-09 - 100000000000D+01 - 222100000000D+04 - 000000000000D+00
26 200000000000D+01 - 000000000000D+00 - 111758708954D-07 - 160000000000D+02
27 456546000000D+06 - 000000000000D+00
28 G15 2022 08 05 08 00 00 - 397786498070D-04 - 306954461848D-11 - 000000000000D+00
29 930000000000D+02 - 254375000000D+02 - 542486882470D-08 - 505896645654D-01
```

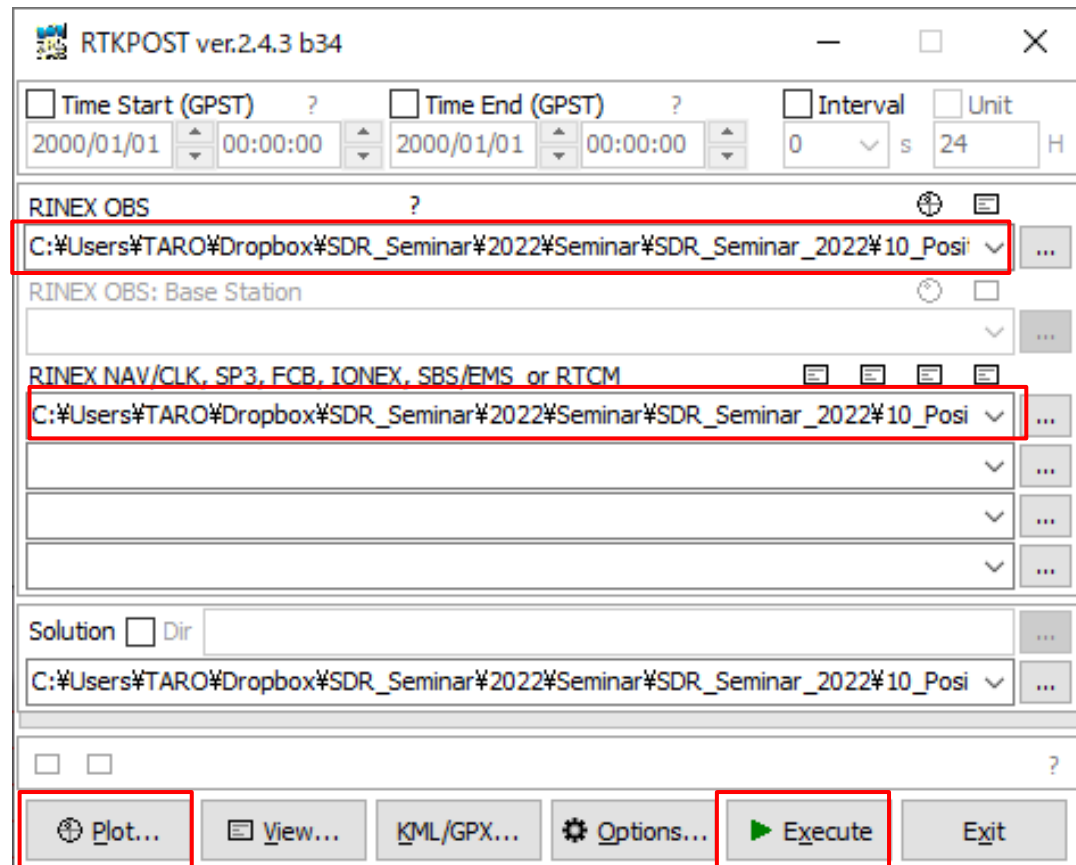


# Exercise 1: Positioning using RTKLIB (3)

## ► Run /rtklib/rtkpost.exe

sdr obs.obs

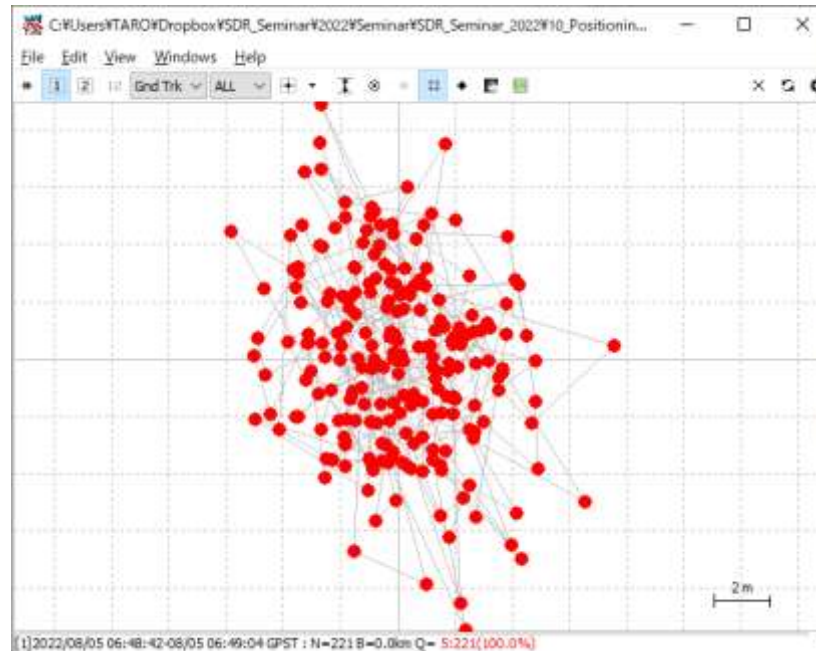
sdr nav.nav





# Exercise 1: Positioning using RTKLIB (4)

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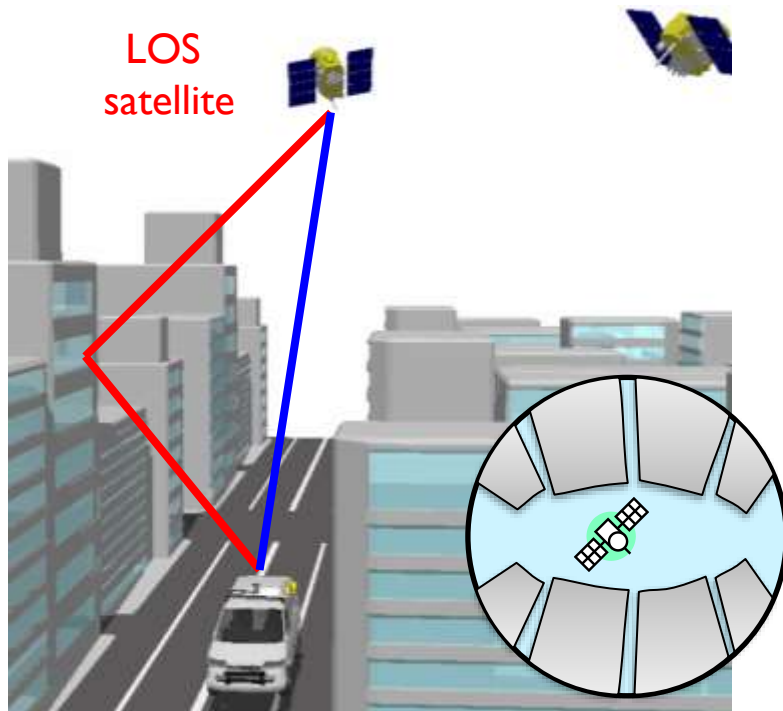
- ▶ **Advanced challenge**
  - ▶ Estimate position with RF data acquired by yourself!
  - ▶ Compare positioning results by changing the DLL noise bandwidth
  - ▶ Realtime processing using RTL-SDR dongle!

# 10: SDR Application

Taro Suzuki

# NLOS Multipath

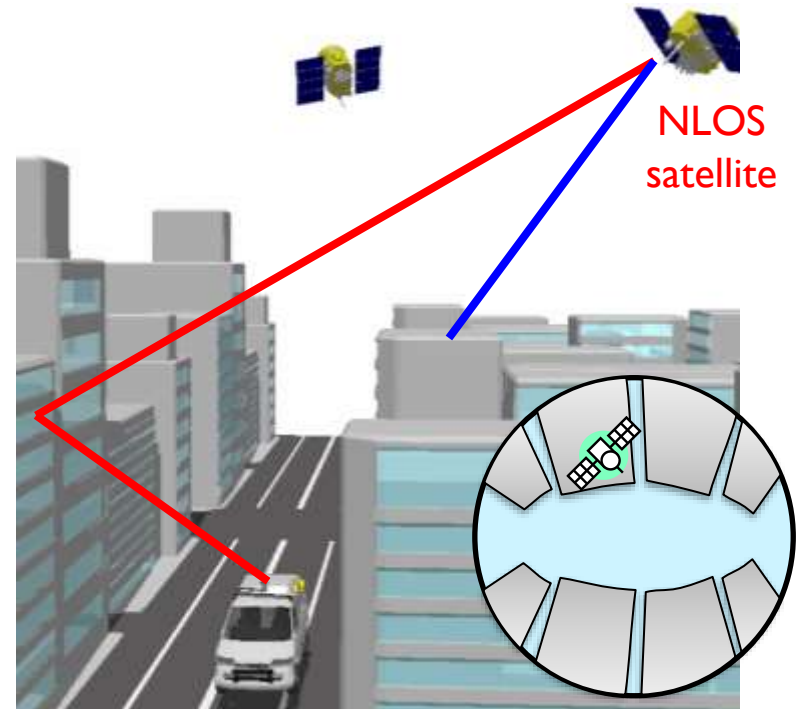
## LOS Multipath



### LOS(Line-Of-Sight) Multipath

- Direct + Reflected / Diffracted Signal
- Correlation technique is available

## NLOS Multipath

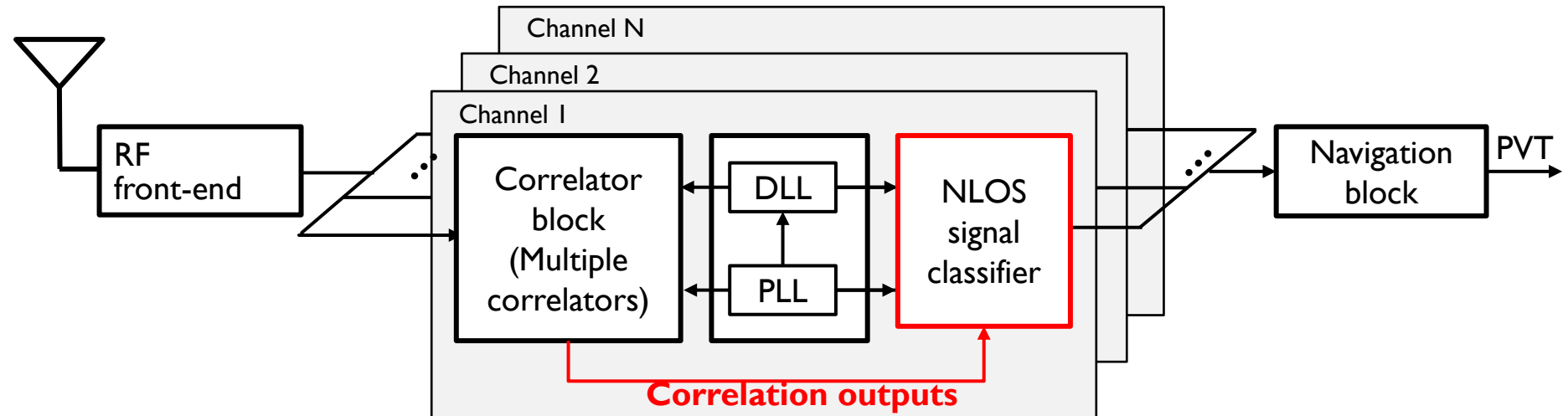


### NLOS(Non-Line-Of-Sight) Multipath

- Only Reflected / Diffracted Signal
- Correlation technique is NOT available

# NLOS Classifier

## Software GNSS Receiver

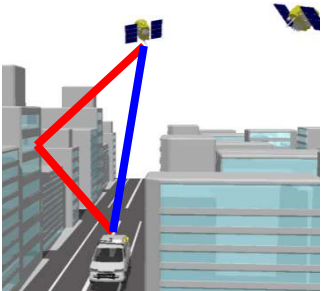


➡ Add **NLOS signal classifier** after signal correlator block

Direct machine learning of GNSS signal correlation output, which is the most primitive GNSS signal processing output!

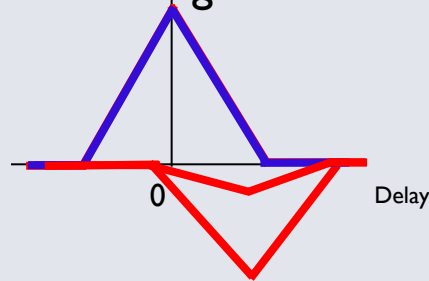
# Multipath Effects for Signal Correlation

## LOS Multipath



Direct and reflection/diffraction signals

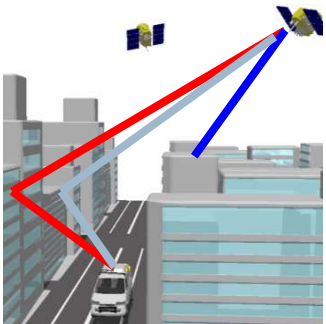
- Direct signal is distorted by the reflected signal



Distorted correlation function

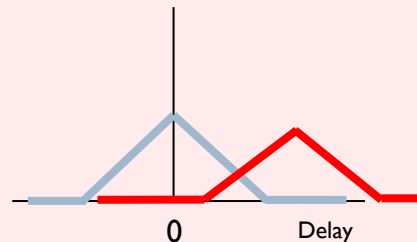


## NLOS Multipath

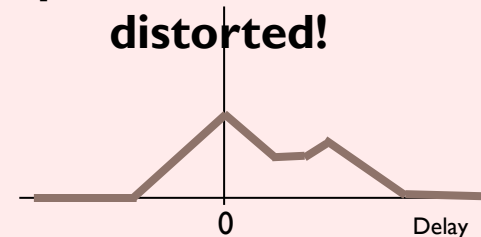


Only reflection/diffraction signals

- No direct signal
- All signals are weak



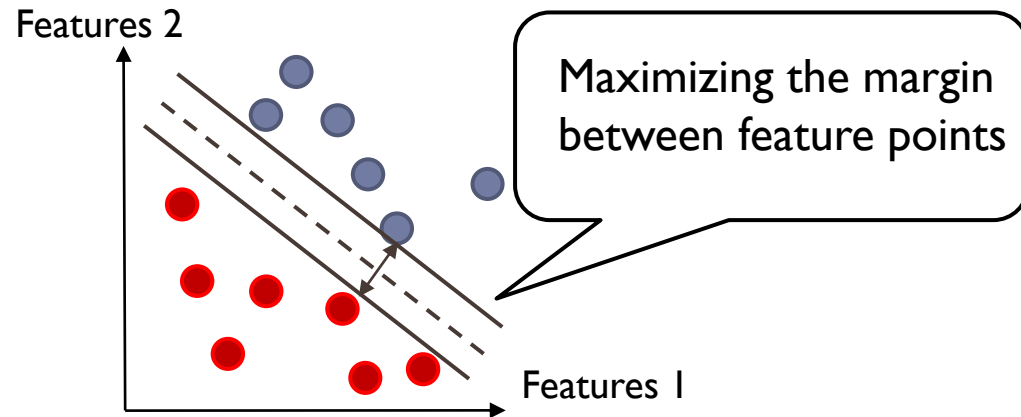
**Correlation function is expected to be more distorted!**



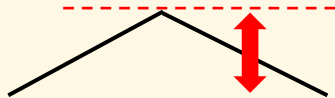
# Related Research (Support Vector Machine)

## Support Vector Machine

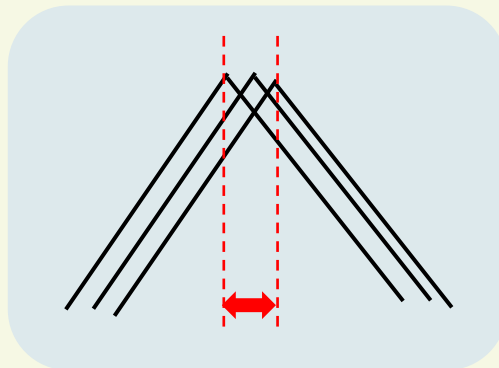
- Classification method using Supervised learning
- Maximizing the margin



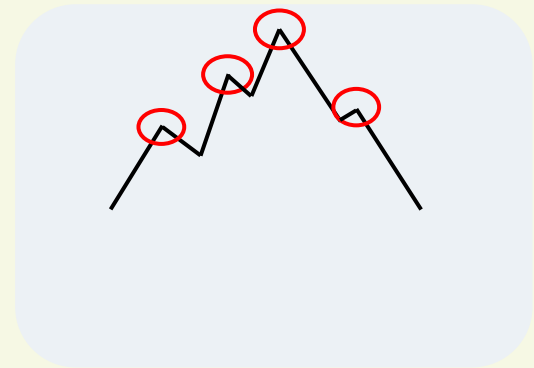
## NLOS features for SVM



Signal strength vs. elevation angle

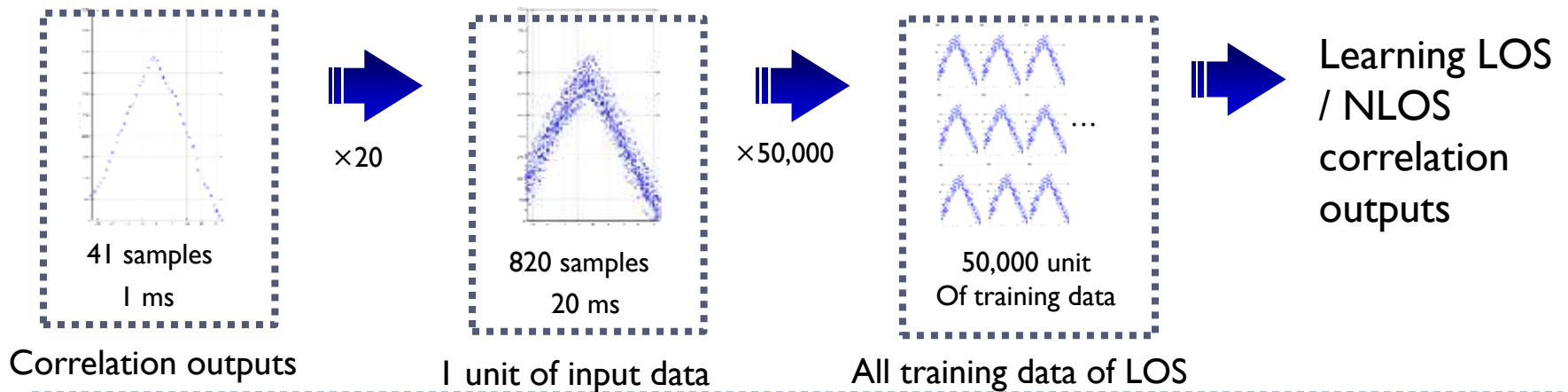
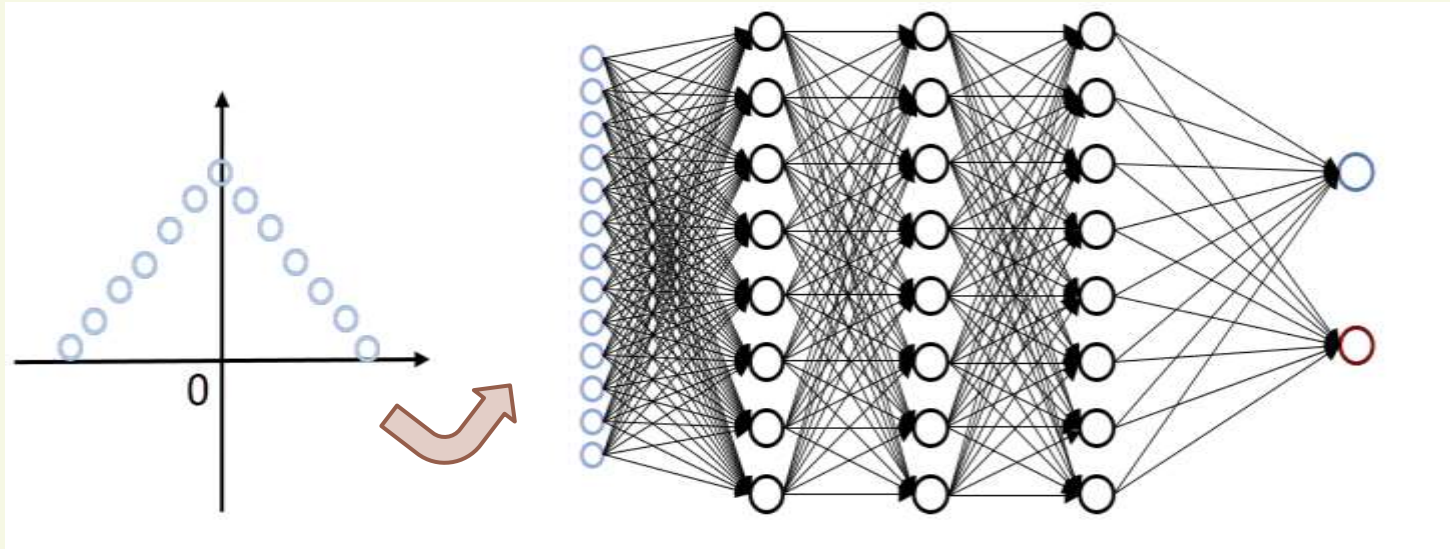


Distribution of delay of maximum correlation output



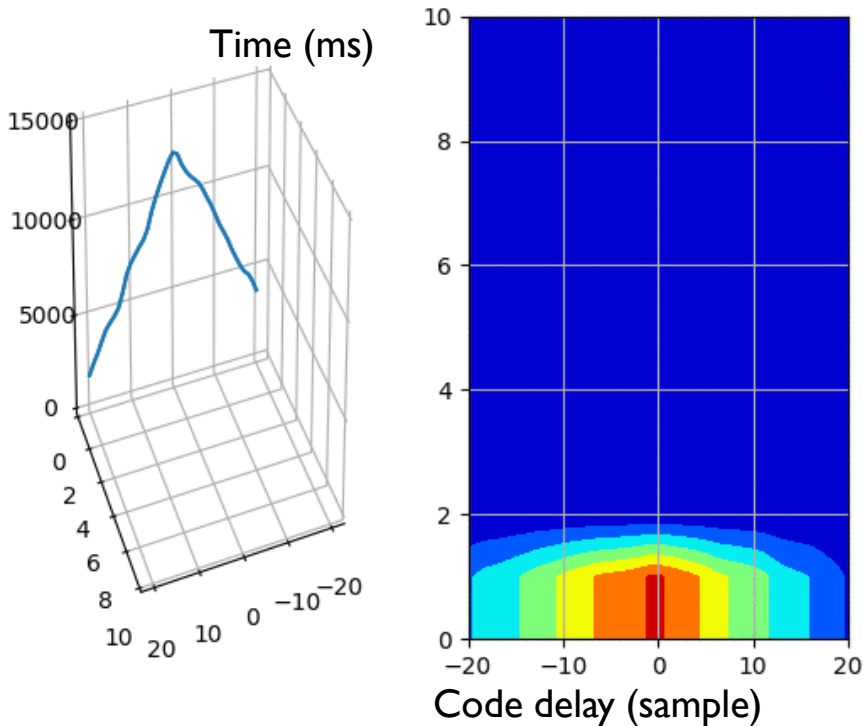
The number of local maxima

# Related Research (Neural Network)

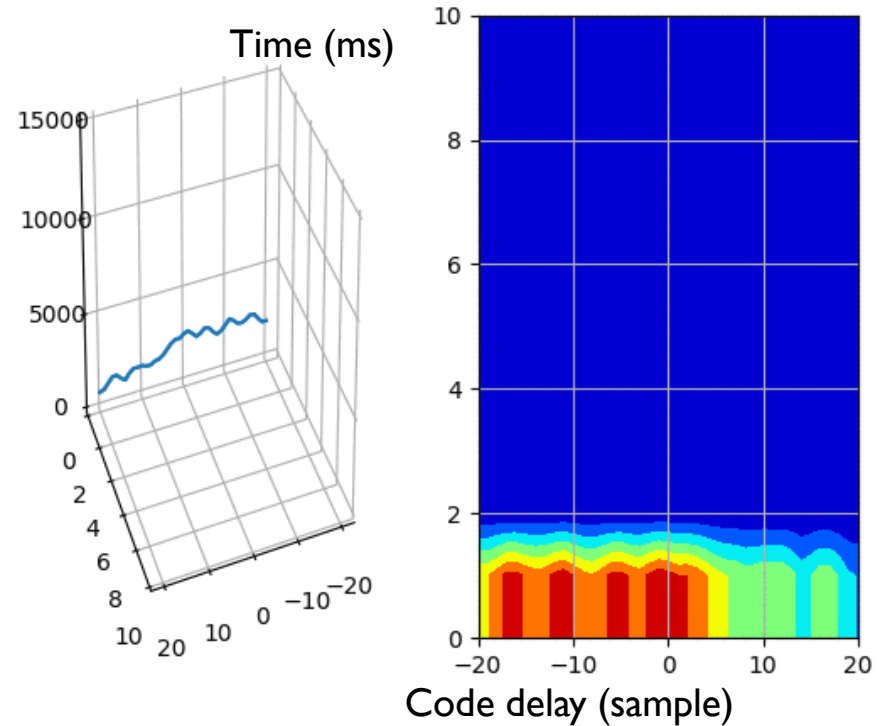


# Idea – using Convolutional NN –

LOS



NLOS



CNN can handle the **time-dependent NLOS** features !

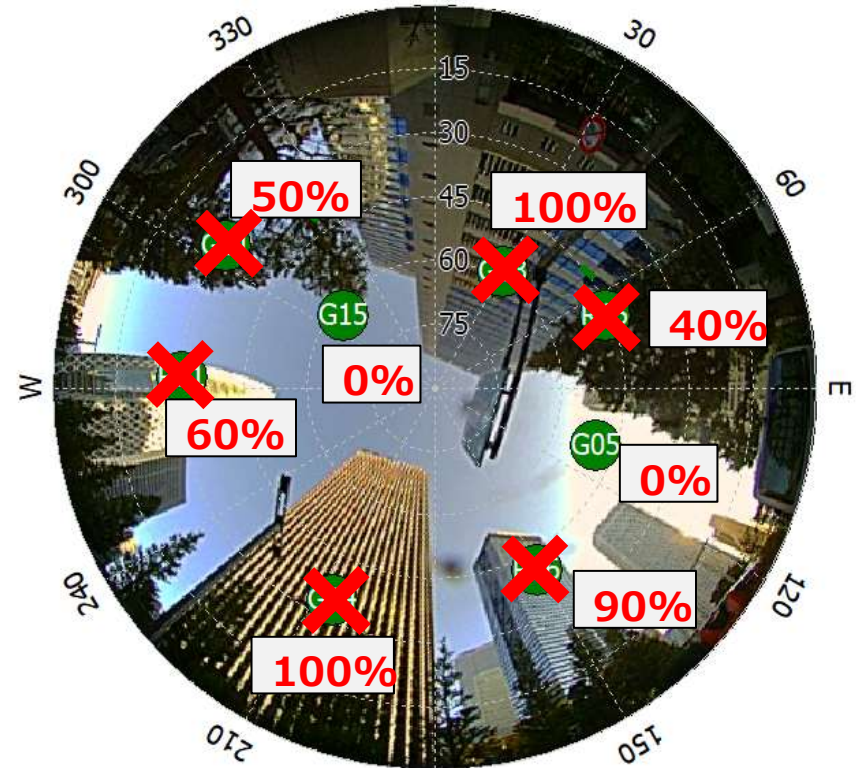


# Idea – using NLOS probability -

- Binary calcification of LOS/NLOS decrease positioning availability...



- Weighting pseudorange observations of GNSS using the probability of NLOS estimated by CNN

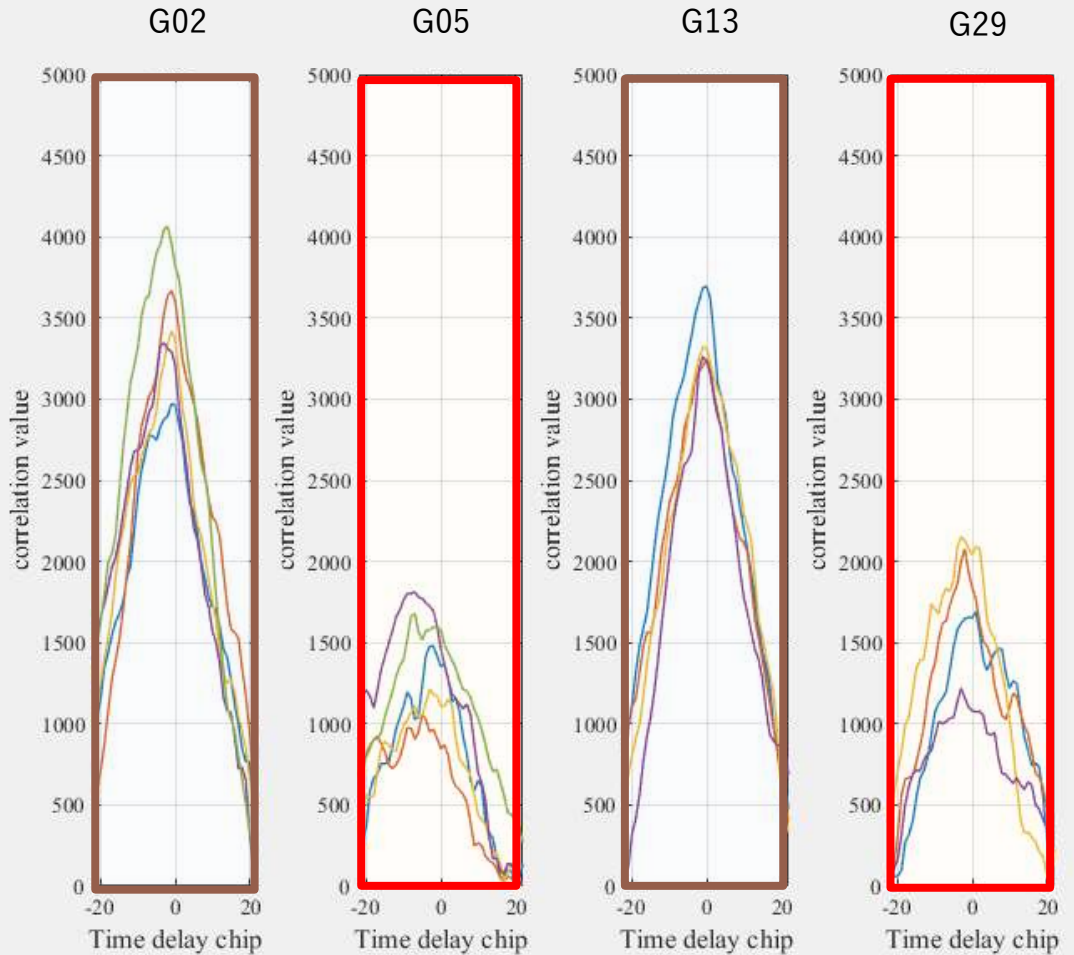
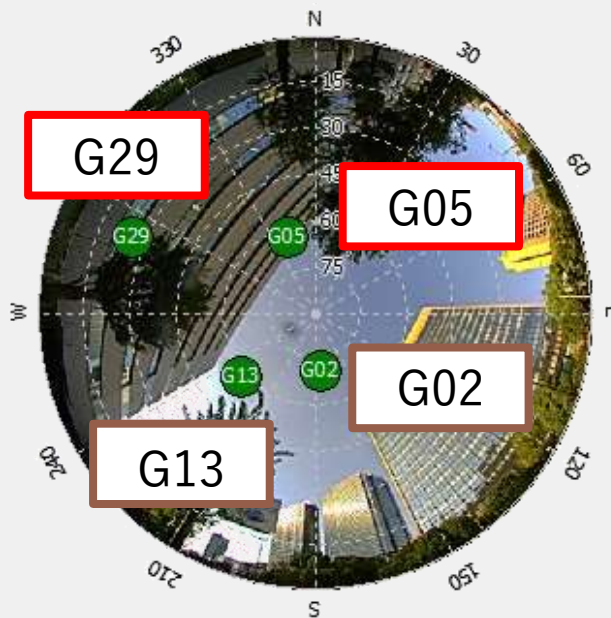


We can improve **positioning accuracy and availability** !

# Correlation Outputs (GPS L1C/A)

LOS:

NLOS:

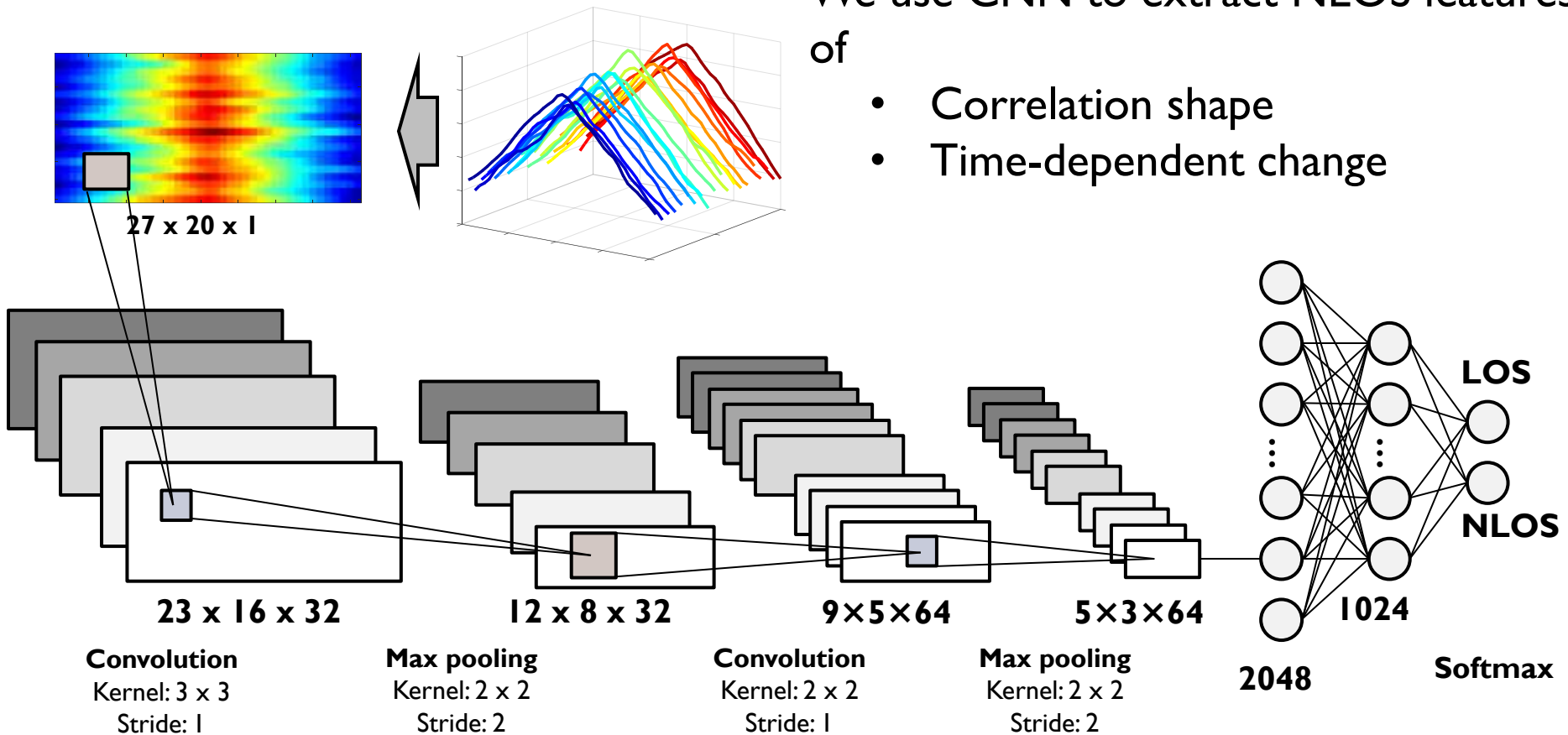


Displays 20 correlation outputs (20 ms)

# Network Configuration

We use CNN to extract NLOS features of

- Correlation shape
- Time-dependent change



# Weighting with NLOS probability

Pseudorange observations equation:

$$\Delta\rho = H\Delta x + \epsilon$$

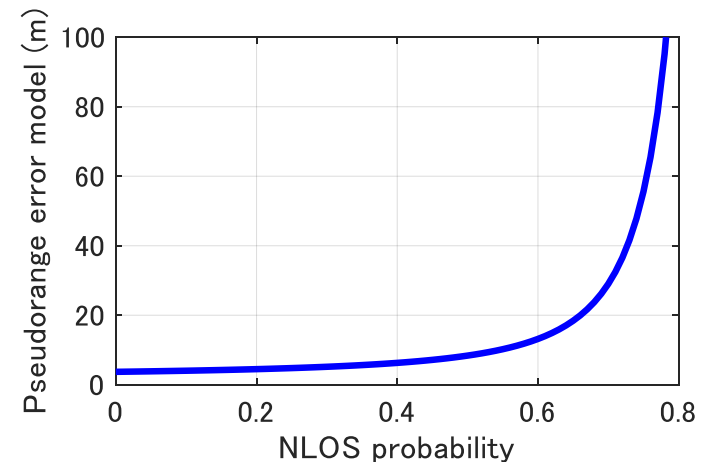
Weighted least squares method:

$$\Delta x = (H^T W H)^{-1} H^T W \Delta\rho$$

**weighting matrix:**  $W = \begin{bmatrix} 1/\sigma_1^2 & 0 & \dots \\ 0 & 1/\sigma_2^2 & \dots \\ \vdots & \vdots & \ddots \end{bmatrix}$

Pseudorange error model:

$$\sigma = \beta + \exp\left(\frac{1}{(1 - p_{NLOS})}\right)$$



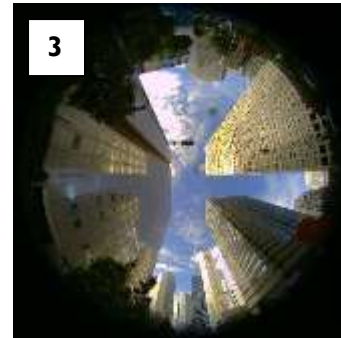
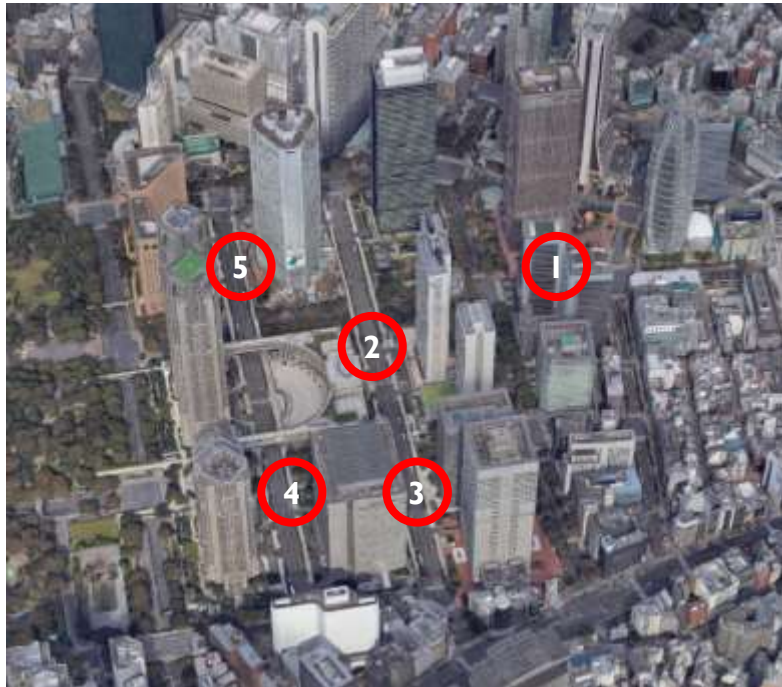
# Evaluation

## Evaluation of proposed technique in **urban environment**

Location: Shinjuku area in Japan (5 locations)

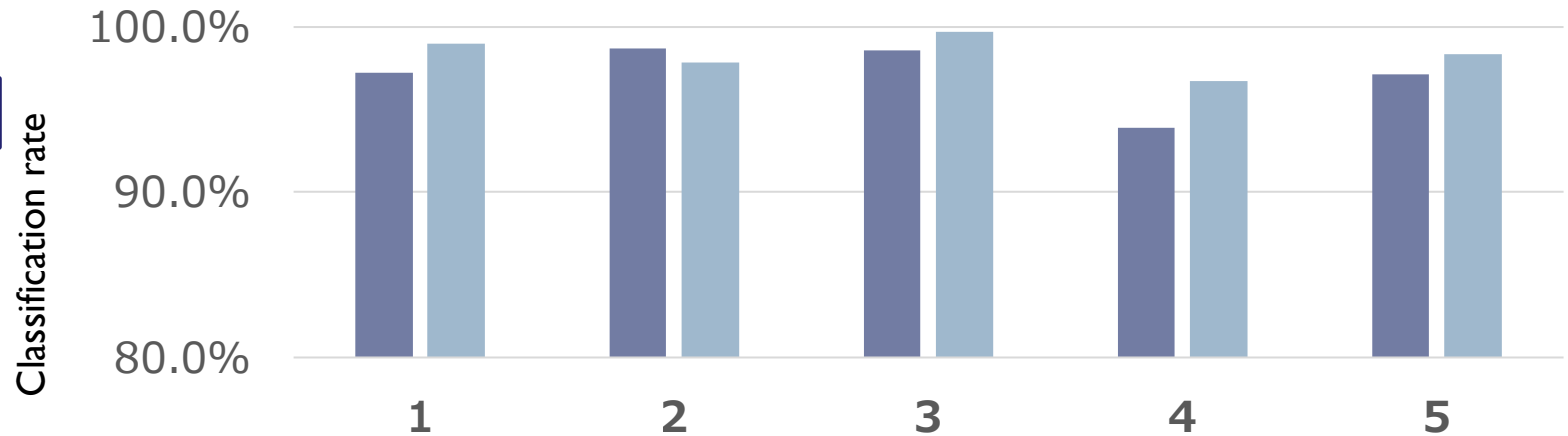
We compare classification performance of **NN** and **CNN**

- Reference NLOS signals are manually determined from fish-eye image

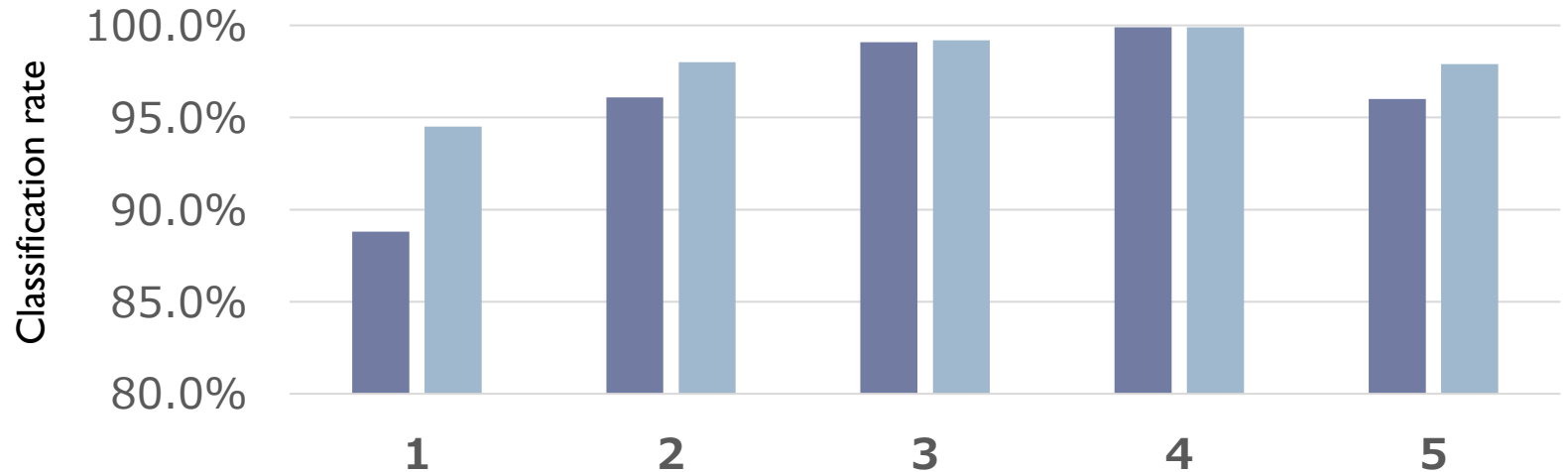


# Classification Results

LOS



NLOS



**97.9 %** NLOS signals were **correctly classified as** NLOS signals !



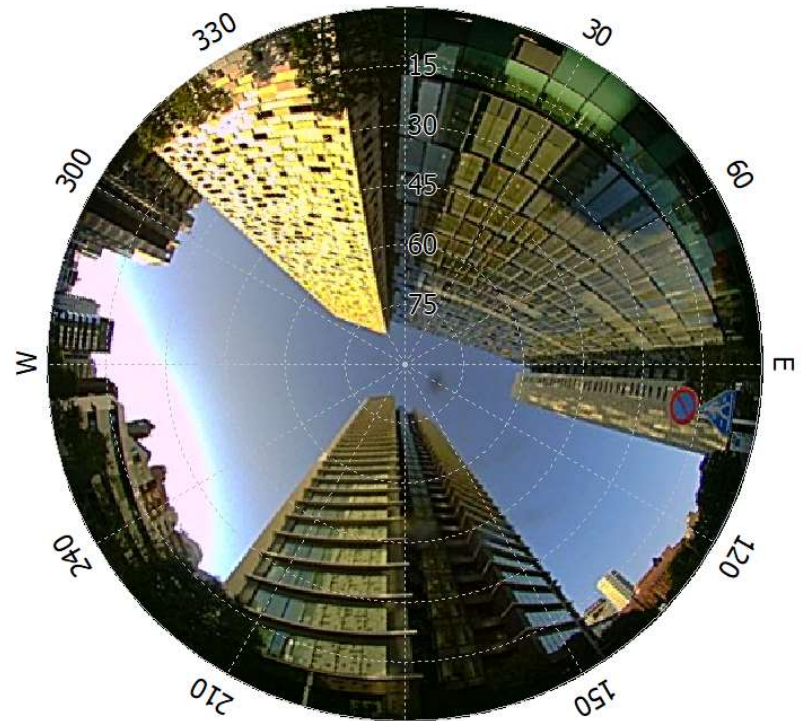
# Positioning Test

## Evaluation of proposed technique in **urban environment**

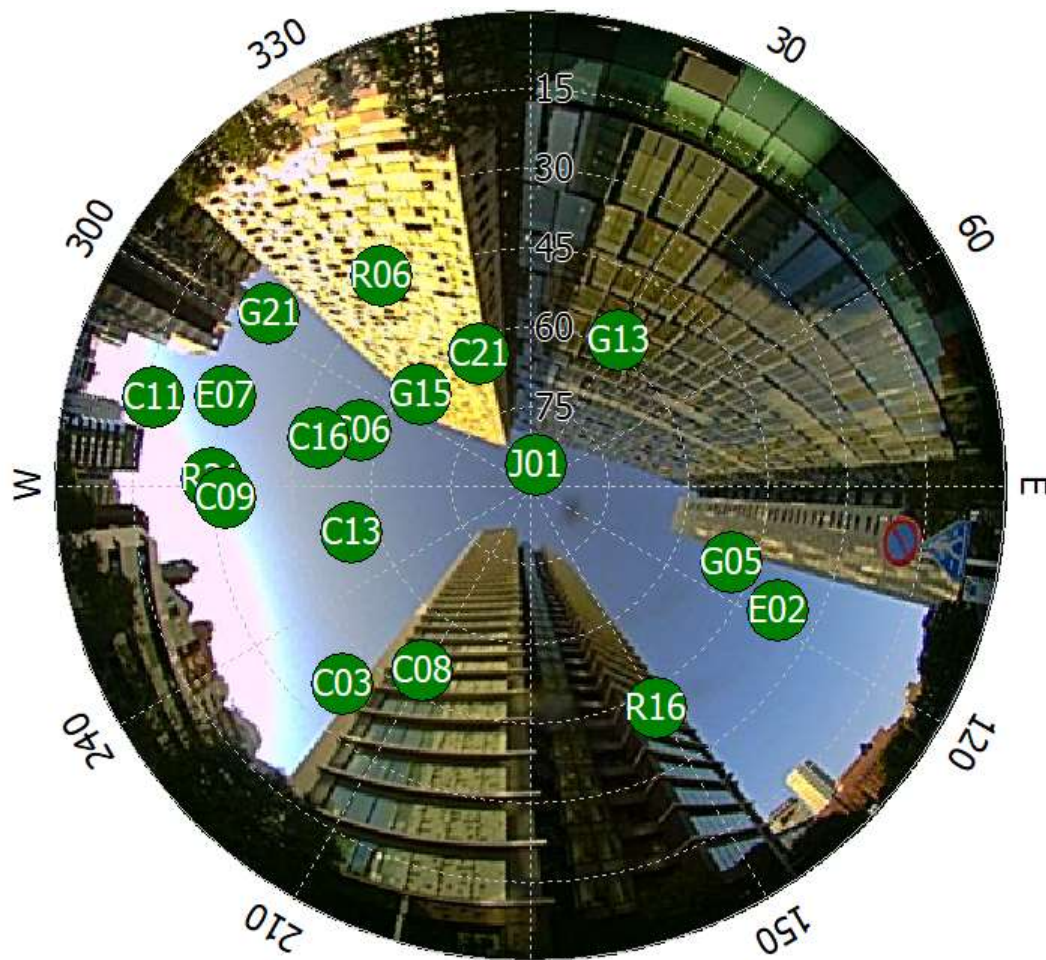
Location: Shinjuku area in Japan

Method: Single Point Positioning (RTKLIB)

SNR mask: 35 dB-Hz, Elevation mask:  $15^\circ$



# Classification Result



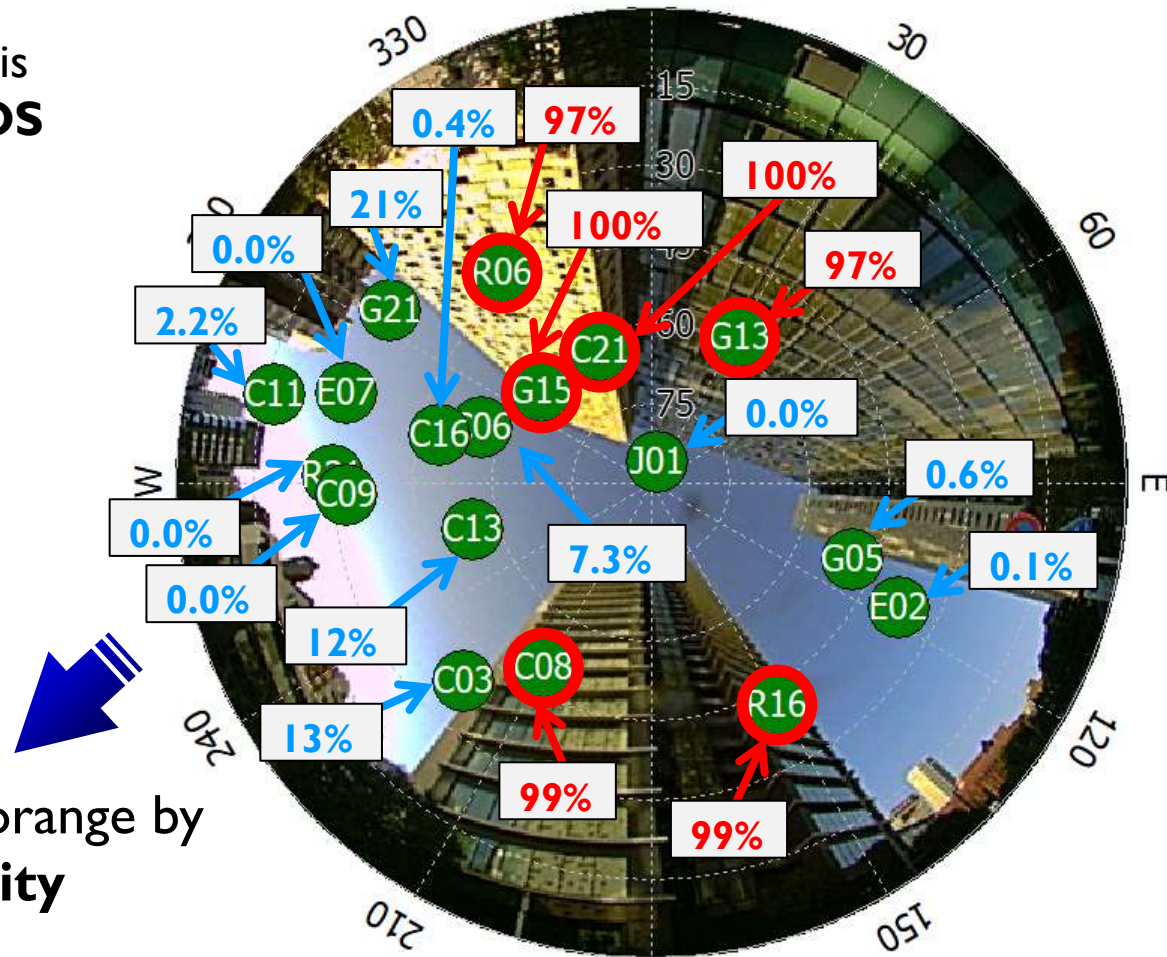


# Classification Result

## Satellite visibility is determined by NLOS probability

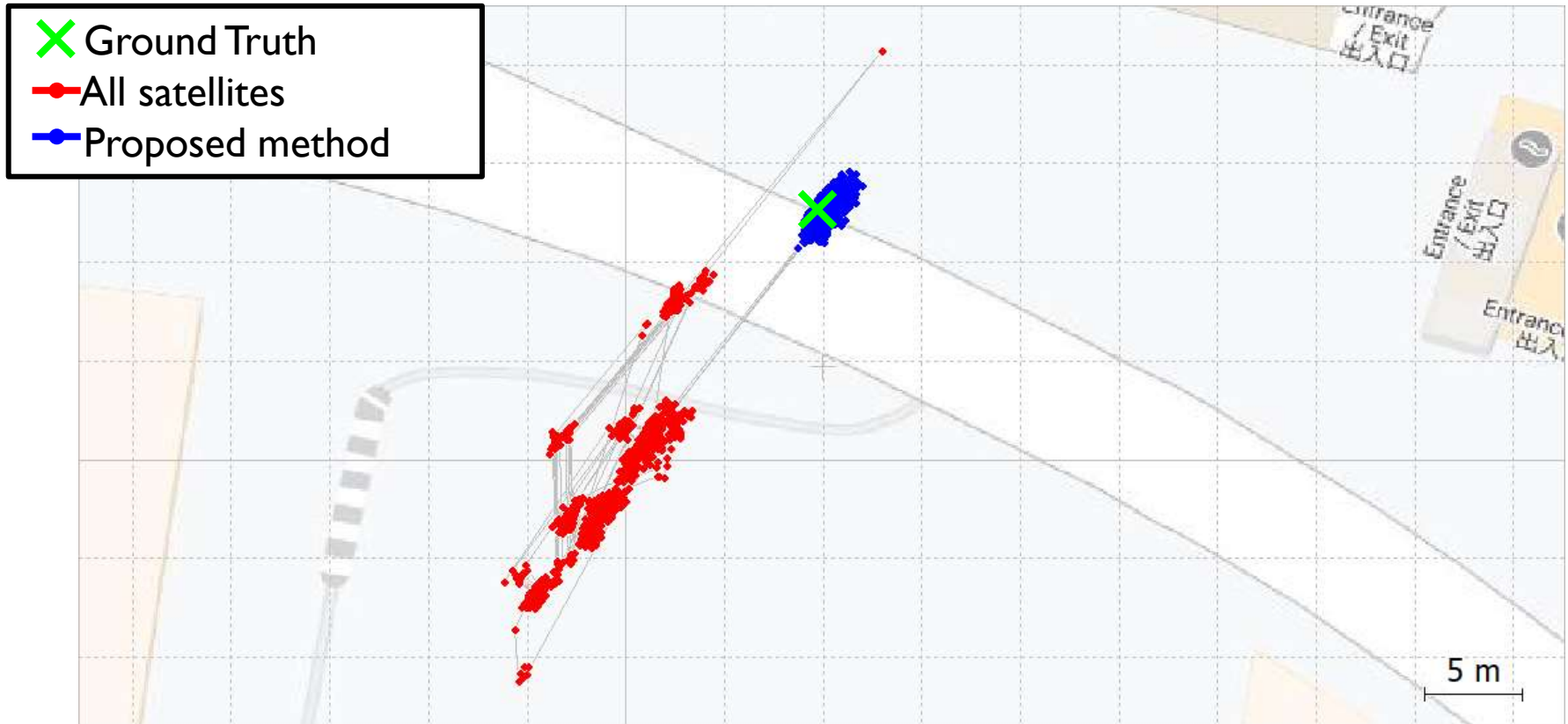
## >50% NLOS

## <=50% LOS



## Weighting pseudorange by NLOS probability

# Positioning Result



➡ Horizontal root mean square error was reduced from **34.1 m** to **1.6 m** !

# Conclusion

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- ▶ GNSS receiver is an aggregation of technologies from various fields
- ▶ By modifying the GNSS signal processing, a greater variety of ideas can be used than if only the normal receiver outputs were available
- ▶ The contribution of software GNSS receivers is especially important for the GNSS field, where new satellites and signals are increasing
- ▶ Software receiver technology is extremely useful for research