

GNSS basic2

Tokyo University of Marine Science and Technology
Nobuaki Kubo

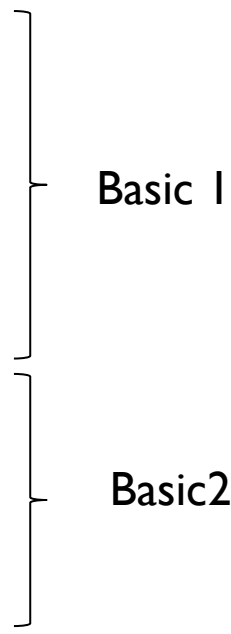
主な引用先

- ▶ 精説GPS(測位航法学会)
- ▶ よくわかる衛星測位と位置情報(日刊工業新聞社)
- ▶ 内閣府準天頂衛星のHP
- ▶ 高須様のHP
- ▶ Dr. Feng-Yu Chu氏の資料

- ▶ 上記以外は個別に記載



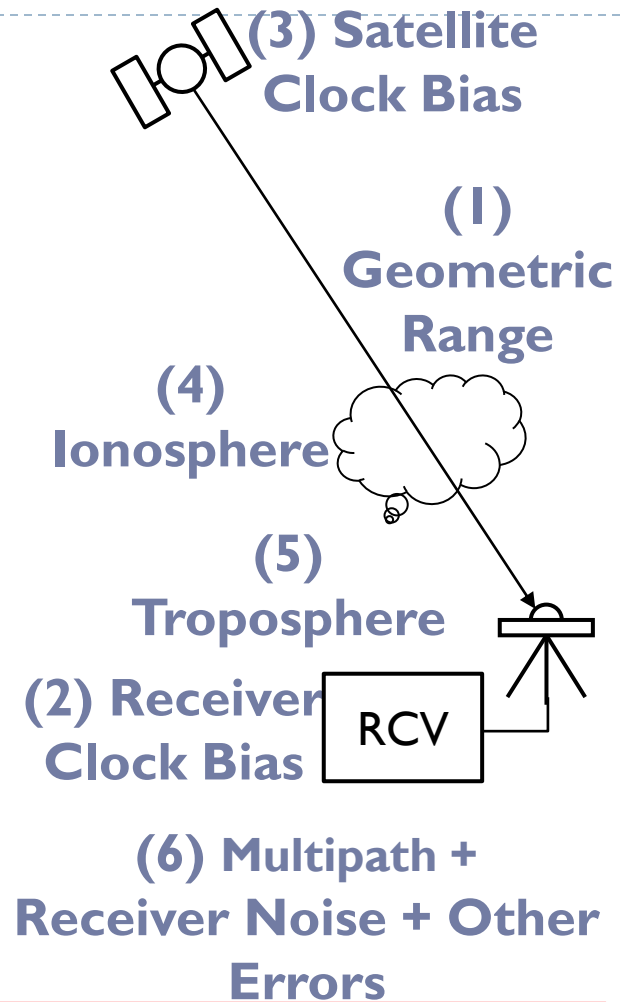
Contents

- Introduction
 - Coordinate Systems
 - Satellite Position
 - Measurements and Errors
 - Calculating Position
 - Improved Position (DGNSS, RTK, PPP)
 - QZSS
- 
- The diagram consists of two vertical curly braces on the right side of the list. The top brace groups the first four items (Introduction, Coordinate Systems, Satellite Position, and Measurements and Errors) and is labeled 'Basic I'. The bottom brace groups the last three items (Calculating Position, Improved Position (DGNSS, RTK, PPP), and QZSS) and is labeled 'Basic2'.
- Basic I
- Basic2

79問の確認用のGoogleフォームでの4択問題があります
70問以上正解してください

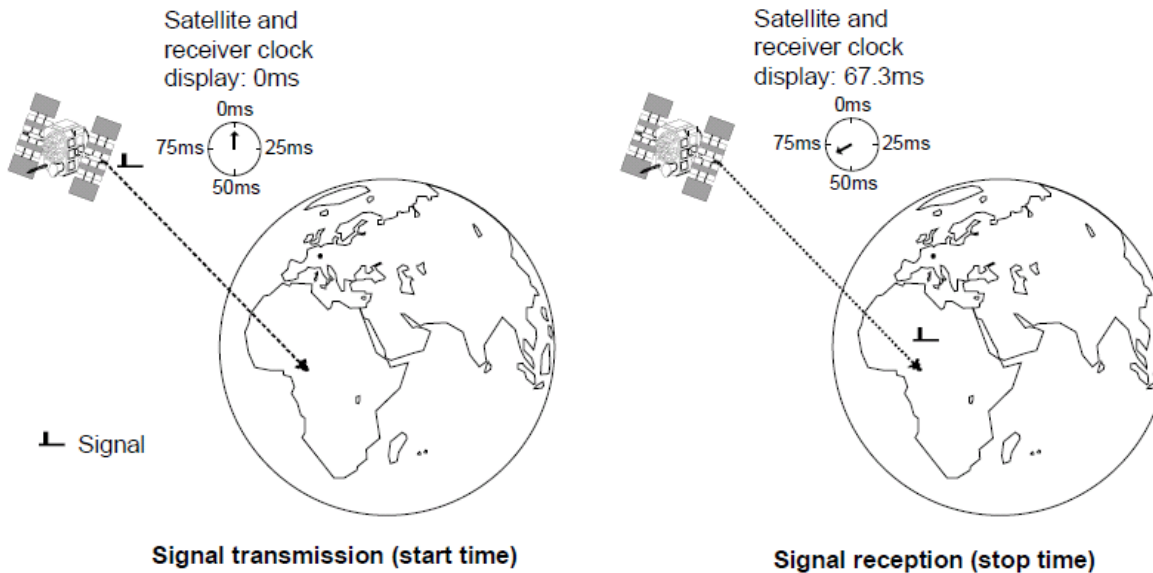
Pseudo-range Model

$$\begin{aligned}
 P_r^s &\equiv c\tau \\
 &= c(\bar{t}_r - \bar{t}^s) \\
 &= c((t_r + dt) - (t^s + dT^s)) + \varepsilon_p \\
 &= c(t_r - t^s) + c(dt_r - dT^s) + \varepsilon_p \\
 &= (\rho_r^s + I_r^s + T_r^s) + c(dt_r - dT^s) + \varepsilon_p \\
 &= \underbrace{\rho_r^s}_{(1)} + \underbrace{c(dt_r)}_{(2)} - \underbrace{dT^s}_{(3)} + \underbrace{I_r^s}_{(4)} + \underbrace{T_r^s}_{(5)} + \underbrace{\varepsilon_p}_{(6)}
 \end{aligned}$$

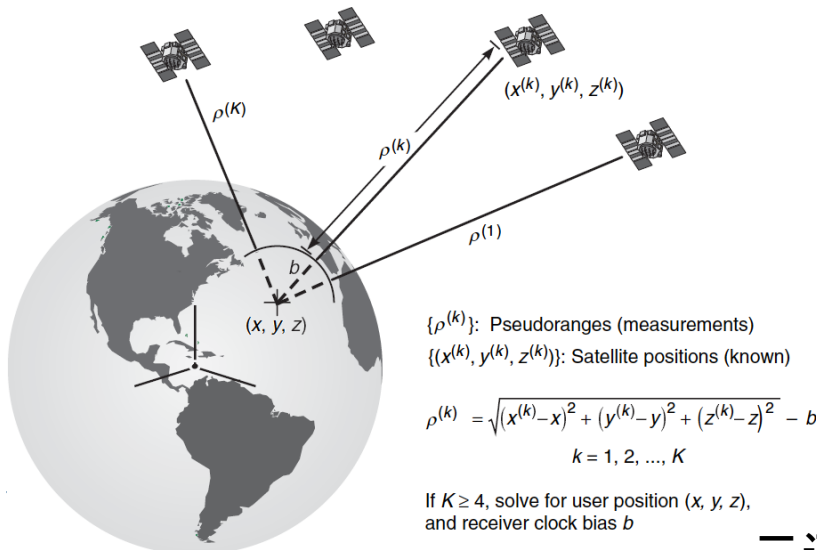


The reason why we call “pseudo-range” is from (2).
How many unknown parameter do we have ?

x, y, z, receiver clock offset



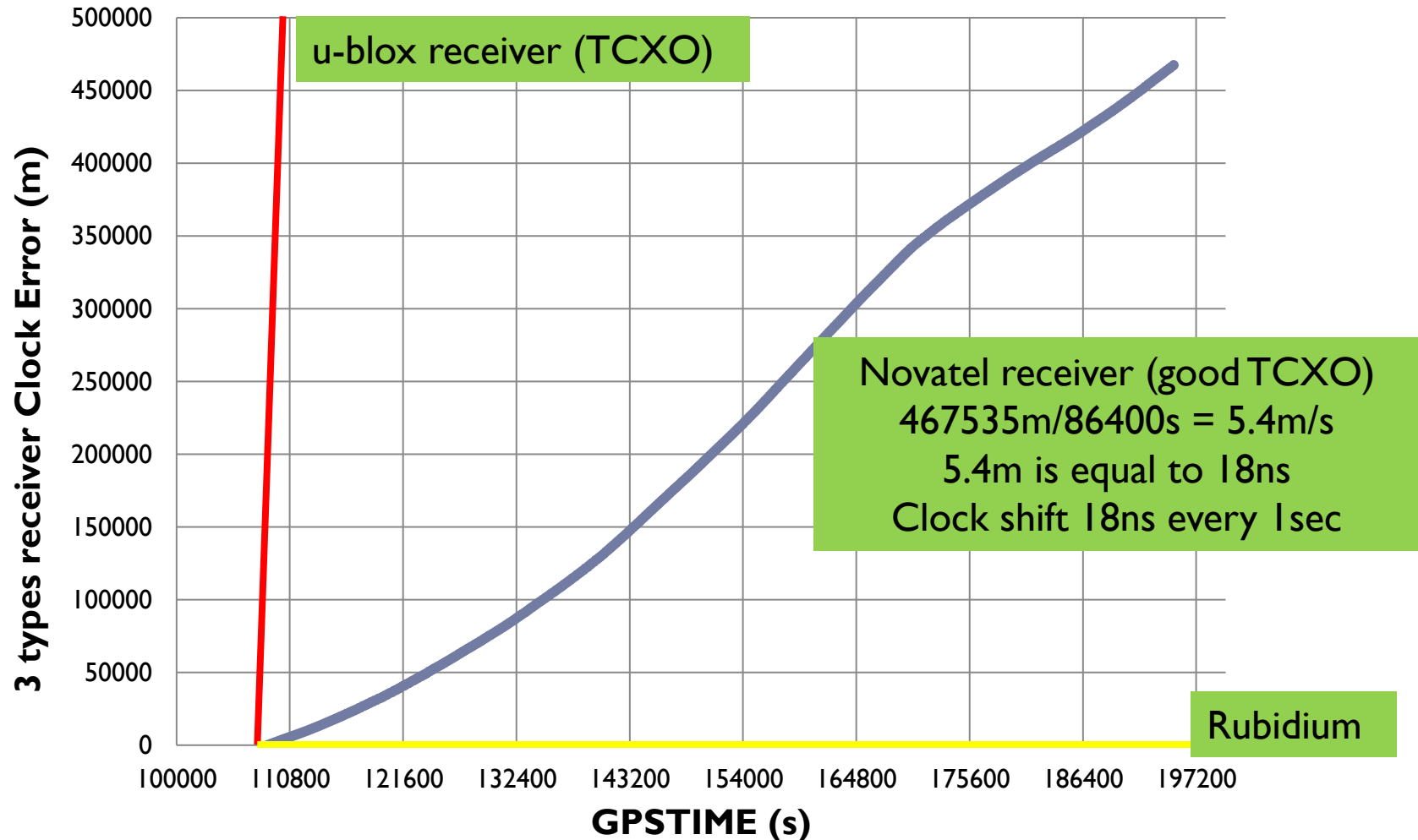
- ▶ Satellite clock is corrected using navigation data.
- ▶ Fortunately, receiver clock offset is **same** for all satellites.
- ▶ Therefore, unknown variables should be solved are **x, y, z** and **receiver clock offset**.



クロック推定精度は位置の精度とほぼ同等→3mで10ns程度

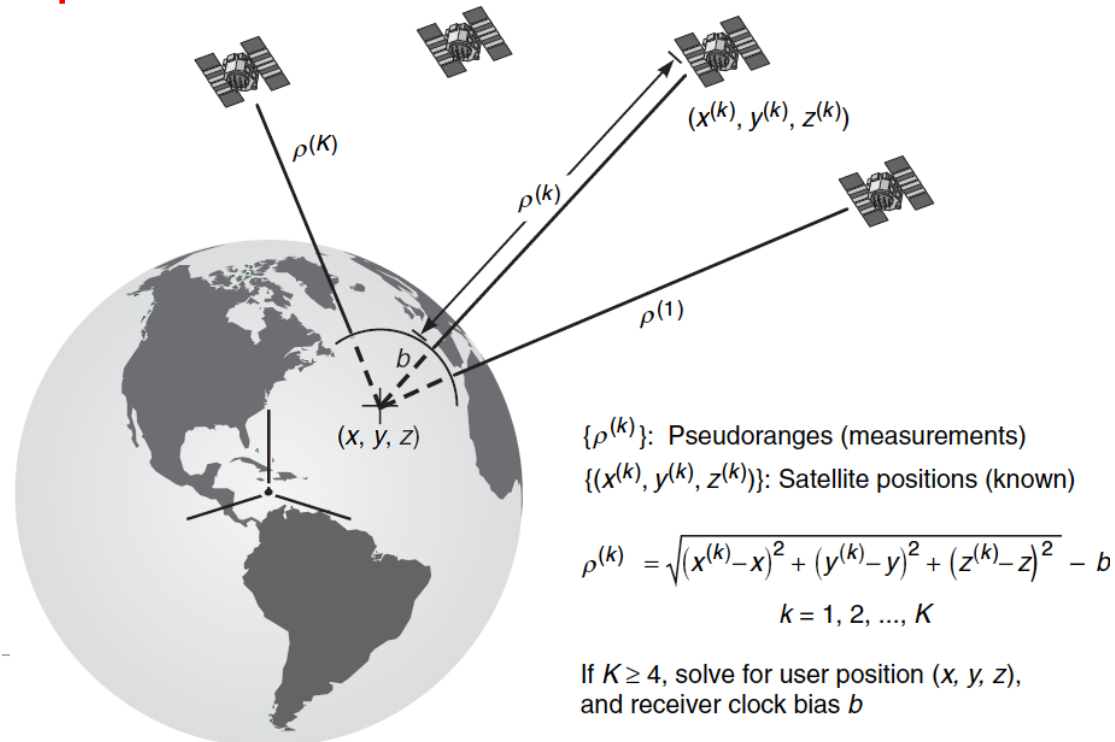
What is receiver clock offset ?

Receiver clock offset is coproduct of single positioning



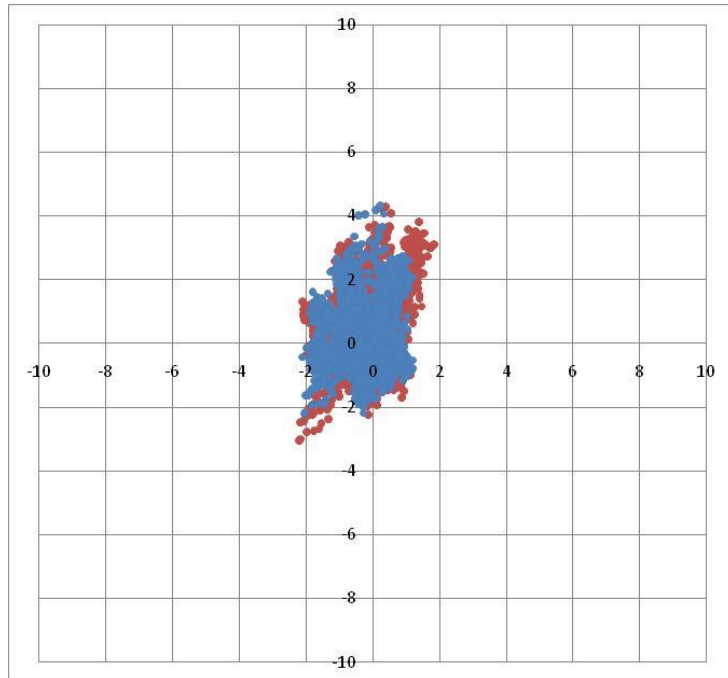
Single Point Positioning

- ▶ Using the code measurements only to compute the absolute position of the receiver in the ECEF system.
- ▶ Satellite clock can be corrected using navigation data.
- ▶ Fortunately, receiver clock offset is same for all satellites.
- Other measurement errors are assumed negligible, being absorbed in parameter of the receiver clock error.
- ▶ Therefore, there are 4 unknown parameters here. They are three range components x, y, z and the receiver clock error.
- LSE is used to compute the unknown parameters.

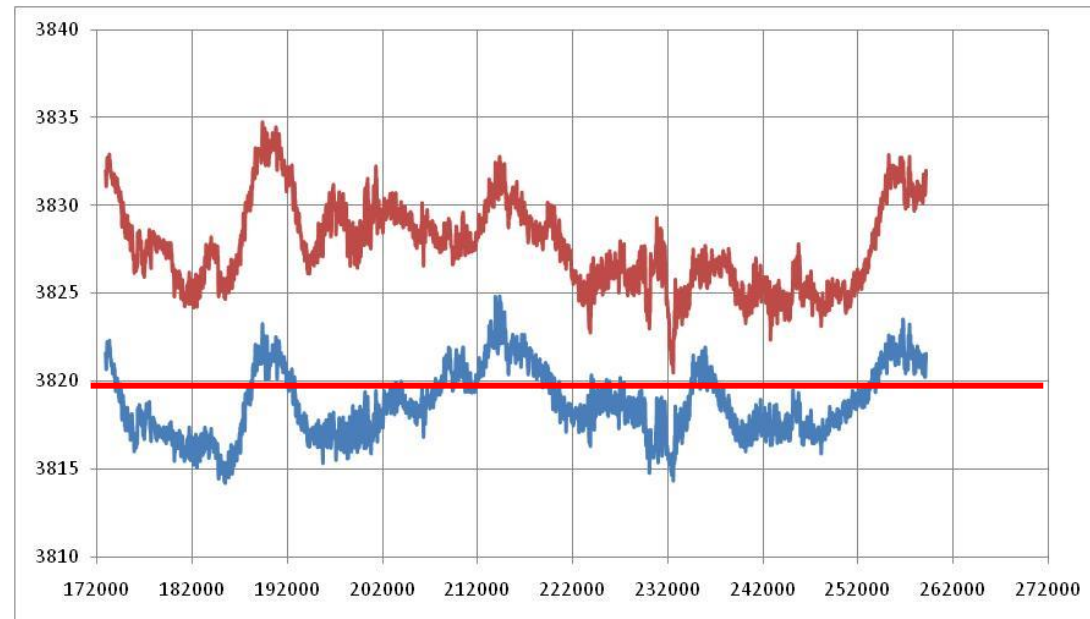


Single point positioning at Mt. Fuji

(6/1/2010)



Horizontal Errors (m)



Height Errors (m)

Blue : Stand Alone Positioning

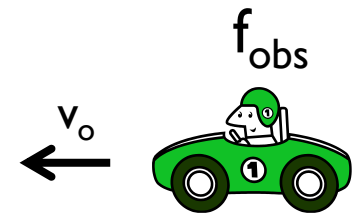
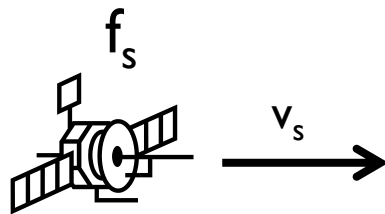
Red : Stand Alone Positioning without Iono and Tropo Estimation

Demonstration of SPP

- ▶ Excel is used.



Doppler Effect



One dimension is assumed. Right direction is positive.

+

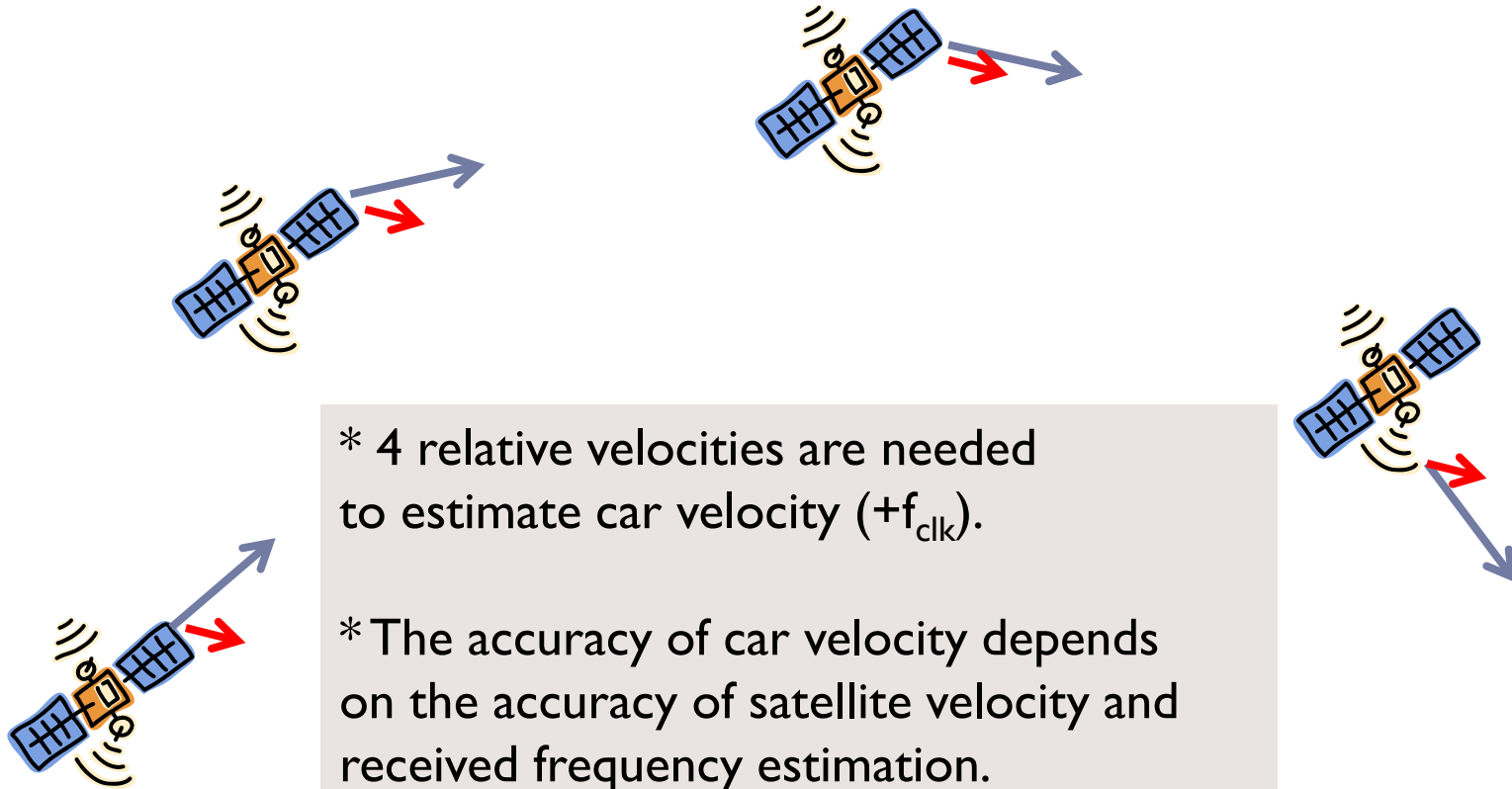
- ▶ Receiver is set in the car.
- ▶ Received frequency is
- ▶ “cs” is speed of light.
- ▶ Doppler frequency “ f_D ” is equal to “ $f_{obs} - f_{source}$ ”
- ▶ FLL (frequency lock loop) tries to estimate “ f_D ”.
- ▶ Once we can estimate “ f_D ”, “ v_o ” can be resolved.

$$f_{obs} = f_s \frac{cs - v_o}{cs - v_s}$$

Velocity Estimation

- ▶ Velocity estimation in GPS is just same as shown in the previous slide.
- ▶ The differences are as follows.
- * **3 dimension velocity (v_x, v_y, v_z) have to be estimated.**
- * **Frequency in the receiver is based on on-board clock.**
- ▶ 4 unknown variables (v_x, v_y, v_z, f_{clk}) have to be estimated using at least 4 visible satellites. DOP is also important.
- ▶ Velocity estimation is same as position estimation.

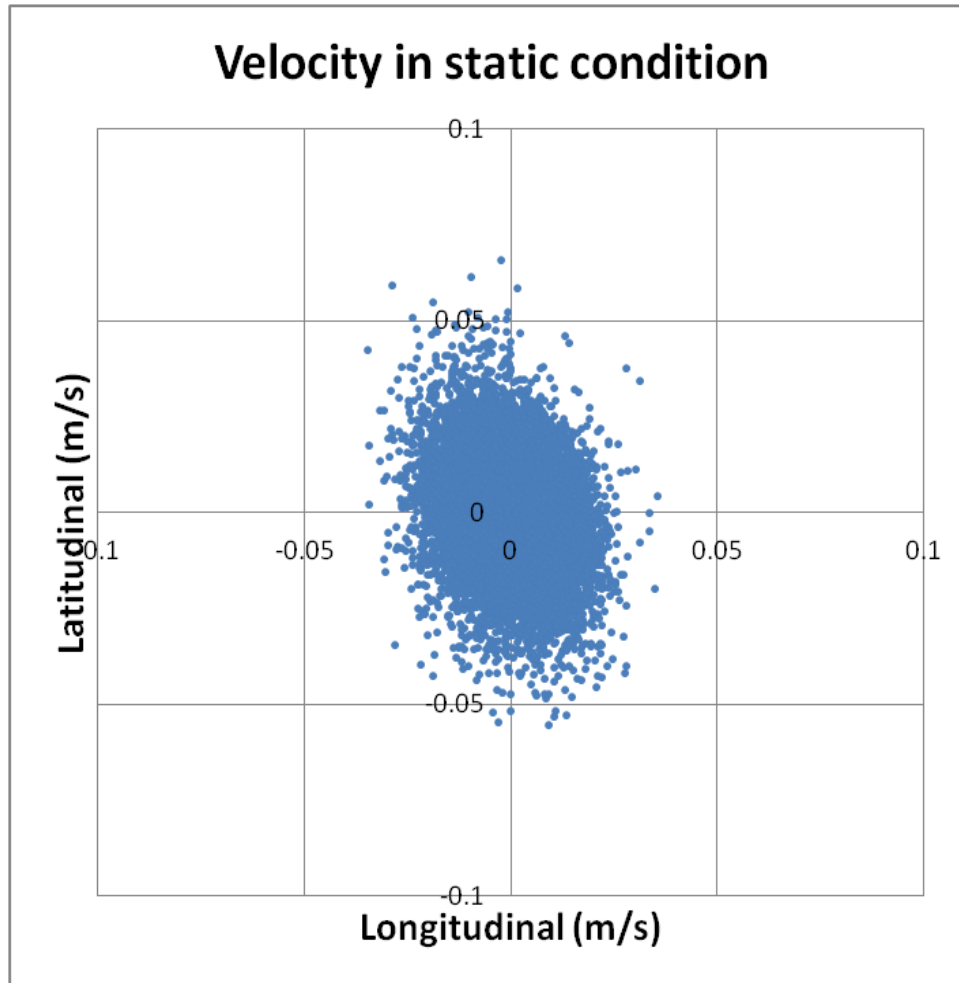
Image of Velocity Estimation



$$(\mathbf{v}_x, \mathbf{v}_y, \mathbf{v}_z) + \mathbf{f}_{\text{clk}}$$



Performance of GPS based Velocity



std = 1.6 cm/s

Accuracy in terms of frequency

GPS L1 wavelength = 19cm

1Hz : 19cm

0.1Hz : 1.9cm

Accuracy in terms of satellite velocity

$sv_vel[t] = (sv_vel[t+1] - sv_vel[t-1]) / 2$

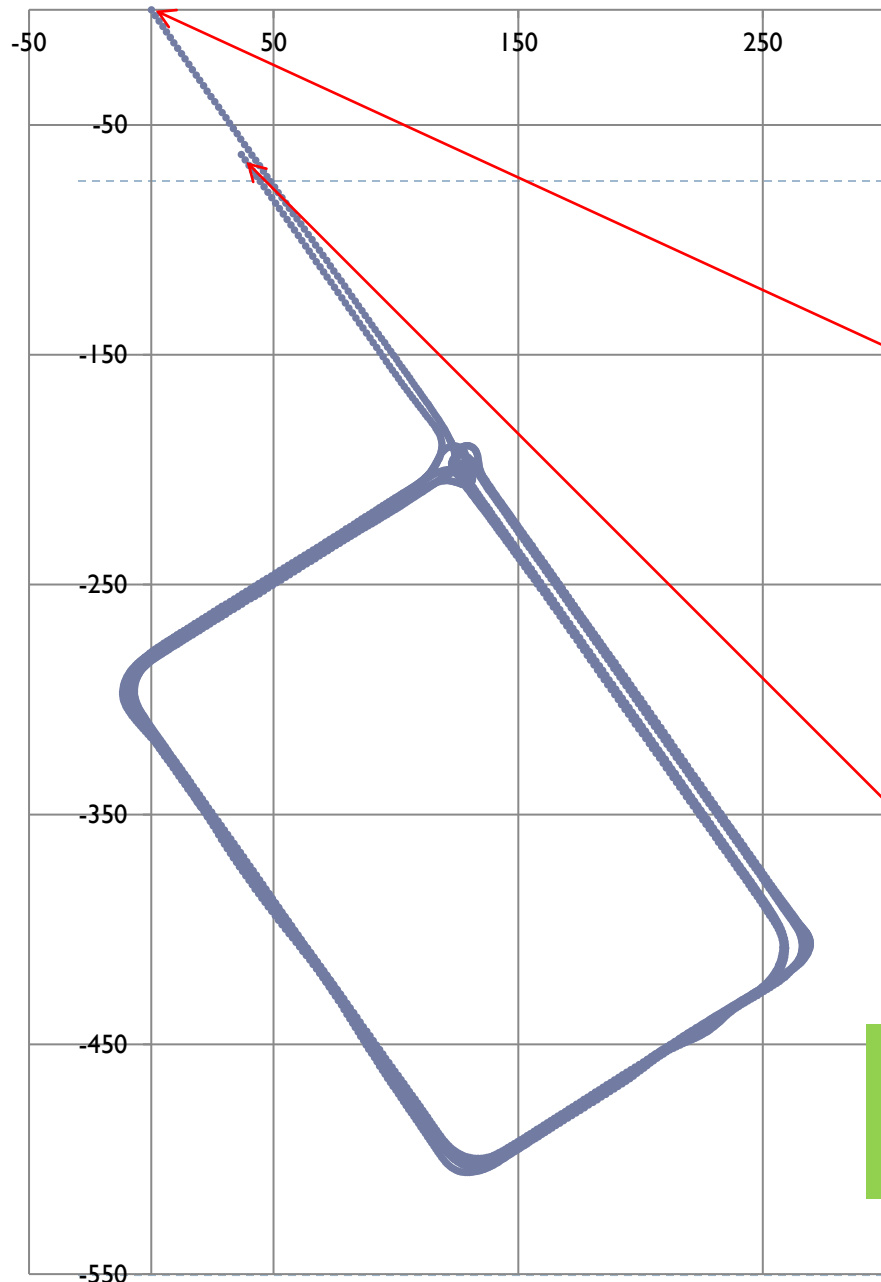
based on ephemeris parameters

Accuracy is quite good.

Moving Platform (Koto-ku Ariake)

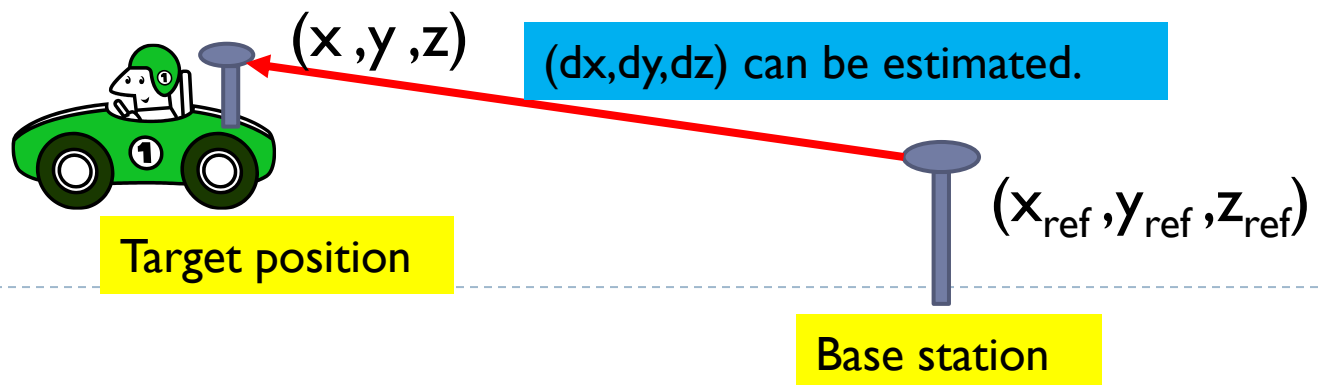
- Origination : 0,0
- Velocity was accumulated.
- Data Rate : 5Hz
- Period : 650 sec
- Receiver : NovAtel OEM6
- Left and right rounds : 6 times
- End point : 36.76m,-62.91m
- RTK : 35.75m,-65.18m

Deviation after 11 minutes velocity accumulation was about 2-3 m.



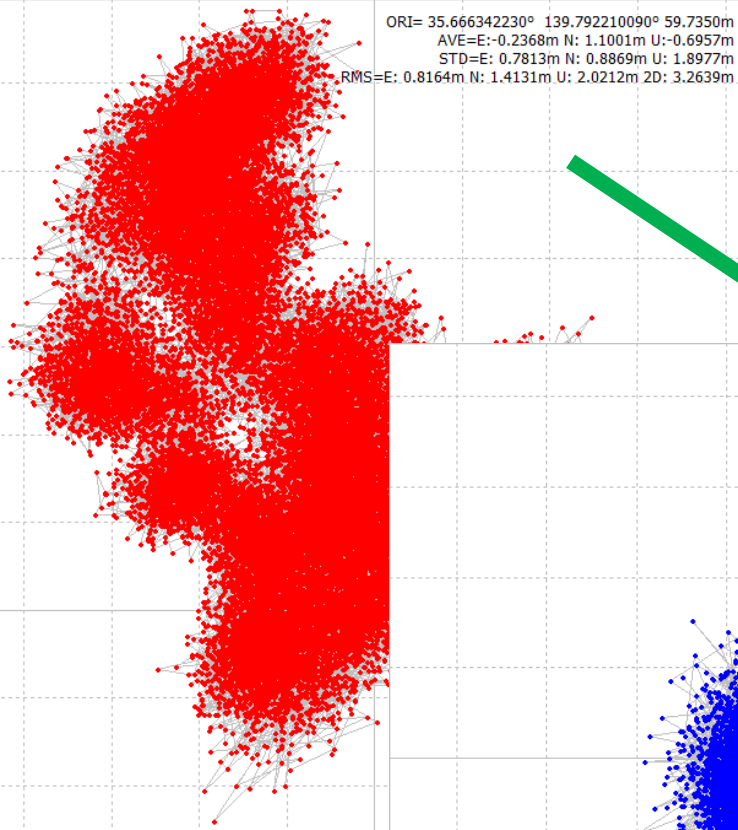
Improved GPS (relative positioning)

- ▶ **DGPS** and RTK are powerful method for error mitigation.
- ▶ DGPS uses the fact that the **most of error sources change slowly** in the time domain if the distance between reference and user is approx. within 100km.
- ▶ Please remember **that differential technique provides only vector solution** from base station to the target position → precise position of base station should be prepared.

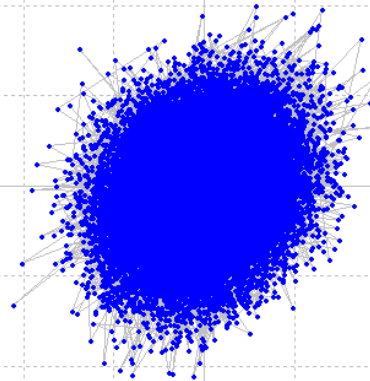


RTK performance

12h, rooftop, our building



ORI= 35.666342230° 139.792210090° 59.7350m
AVE=E:-0.0160m N:-0.0395m U: 0.1039m
STD=E: 0.2253m N: 0.2461m U: 0.5594m
RMS=E: 0.2259m N: 0.2493m U: 0.5689m 2D: 0.6728m



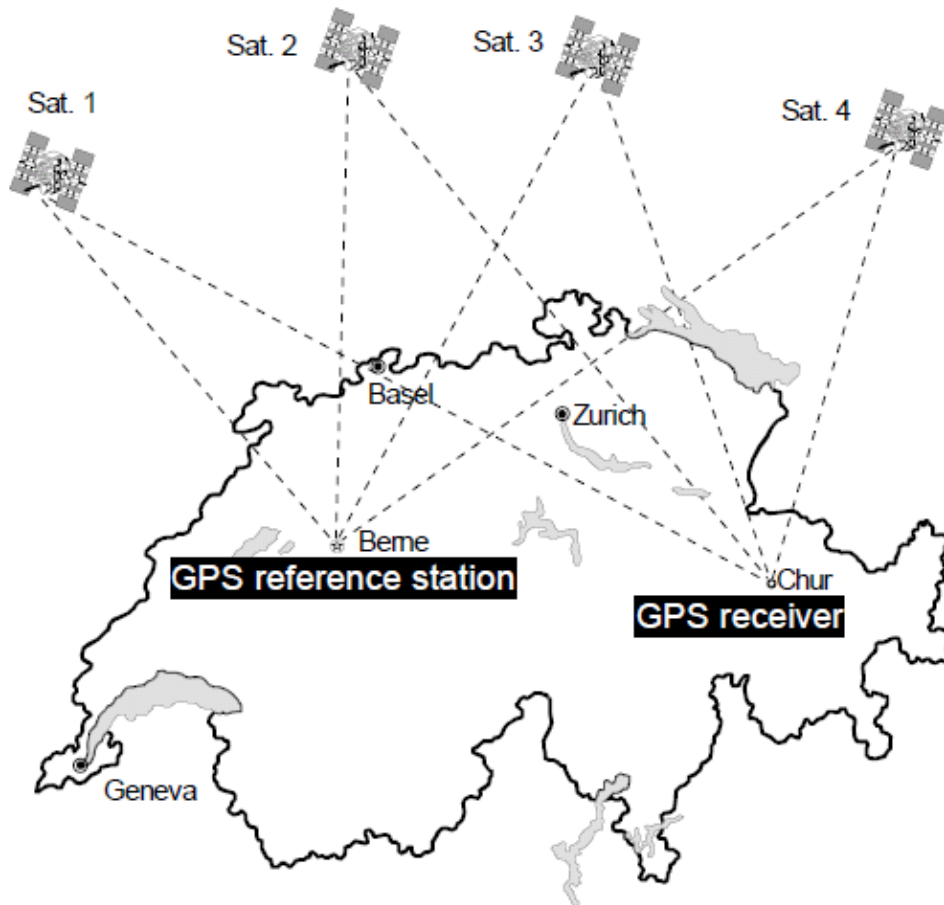
ORI= 35.666342230° 139.792210090° 59.7350m
AVE=E:-0.0000m N: 0.0002m U: 0.0028m
STD=E: 0.0017m N: 0.0015m U: 0.0039m
RMS=E: 0.0017m N: 0.0016m U: 0.0048m 2D: 0.0046m

RTK : mm level

Same scale

50 cm

Image of DGPS



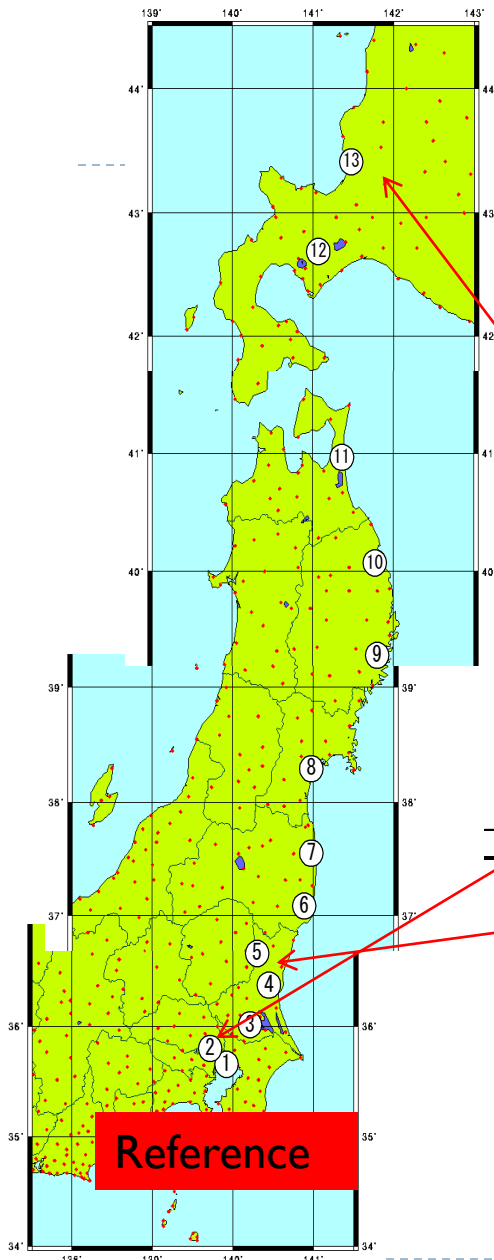
- Determination of the correction values at the reference station
- **Transmission of the correction values** from the reference station to the GPS user
- **Compensation** for the determined pseudo-ranges to correct the calculated position of the GPS user

$$\text{Correction [prn]} = \text{Pseudo-range[prn]} - \text{True-range [prn]}$$

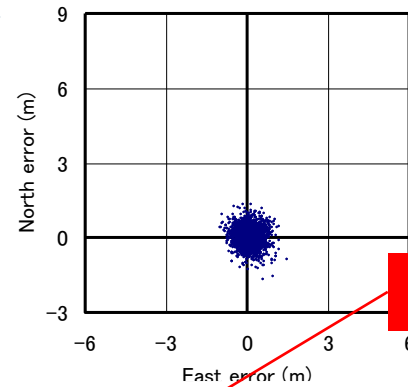
DGPS mitigates ...

Source	Potential error size	Error mitigation using DGPS
Satellite clock model	2 m (rms)	0.0 m
Satellite ephemeris prediction	2 m (rms) along the LOS	0.1 m (rms)
Ionospheric delay	2-10 m (zenith) Obliquity factor 3 at 5°	0.2 m (rms)
Tropospheric delay	2.3-2.5m (zenith) Obliquity factor 10 at 5°	0.2 m (rms) + altitude effect
Multipath (open sky)	Code : 0.5-1 m Carrier : 0.5-1 cm	→
Receiver Noise	Code : 0.25-0.5 m (rms) Carrier : 1-2 mm (rms)	→

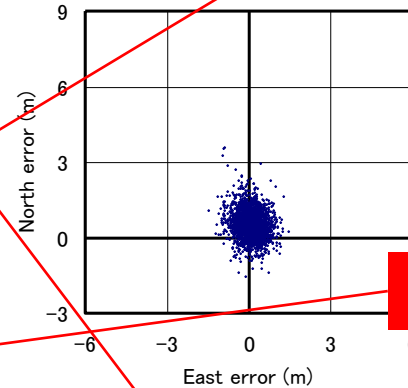
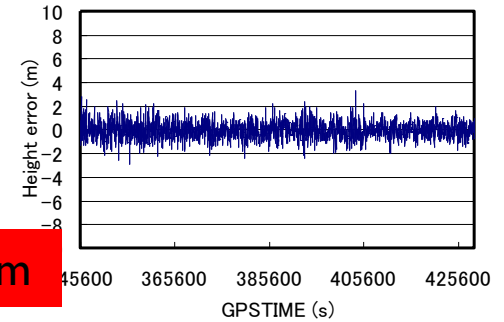
Limitation of DGPS



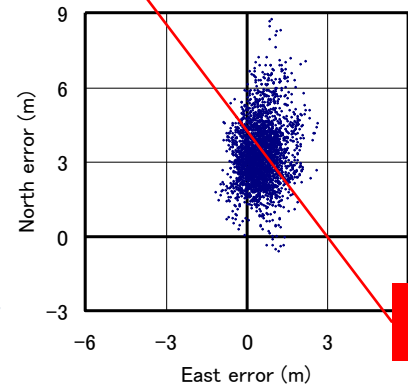
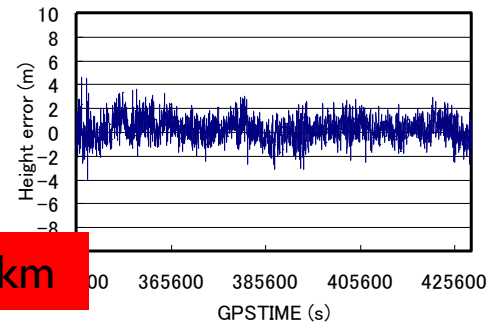
number	name	type
①	千葉市川	基準局
②	足立	未知点
③	阿見	未知点
④	水戸	未知点
⑤	大田原	未知点
⑥	いわき	未知点
⑦	小高	未知点
⑧	利府	未知点
⑨	金石	未知点
⑩	久慈	未知点
⑪	六ヶ所	未知点
⑫	大滝	未知点
⑬	厚田	未知点



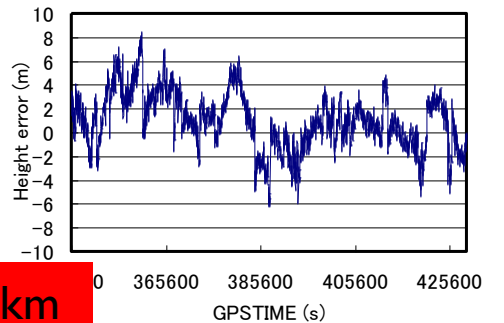
14 km



131 km



867 km



Reason for Accuracy Deterioration

- ▶ The main reason for the accuracy deterioration is atmospheric errors.
- ▶ For example, the tropospheric errors between Tokyo and Sapporo will be different (due to the difference of satellite elevation). The amount of error depends on the elevation angle.
- ▶ Normal DGPS doesn't take into the above error account at all.

RTK (Real Time Kinematic)

- The concept of **RTK** is same as **DGPS**.
- RTK uses **carrier phase measurements**. DGPS uses pseudo-range measurements.
- GPS receiver is able to measure 1/100 of wavelength of L1 frequency (19 cm).
- If you have high-end receiver, you know your position **within 1-2cm accuracy** as long as you have 5 or more LOS satellites.

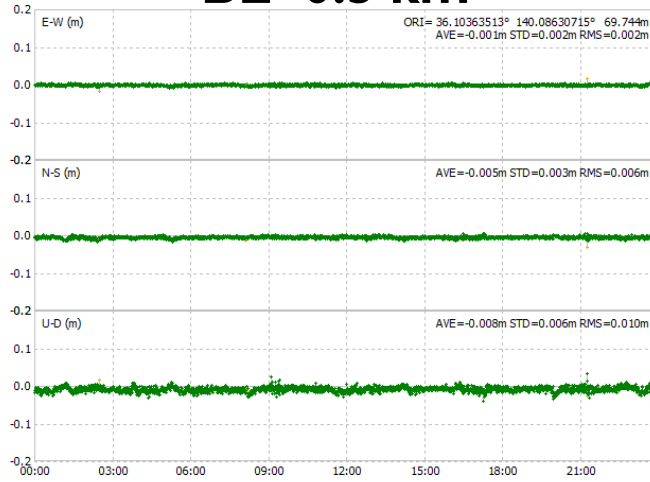
Real-time Kinematic

- **Technique with Carrier-based Relative Positioning**
 - Real-time Position of Rover Antenna
 - Transmit Reference Station Data to Rover via Comm. Link
 - **OTF** (On-the-Fly) Integer Ambiguity Resolution
 - Typical Accuracy: $1 \text{ cm} + 1 \text{ ppm} \times \text{BL RMS}$ (Horizontal)
 - Applications:
Land Survey, Construction Machine Control, Precision Agriculture etc.



Effect of Baseline Length

BL=0.3 km



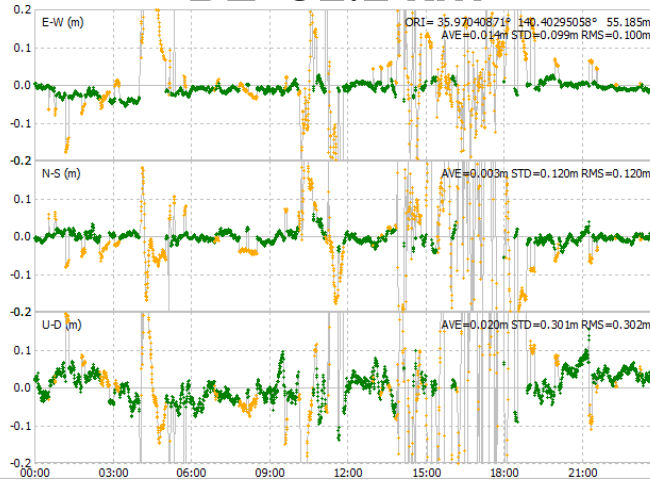
**RMS
Error:**
E: 0.2cm
N: 0.6cm
U: 1.0cm
Fix Ratio:
99.9%

BL=13.3 km



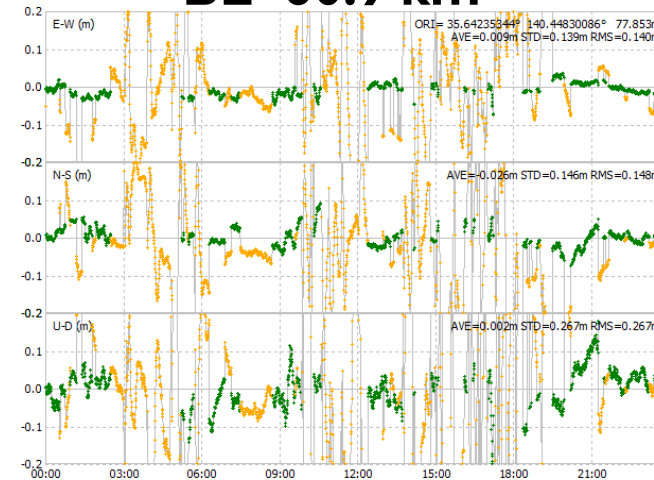
**RMS
Error:**
E: 2.2cm
N: 2.4cm
U: 10.6cm
Fix Ratio:
94.2%

BL=32.2 km



**RMS
Error:**
E: 10.0cm
N: 12.0cm
U: 30.2cm
Fix Ratio:
64.3%

BL=60.9 km



**RMS
Error:**
E: 14.0cm
N: 14.8cm
U: 26.7cm
Fix Ratio:
44.4%

(24 hr Kinematic ●: Fixed Solution ●: Float Solution)

Network RTK (NRTK)

▶ **Extension of RTK**

- ▶ RTK without User Reference Station
- ▶ Sparse Networked Reference Stations
- ▶ Correction Messages via Mobile-Phone Network
- ▶ Format: **VRS**, **FKP**, MAC, RTCM 2.3, RTCM 3.1
- ▶ Server S/W: Trimble GPSNet, GEO++ GNSMART, ...
- ▶ NTRIP Networked Transport of RTCM via Internet Protocol

▶ **NRTK Service in Japan**

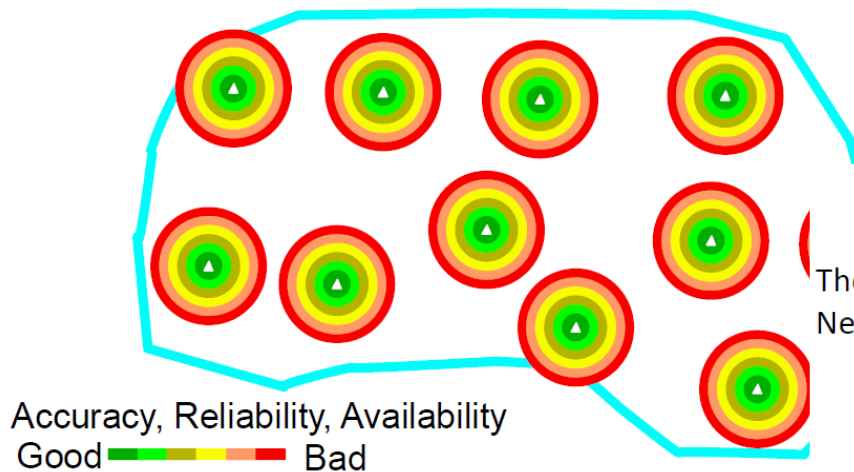
- ▶ GEONET: ~1200 Reference Stations by GSI
- ▶ NGDS, JENOBAB, Terasat



Concept of NRTK

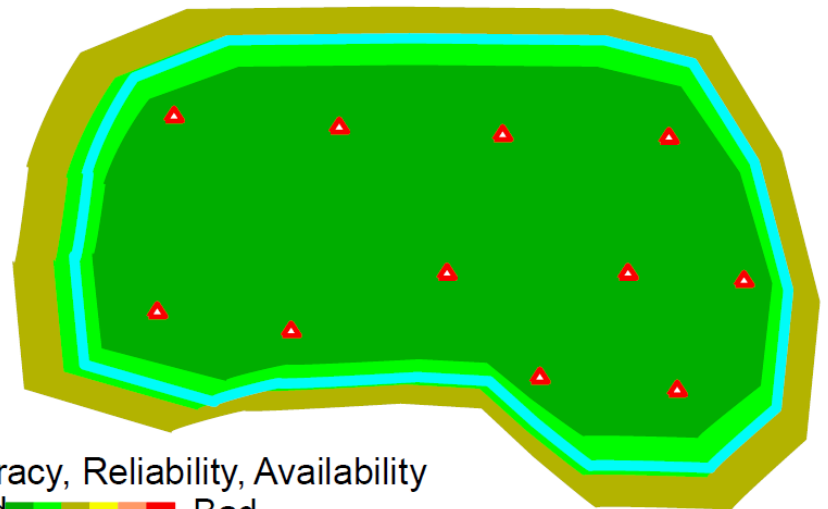
Network of Individual Reference Stations

To cover a large area with single reference stations to run RTK, we need multitude of points and still we have huge gaps between the points.

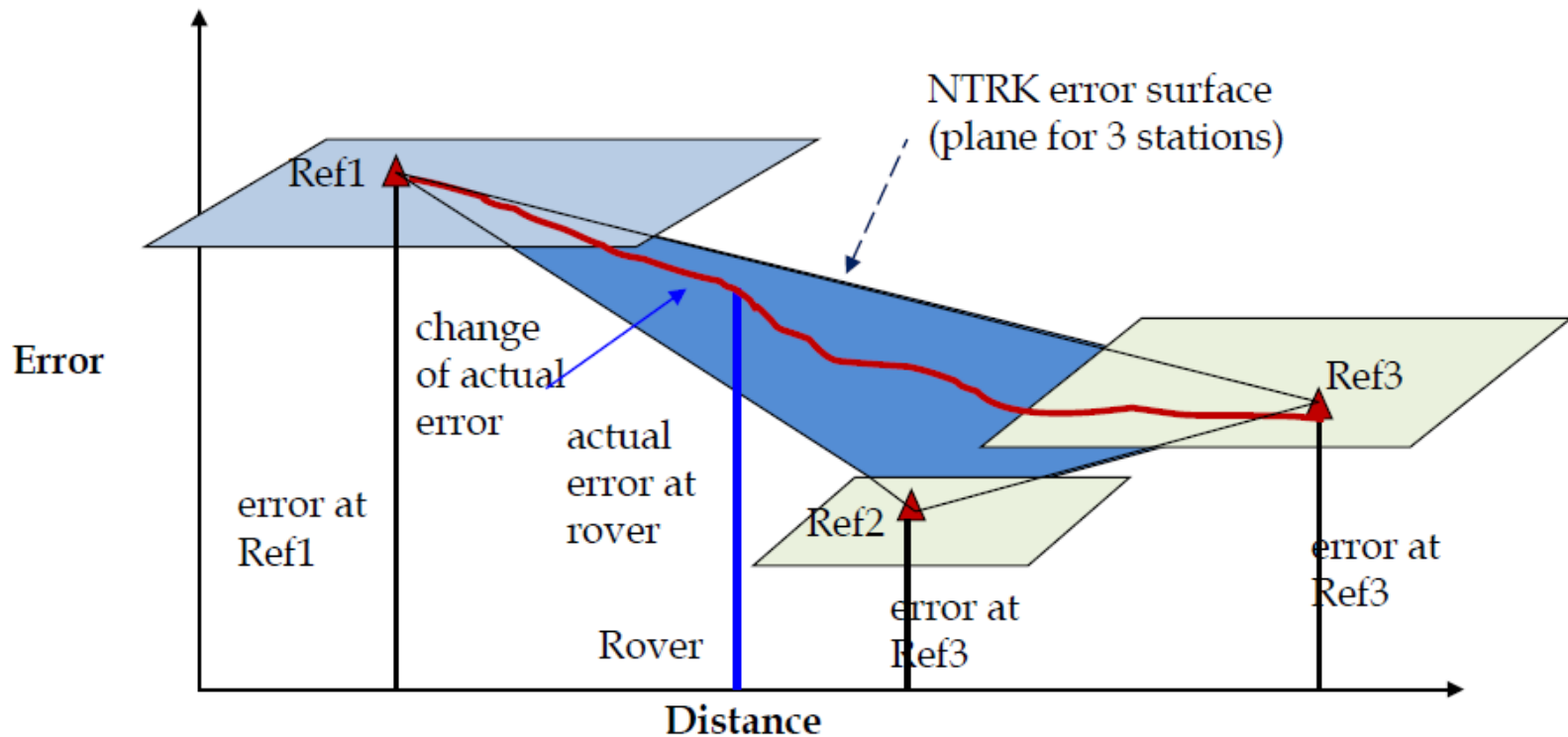


The Solution is Network RTK (NRTK)!

The same area is covered with much less number of points using the Network RTK concept. All the area is covered with no gaps.



Relationship between Errors



Several interpolation algorithms

Actual Steps of RTK

- ▶ After this summer school, please check the followings regarding the process of RTK to deepen your understanding !

1. Generating “double difference”
2. Finding “integer ambiguities”
3. Baseline processing



1. DD (Double Difference)

$$\begin{aligned}\Phi_{ub}^{ij} &\equiv \lambda((\phi_u^i - \phi_b^i) - (\phi_u^j - \phi_b^j)) \\ &= \rho_{ub}^{ij} + c(dt_{ub}^{ij} - dT_{ub}^{ij}) - I_{ub}^{ij} + T_{ub}^{ij} + \lambda N_{ub}^{ij} + d_{ub}^{ij} + \varepsilon_\phi \\ &= \rho_{ub}^{ij} - I_{ub}^{ij} + T_{ub}^{ij} + \lambda N_{ub}^{ij} + d_{ub}^{ij} + \varepsilon_\phi\end{aligned}$$

$$dt_{ub}^{ij} = dt_u^{ij} - dt_b^{ij} = 0, \quad dT_{ub}^{ij} = dT_{ub}^i - dT_{ub}^j \approx 0$$

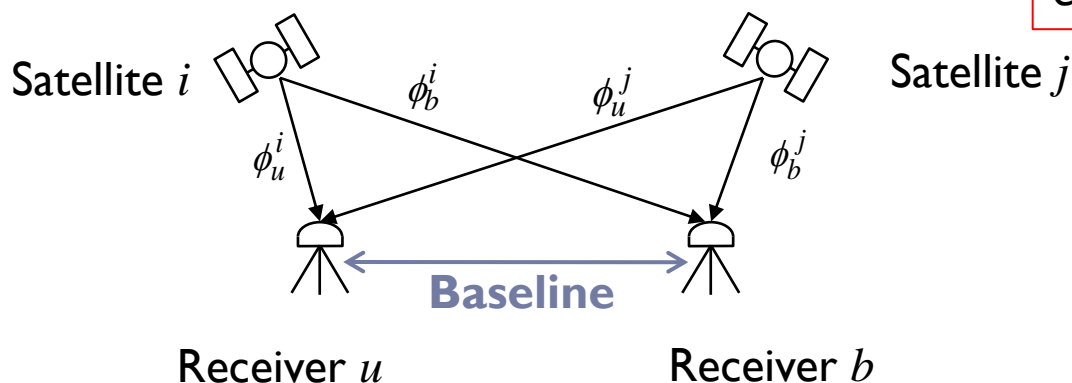
$$B_{ub}^{ij} = (\phi_{u,0} - \phi_0^i + N_u^i) - (\phi_{b,0} - \phi_0^i + N_b^i) - (\phi_{u,0} - \phi_0^j + N_u^j) + (\phi_{b,0} - \phi_0^j + N_b^j) = N_{ub}^{ij}$$

(short Baseline and same antenna type)

$$I_{ub}^{ij} = I_{ub}^i - I_{ub}^j \approx 0, T_{ub}^{ij} = T_{ub}^i - T_{ub}^j \approx 0, d_{ub}^{ij} = d_{ub}^i - d_{ub}^j \approx 0$$

Without reference station,
it is impossible to remove “receiver
And satellite clock error” completely !
Generate new observation
which means double difference.

Why do we say the
baseline limitation of RTK ?
(10-100 km or more)
It strongly depends on
each RTK engine !

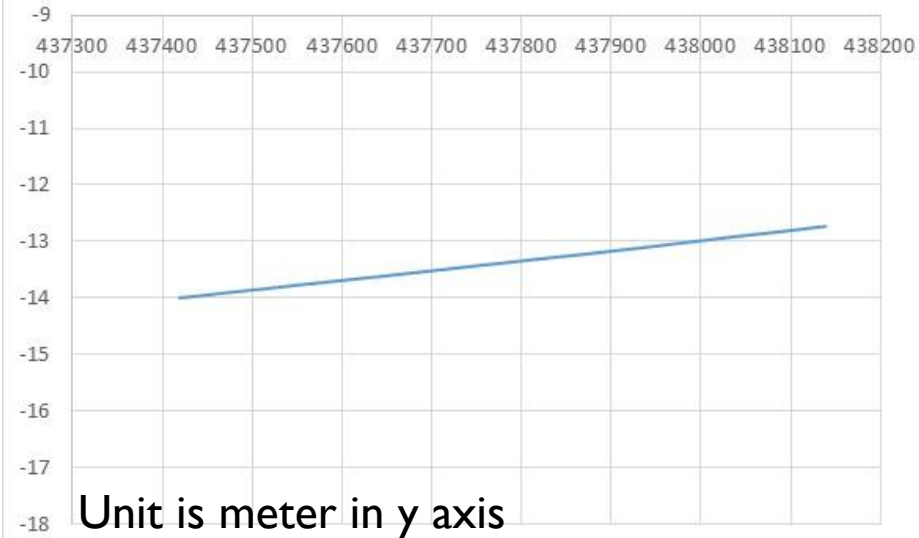
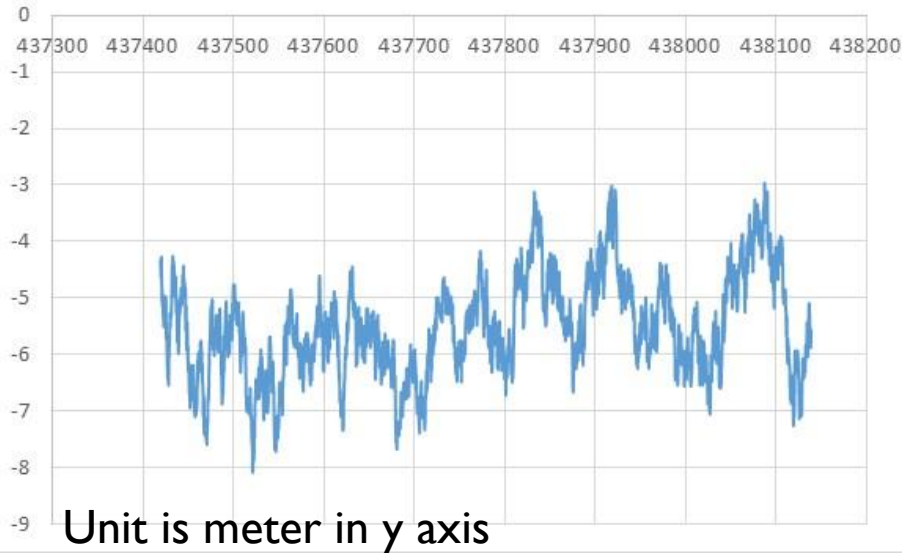


2. Integer Ambiguity Resolution

$$P_{rov_ref}^{sv1_sv2} = r_{rov_ref}^{sv1_sv2} + \varepsilon_{p,rov_ref}^{sv1_sv2}$$
$$\phi_{rov_ref}^{sv1_sv2} = r_{rov_ref}^{sv1_sv2} + N_{rov_ref}^{sv1_sv2} + \varepsilon_{\phi,rov_ref}^{sv1_sv2}$$

- ▶ Once you can resolve **integer N** in carrier phase double difference, you get accurate position about 1 cm.
- ▶ It can be imagine that the **pseudo-range (code) accuracy** is quite important.
- ▶ Code-phase is **noisy** (1 m-) but **absolute distance**
- ▶ Carrier-phase is **accurate** but **includes integer ambiguity**

3. Test results on the rooftop - double difference of 10 m baseline-



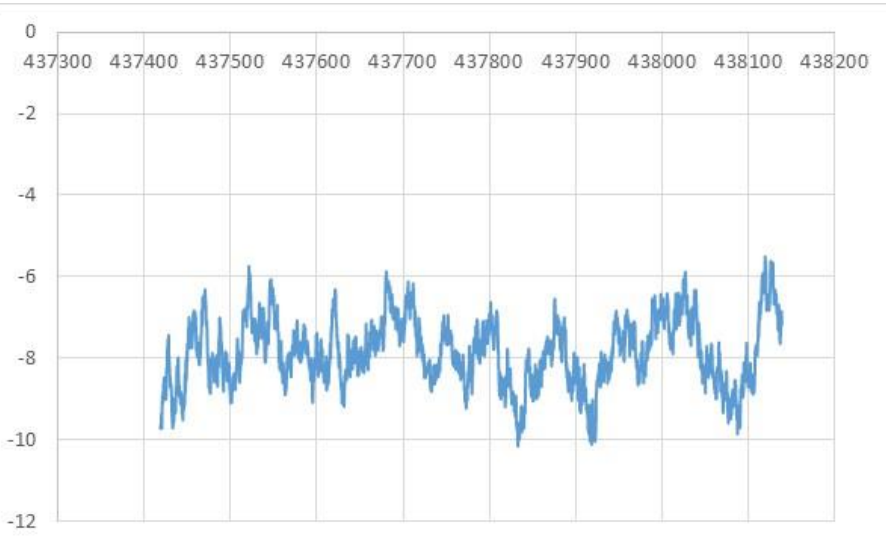
1. Reference satellite GPS PRN 16
and target satellite is GPS PRN 8
2. Which is code-phase double difference ?
3. If you subtract from right to left, what happen ?

$$P_{rov_ref}^{sv1_sv2} = r_{rov_ref}^{sv1_sv2} + \varepsilon_{p,rov_ref}^{sv1_sv2}$$

$$\phi_{rov_ref}^{sv1_sv2} = r_{rov_ref}^{sv1_sv2} + N_{rov_ref}^{sv1_sv2} + \varepsilon_{\phi,rov_ref}^{sv1_sv2}$$

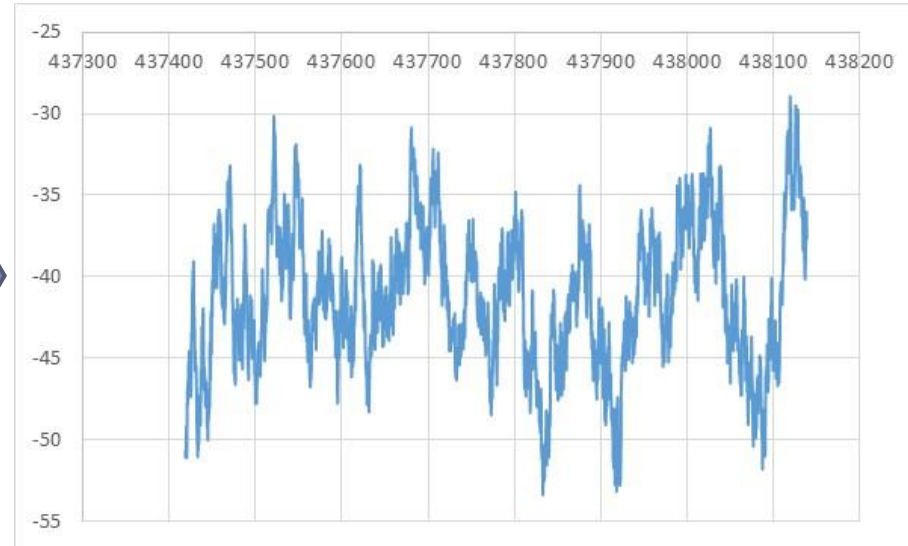


4. (Carrier DD) - (Code DD)



The unit is **meter**

Divided by wavelength
0.19029 m... (L1)



The unit is **cycle**

Probably, we guess the integer ambiguity between PRN16 and PRN8 is about - 40 ?

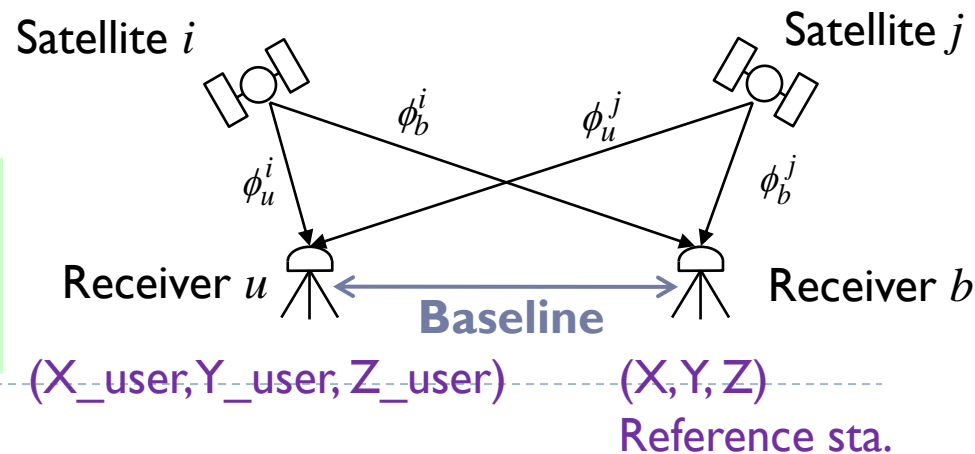
In fact, the average of this right results was - 41.3

5. What is the correct ambiguity ?

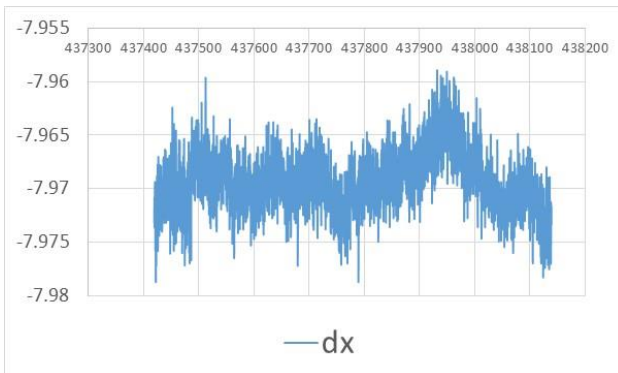
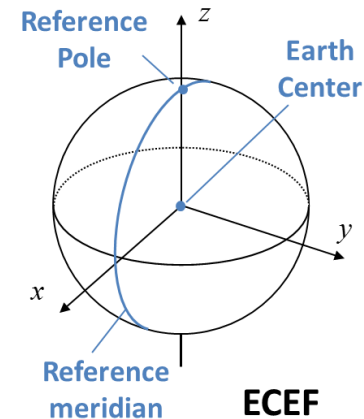
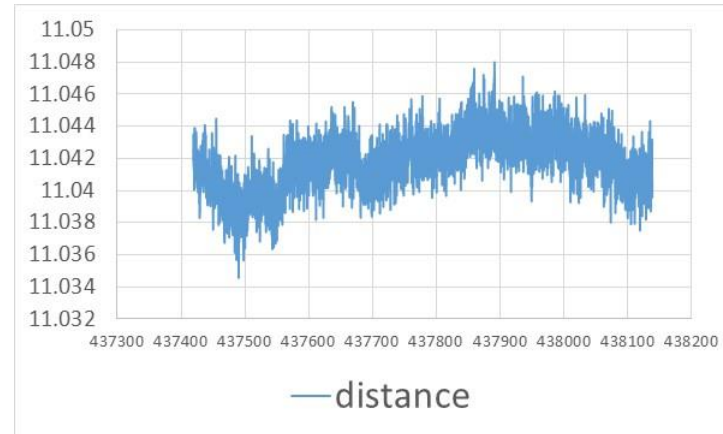
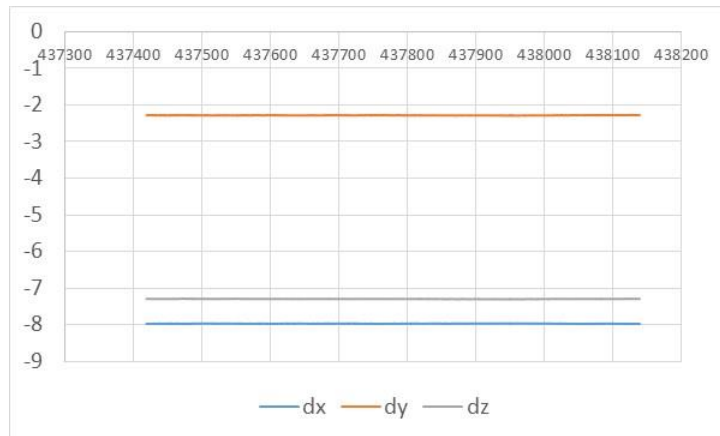
- ▶ “Integer least square method” tells us “– 42” in a single epoch !
- ▶ If you know the 3 or more ambiguities, you can estimate the user position with the level of carrier phase because only 3 unknowns remains.
- ▶ Then, (dx, dy, dz) can be estimated and finally,
- ▶ $(X_user, Y_user, Z_user) = (X, Y, Z) + (dx, dy, dz)$

$$P_{rov_ref}^{sv1_sv2} = r_{rov_ref}^{sv1_sv2} + \varepsilon_{p,rov_ref}^{sv1_sv2}$$

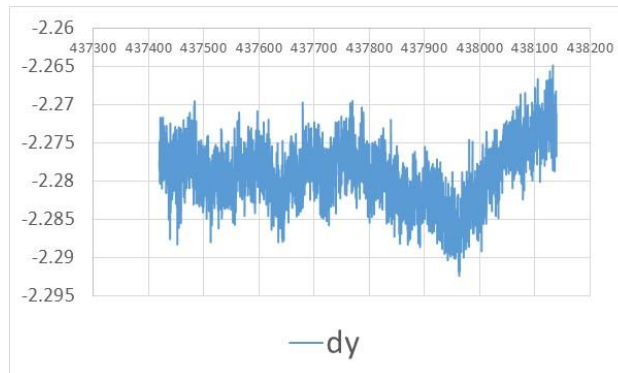
$$\phi_{rov_ref}^{sv1_sv2} = r_{rov_ref}^{sv1_sv2} + \underbrace{N_{rov_ref}^{sv1_sv2}}_{\text{ambiguity}} + \varepsilon_{\phi,rov_ref}^{sv1_sv2}$$



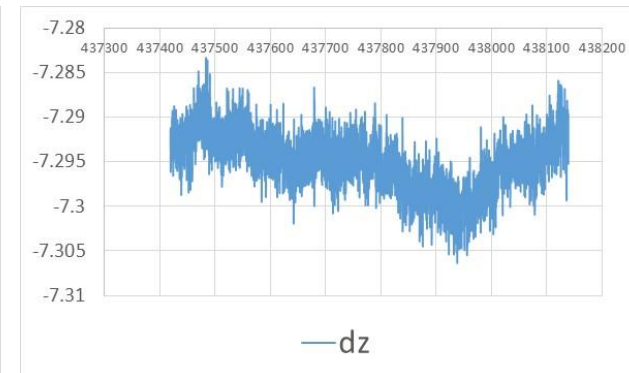
6. Test results (dx, dy, dz)



Std. = 2.8 mm



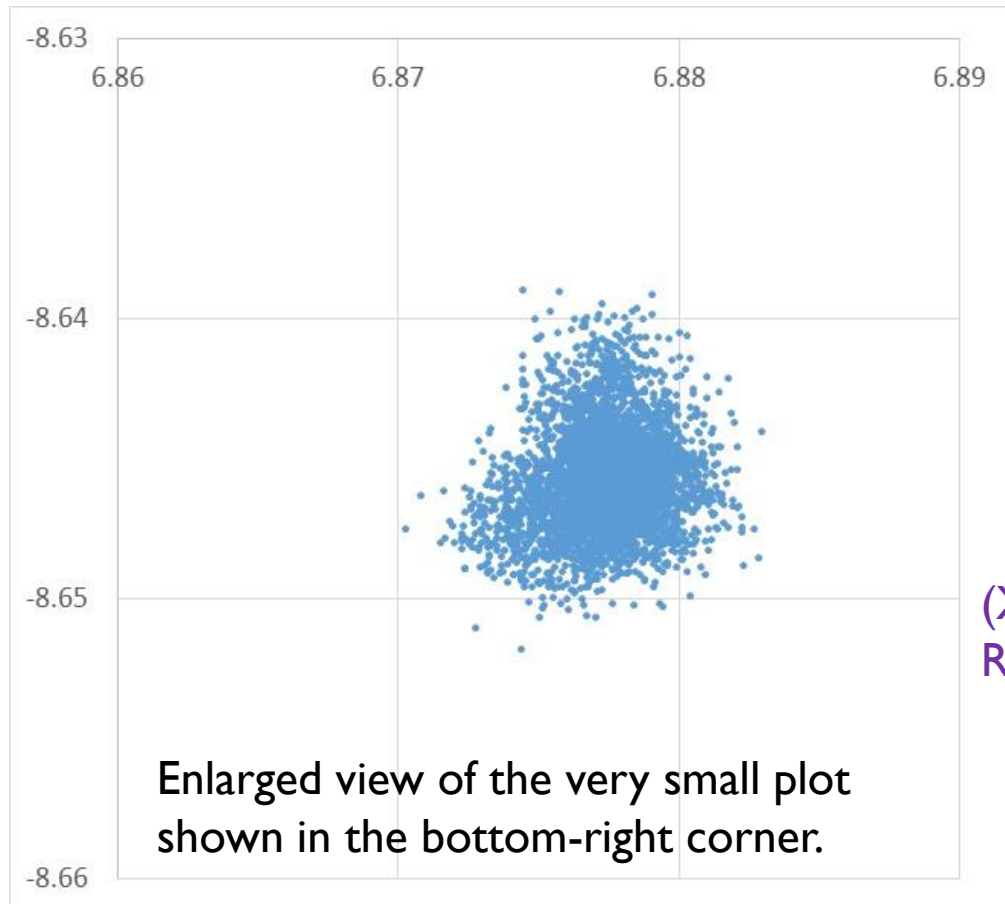
Std. = 4.0 mm



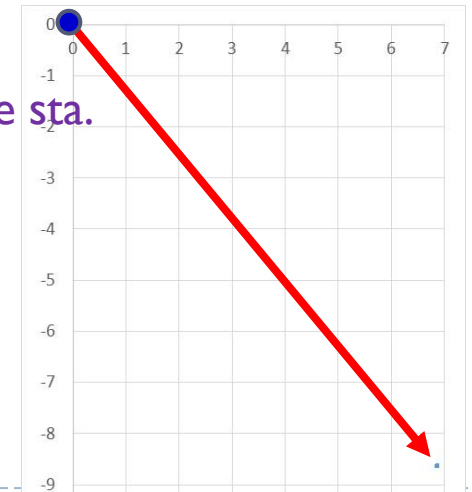
Std. = 3.4 mm



7. Convert to horizontal positions



(X,Y,Z)
Reference sta.



(X_user,Y_user,Z_user)

I am repeating myself, RTK tells you only dx, dy, dz.
You have to decide the precise reference positions !

RTKNAVI demonstration

- ▶ SPP
- ▶ DGNSS
- ▶ RTK-GNSS



QZSS correction services

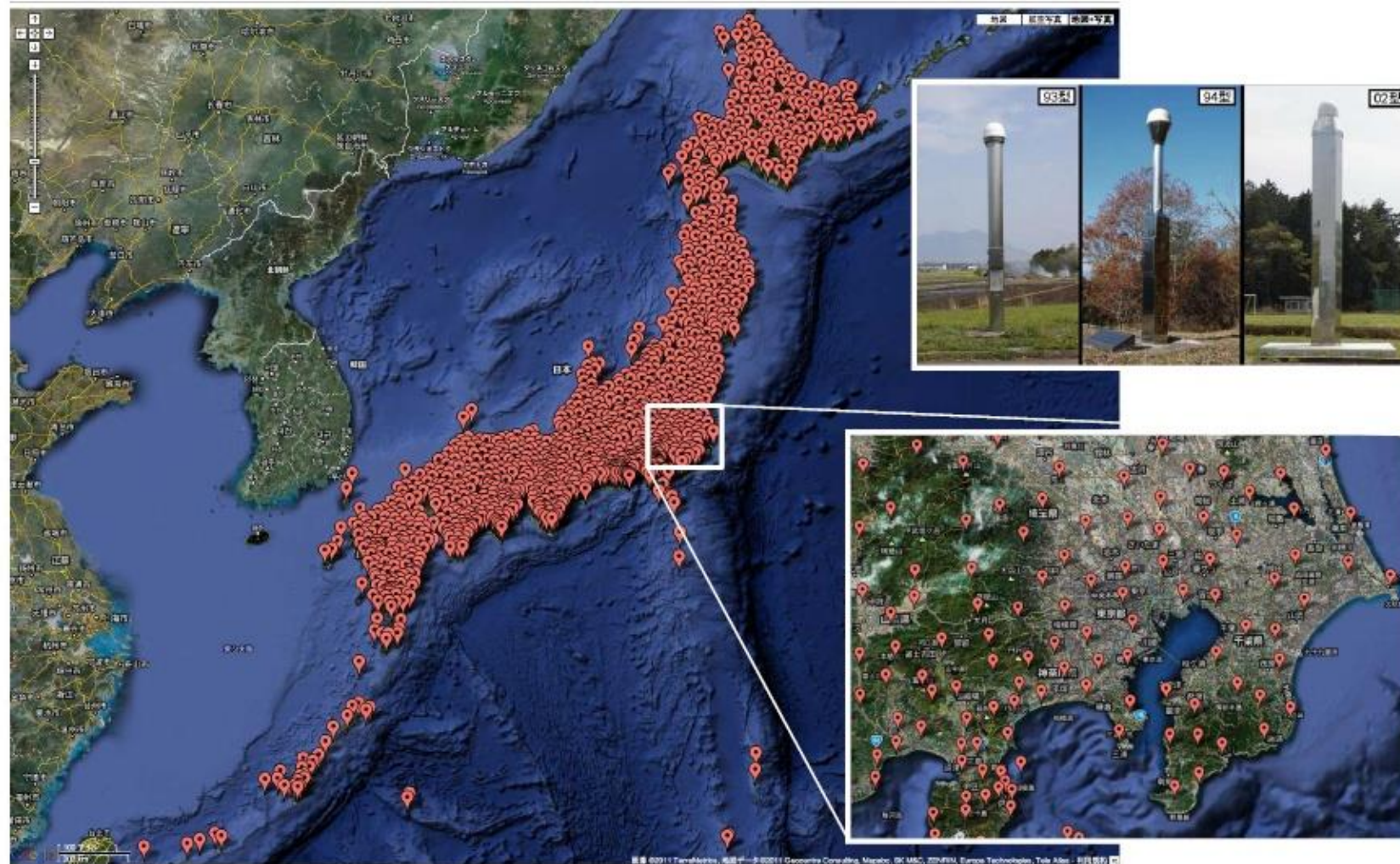
- ▶ See the actual performance
- ▶ GNSS TUTOR 公開



Japanese GEONET

おおむね10-30km程度以内の間隔

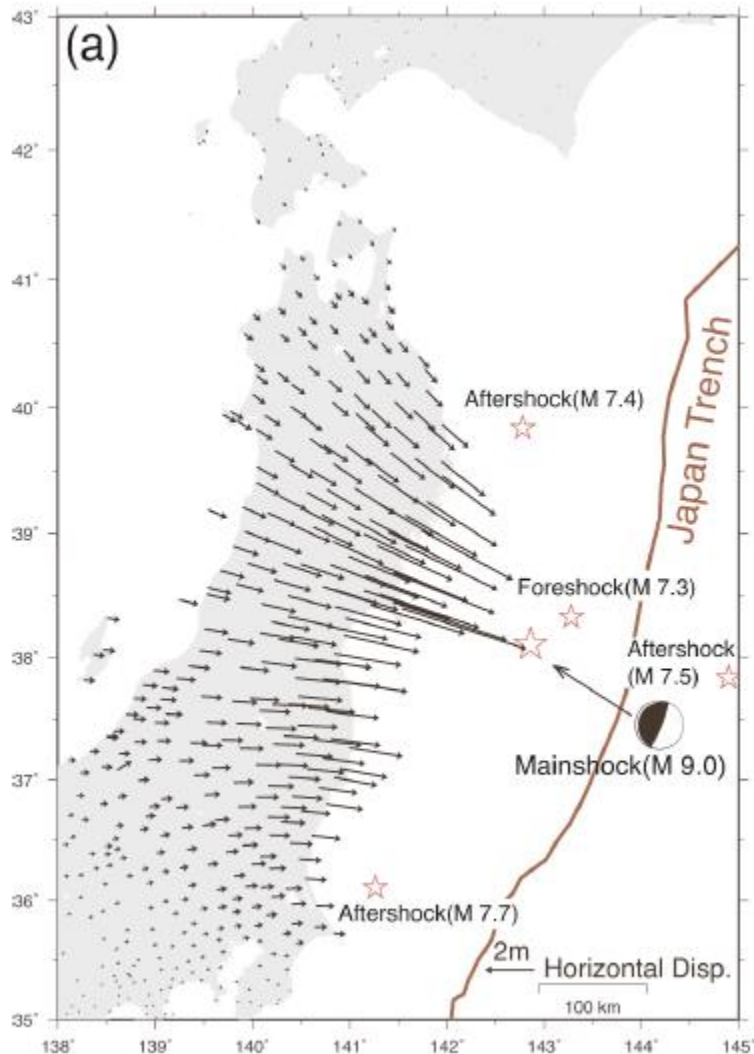
GEONET STATIONS MAP by Google Map : [GEONET Station](http://geonet.gsi.go.jp)



The station coordinates are based on the JGD solutions on 2002/1/1 provided by GSI. Height: ellipsoidal height (MGS55M).

(<http://terras.gsi.go.jp/ja/index.htm>)

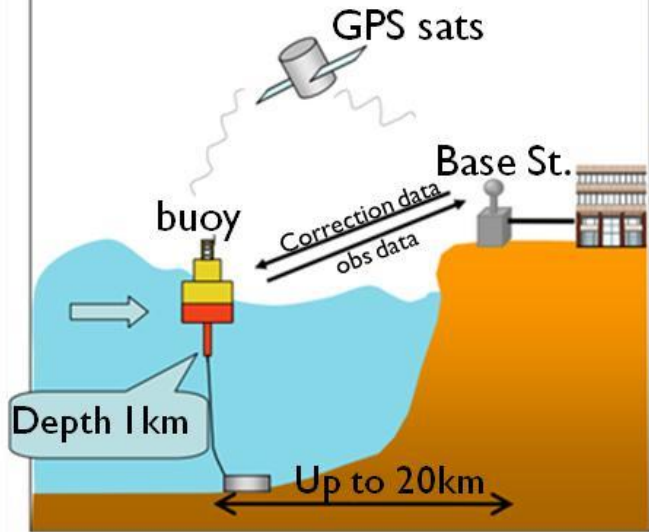
Application of RTK (ground surface movement)



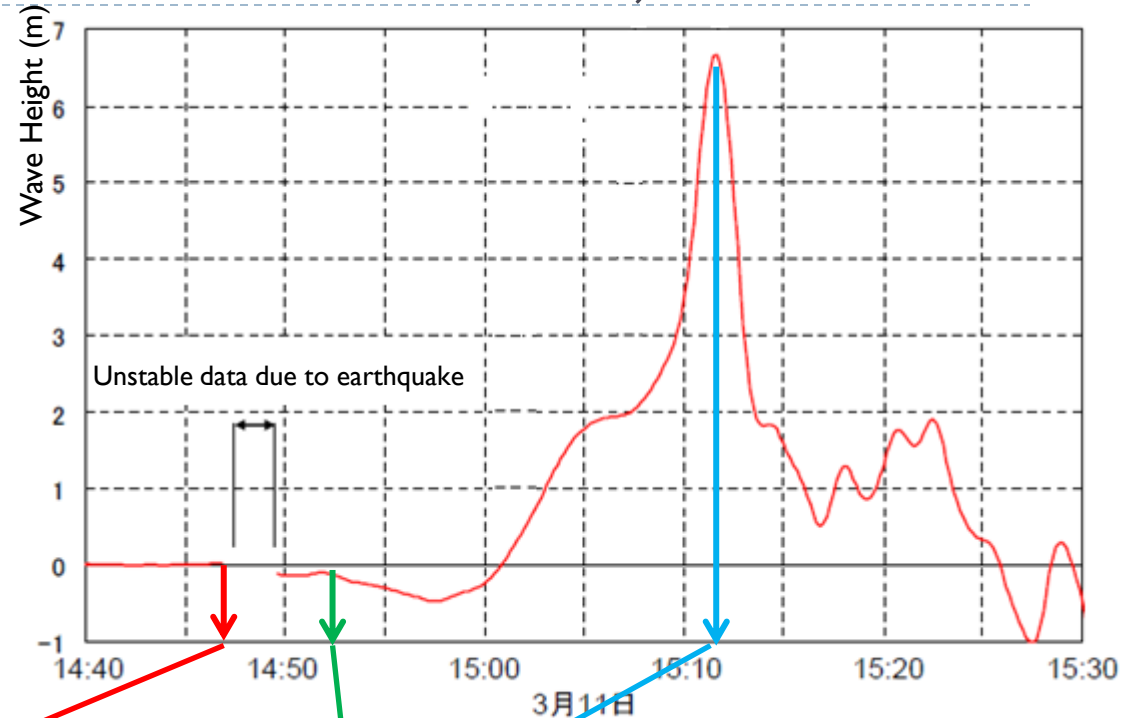
After the main shock of the March 2011 Tohoku Earthquake, GSI tried to get initial results of ground surface movements detected by GEONET, which consists of 1,240 GNSS-based control stations and the central analysis unit. After a week, it revealed that Oshika station, close to the epicenter of the main shock, **moved 5.3 m eastward and subsided 1.2 m, the largest movement ever observed by GEONET.**

by Coordinates and EPS Letter

GPS buoy outline



Application of RTK (Tsunami detection)



- ▶ South Iwate buoy
 - ▶ 10km offshore
 - ▶ Depth 200m

- ▶ 14:46 Earthquake
- ▶ 14:53 First detected Tsunami motion
- ▶ 15:12 Tip of Tsunami wave

by Port and Airport Research Institute

Smart Construction

► Computer aided construction

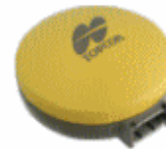


Precision Farming

▶ Precision farming resolves the issue in decreasing farm family



コンパクトなモニターで
タッチスクリーン式
System110 (トプコン製)

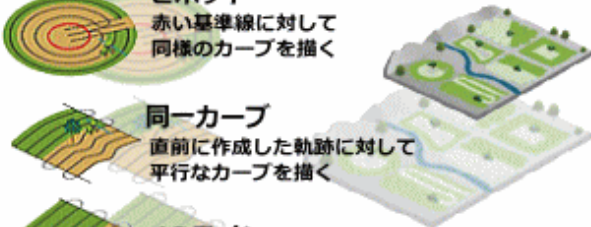


- * Agricultural management
- * Low cost receiver
- * Amateur can control
- * Improvement of harvest
- * Improvement of quality
- * Autonomous helicopter

ガイダンスライン
の設定

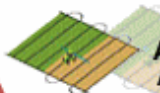
ピボット

赤い基準線に対して
同様のカーブを描く



同一カーブ

直前に作成した軌跡に対して
平行なカーブを描く



ABライン

A、B点を決め、両点から作られる
ラインに沿うように農機を導く

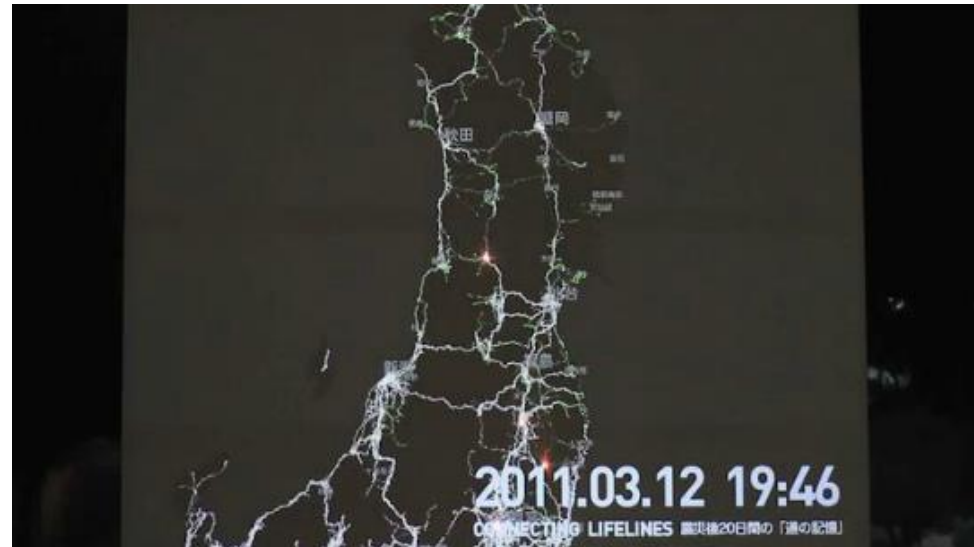
走行軌跡をレポート出力

GISで管理が可能です!!



Quality of Big data

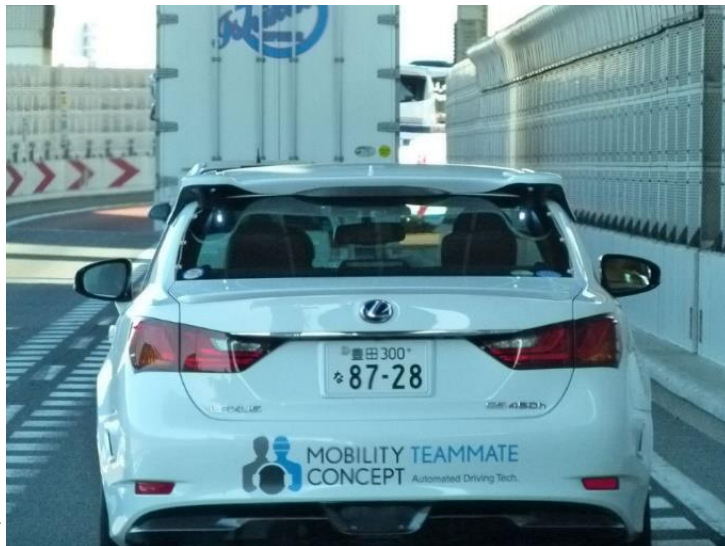
- ▶ Road condition monitoring
- ▶ Traffic information in big disaster



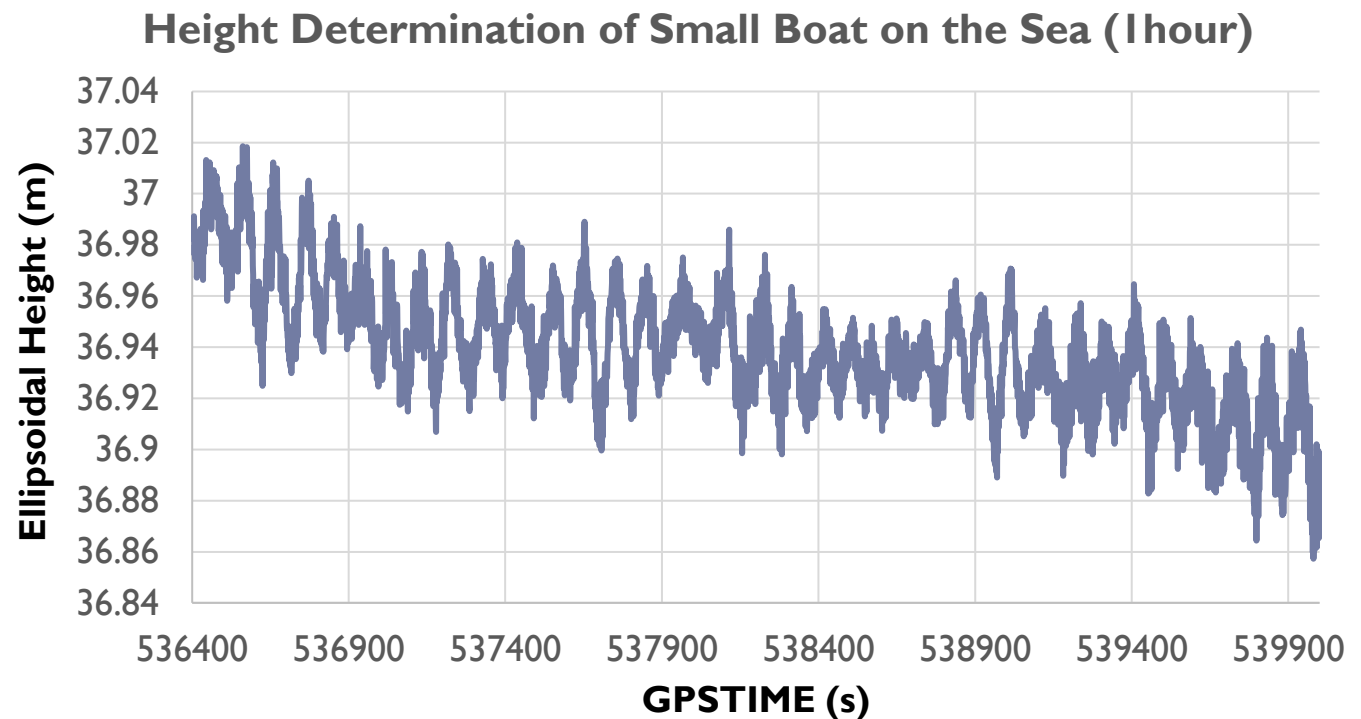
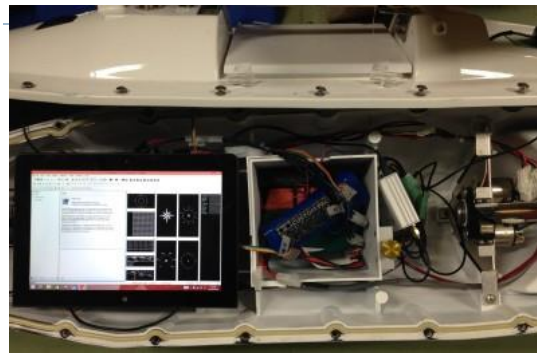
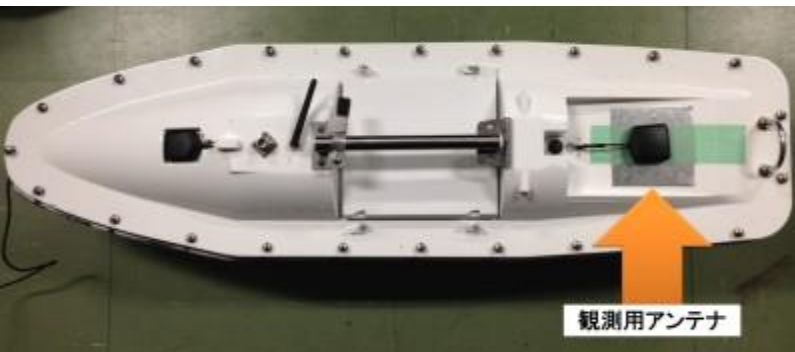
▶ Accuracy improves the quality of Big data

Autonomous car with precise map

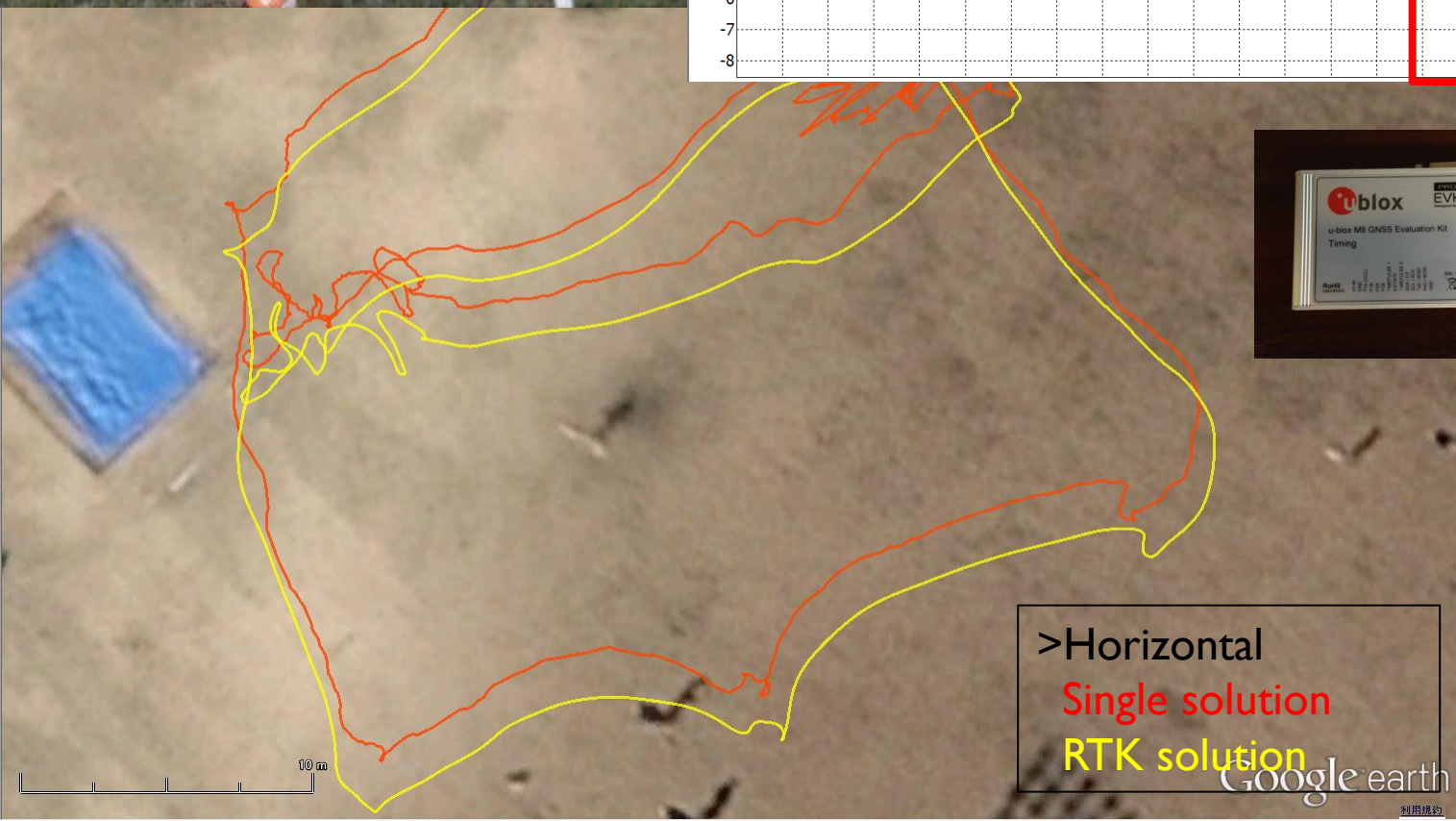
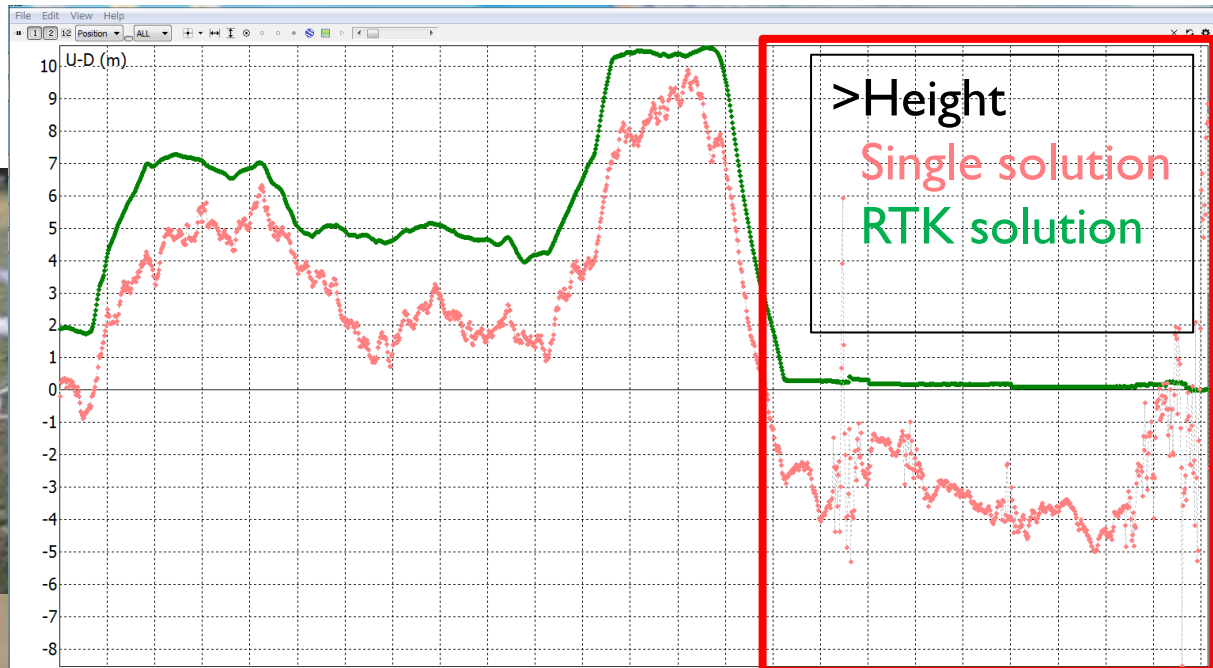
- * **Autonomous car**
- * **Smart control**



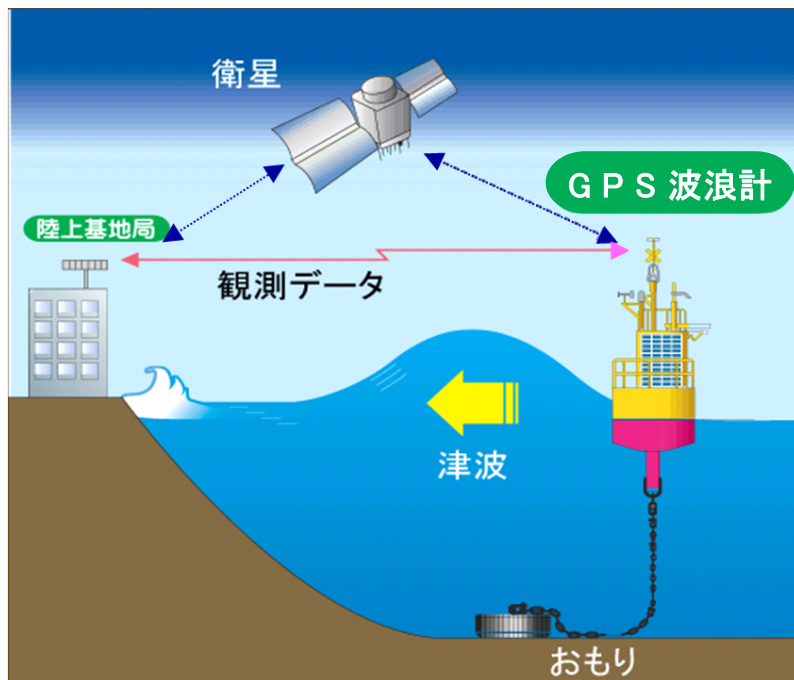
Altitude determination for boat



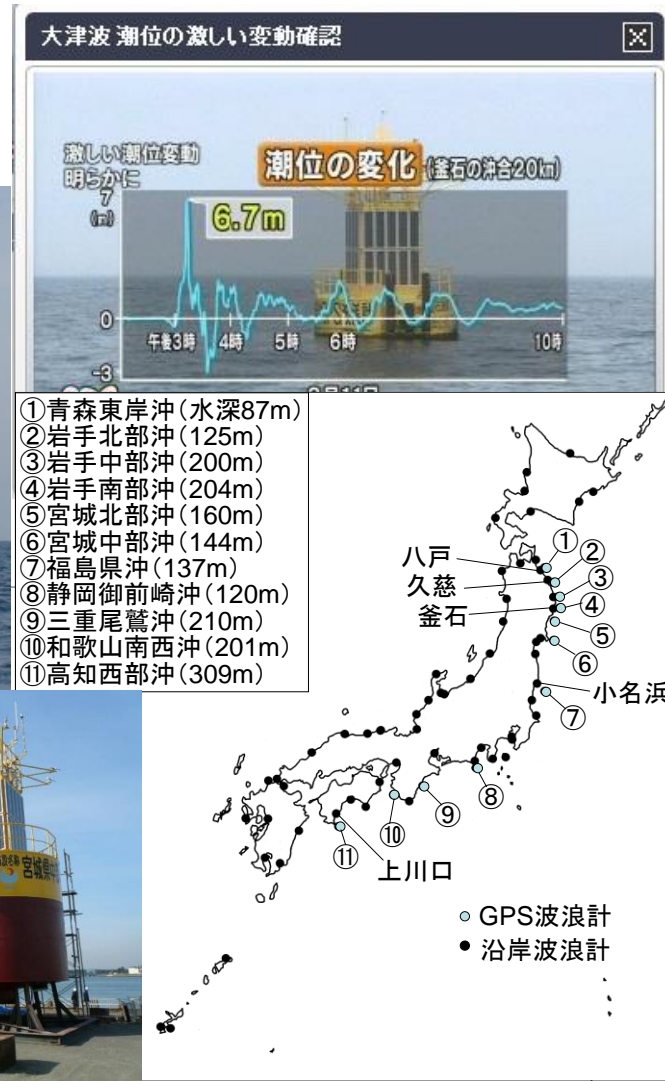
RTK for UAV



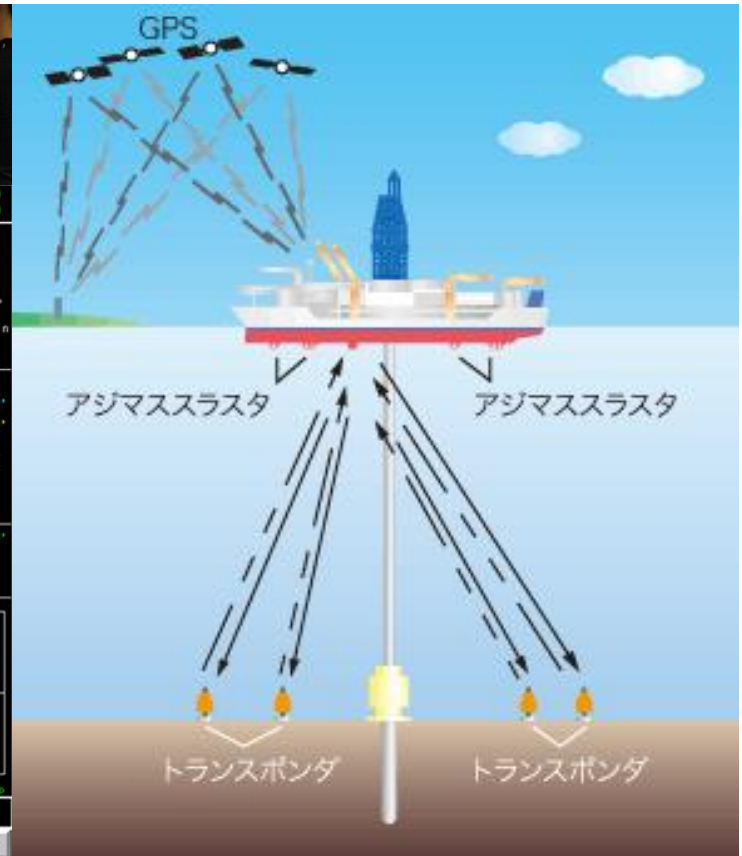
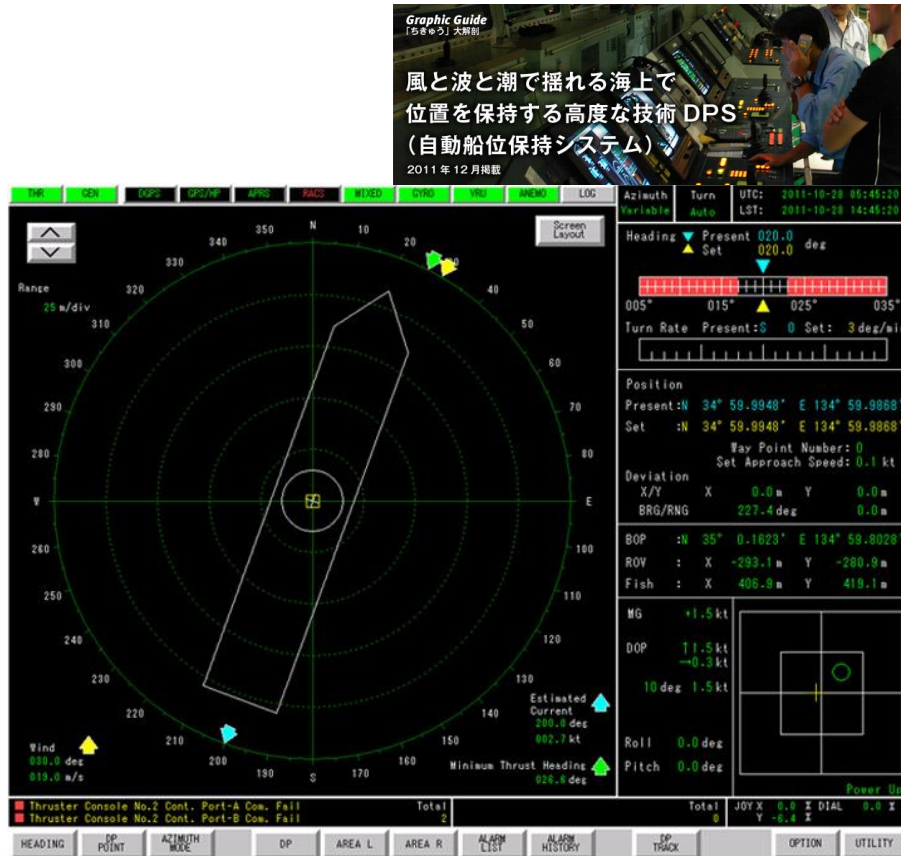
Tsunami Detection



Wide Area RTK / PPP can be used



DPS (Dynamic Positioning System)



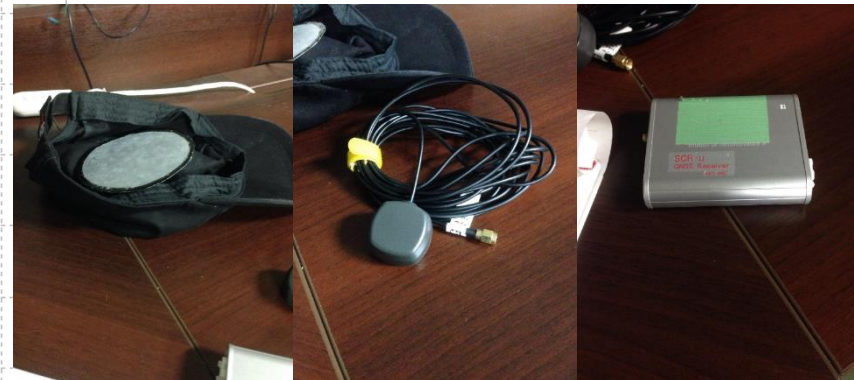
- *Wide Area RTK / PPP / DFMC SBAS can be used
- *50 cm (2drms) is required.
- *DPS is the key technology for autonomous Ship

Recent Test : Running



Horizontal

5 m



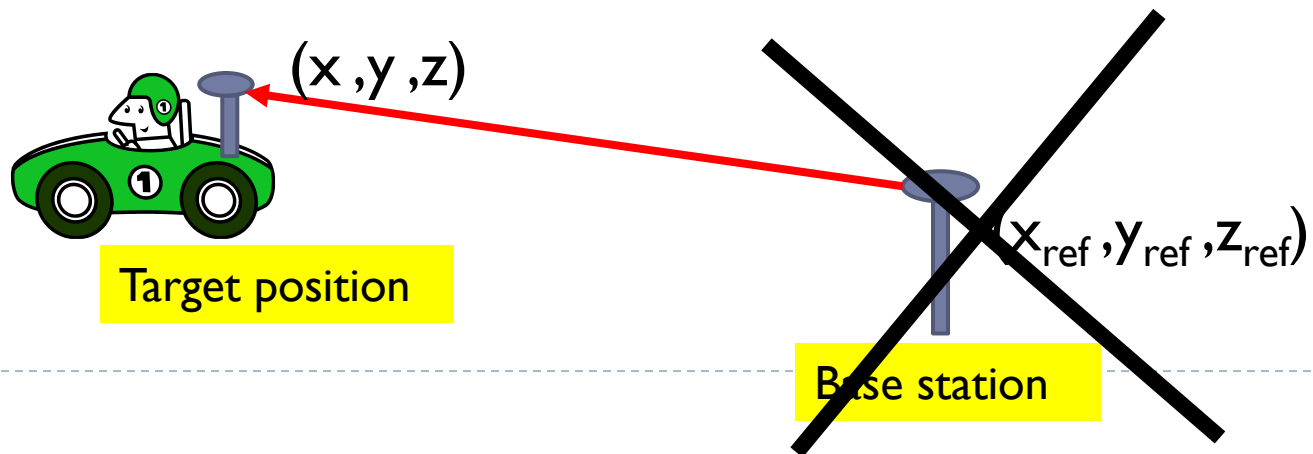
Difference between expensive and low-cost receiver

	Survey-grade receiver	Low-cost receiver
Cost	\$ 100,00~	\$100~
Multiple GNSS	Perfect	BeiDou or Glonass Other are OK
Multiple Frequency	Perfect	L1/B1/E1/G1 only
Number of channel	400-500-	-100
RTK (short baseline) + open sky	Perfect	Almost perfect
RTK (over 20 km baseline) + open sky	Almost perfect up to 100 km or more	Impossible
RTK under mid obstructed area (short)	Almost perfect	May be difficult
RTK under dense obstructed area (short)	Sometimes not good	Difficult
Accuracy of fixed position + open	mm	→
Accuracy of code position + open	Deci-meter	1-2 meter



PPP (Precise Point Positioning)

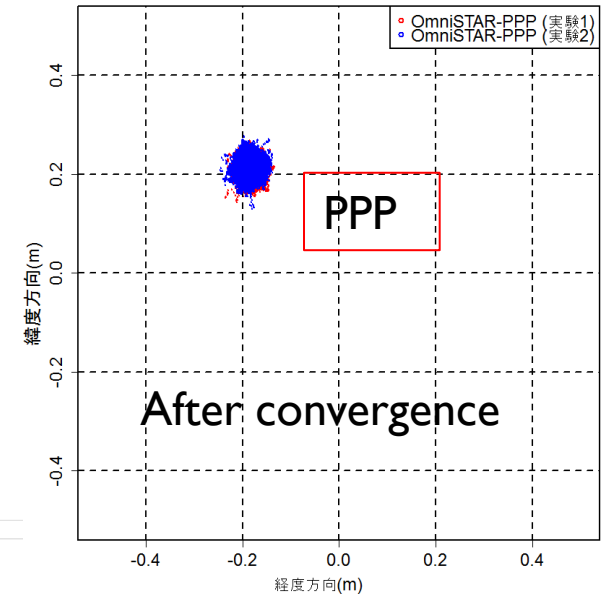
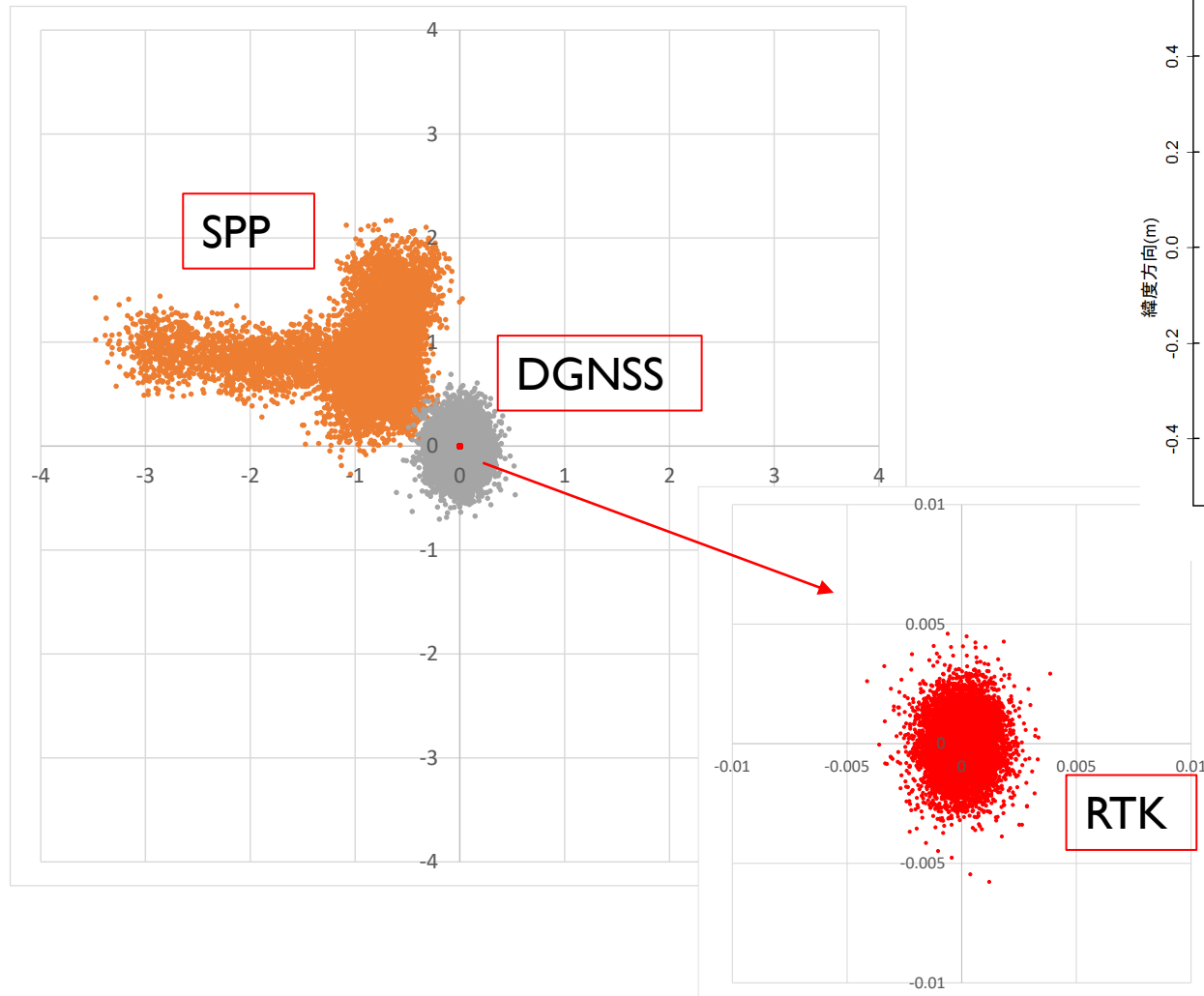
- ▶ We discussed about relative positioning to cancel the common errors.
- ▶ We switch from relative positioning to point positioning. Base station is not required.
- ▶ We need to consider the measurement errors **more in details** if we want to have **centimeter-level accuracy in the point positioning**.



PPP mitigates...

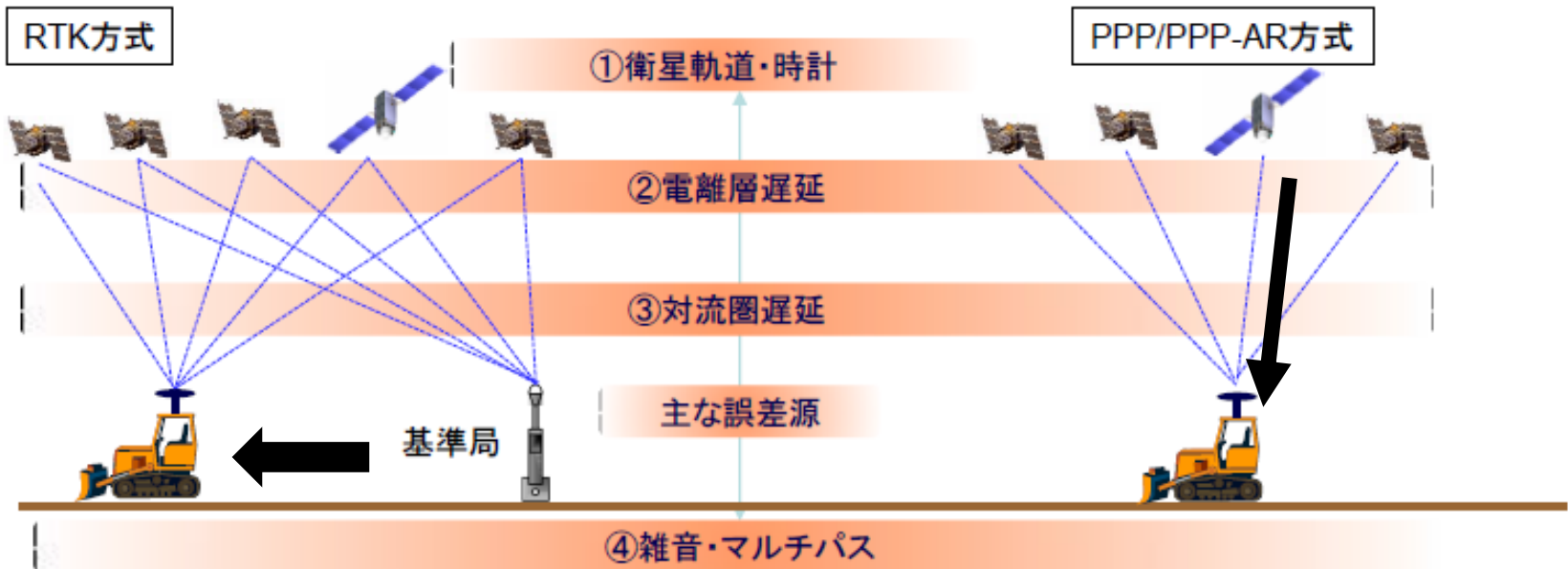
Source	Potential error size	Error mitigation
Satellite clock model	2 m (rms)	Centimeter level precise satellite clock
Satellite ephemeris prediction	2 m (rms) along the LOS	Centimeter level precise satellite position
Ionospheric delay	2-10 m (zenith) Obliquity factor 3 at 5°	Dual-frequency can mitigate it completely
Tropospheric delay	2.3-2.5m (zenith) Obliquity factor 10 at 5°	Precise model with centimeter level
Multipath (open sky)	Code : 0.5-1 m Carrier : 0.5-1 cm	Carrier-phase is used
Receiver Noise	Code : 0.25-0.5 m (rms) Carrier : 1-2 mm (rms)	Carrier-phase is used

Actual performance...



Accuracy (95%)
SPP : 1.36m
DGNSS : 0.44m
RTK : 3 mm
PPP : 3.4 cm

RTK and PPP

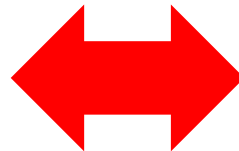


Reference station

Continuous communication

Instantaneous position

10-100km limitation



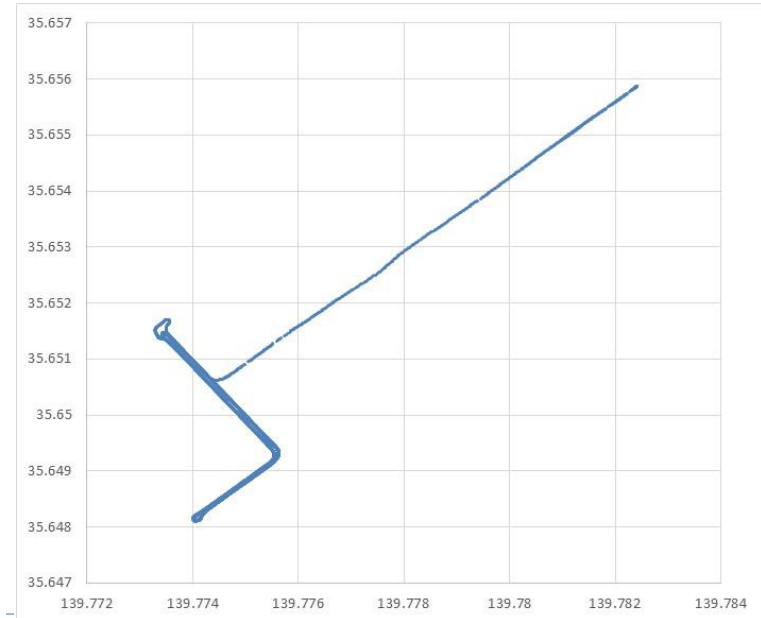
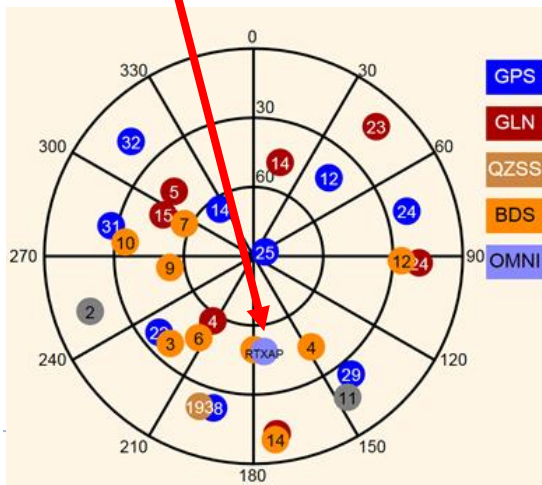
Continuous communication
through satellite

Wait for 5-30 minutes

No limitation

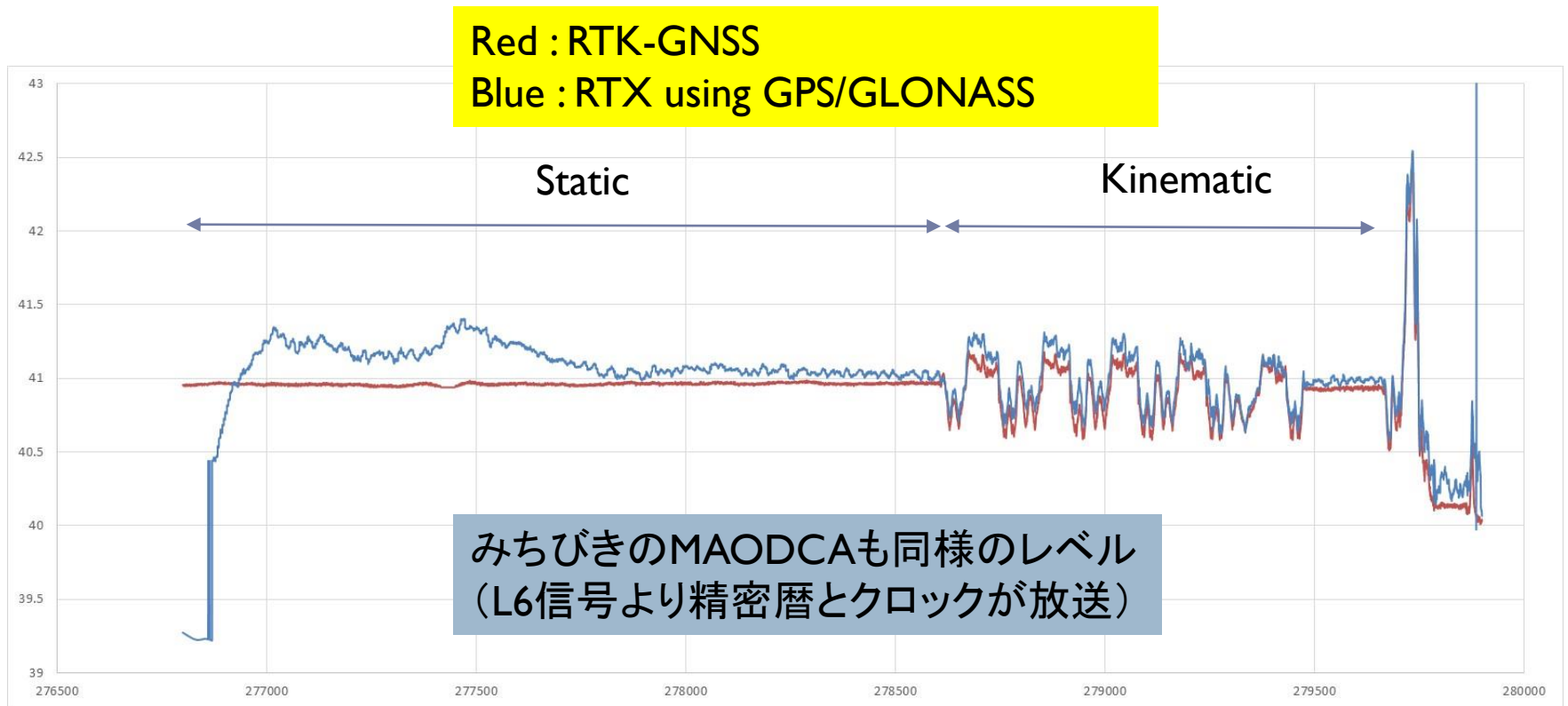
Precise Point Positioning Test using commercial service

- ▶ 30 minutes static and 15 minutes kinematic
- ▶ Trimble SPS855+RTX (PPP) option
- ▶ Comparison with RTK results
- ▶ Omni-star was used
- ▶ Open Sky



Horizontal plots at Harumi Area

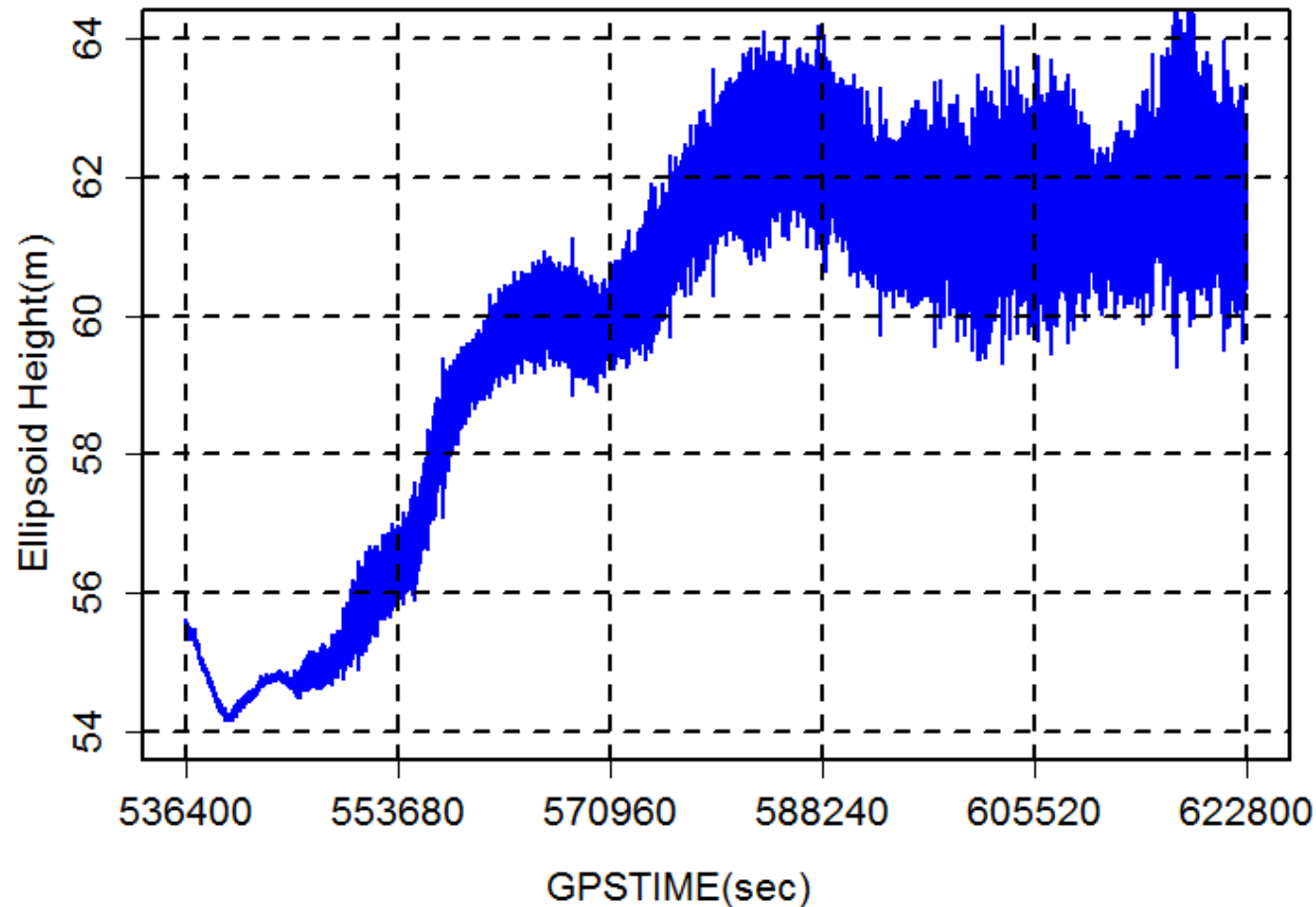
Altitude Comparison between RTK and RTX (PPP)



The accuracy was maintained within several centi-meters after 15 minutes of power on. Small bias (about 10cm) was deduced from other reason.

PPP Positioning Results of Height for SHIP

▶ 2015.7.25. (Tokyo ~ Ogasawara)



What is RTCM ?

- ▶ The standard for differential global navigation satellite system was defined in RTCM Special Committee 104 and its current version is Version 3. RTCM standard for differential global navigation satellite services are **communication protocols between reference stations and mobile receivers** which allow very high accurate positioning, when compared with positioning system without augmentation.



What is NTRIP ?

- ▶ The NTRIP was also defined in the RTCM Special Committee 104. NTRIP stands for “**Networked Transport for RTCM via Internet Protocol**”. It is based on Hypertext transfer Protocol version 1.1 and the intention is to disseminate differential correction data through the internet.